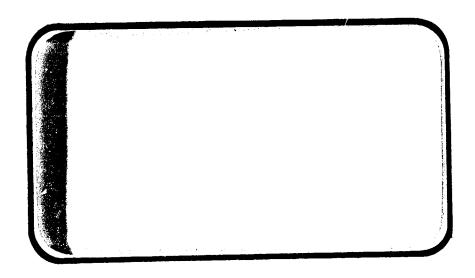


# NATIONAL AERONAUTICS AND SPACE ADMINISTRATION



(NASA-CR-134101) HEAT TRANSFER TESTS OF AN 0.006-SCALE THIN SKIN SPACE SHUTTLE THERMOCOUPLE MODEL (41-0) IN THE LANGLEY RESEARCH CENTER VARIABLE DENSITY (Chrysler Corp.) 243 p HC \$15.25 CSCL 22B N74-33321

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SPACE SHUTTLE

AEROTHERMODYNAMIC DATA REPORT



JOHNSON SPACE CENTER
HOUSTON, TEXAS

DATA DANagement services

SPACE DIVISION CHRYSLER
CORPORATION

DMS-DR-2096 NASA CR-134,101

HEAT TRANSFER TESTS OF AN 0.006-SCALE THIN SKIN

SPACE SHUTTLE THERMOCOUPLE MODEL (41-0) IN THE

LANGLEY RESEARCH CENTER VARIABLE DENSITY

TUNNEL AT M = 8 (OH13)

bу

D. G. Walstad Rockwell International

Prepared under NASA Contract No. NAS9-13247

by

Data Management Services
Chrysler Corporation Space Division
New Orleans, La. 70189

for

Engineering Analysis Division

Johnson Space Center
National Aeronautics and Space Administration
Houston, Texas

#### WIND TUNNEL SPECIFICS:

Test Number:

Larc/VDT 644

NASA Series No:

OH-13

Model Number:

41-0

Date:

13 June 1973

Occupancy:

8 Hours

#### FACILITY COORDINATOR:

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Chrysler Corporation Space Division assumes no responsibility for the data presented other than display characteristics.

HEAT TRANSFER TESTS OF AN 0.006-SCALE THIN SKIN SPACE SHUTTLE THERMOCOUPLE MODEL (41-0) IN THE

LANGLEY RESEARCH CENTER

VARIABLE DENSITY TUNNEL AT M = 8 (OH13)

by

D. G. Walstad, Rockwell International

#### ABSTRACT

This report presents data obtained from heat transier tests on an 0.006-scale, Space Shuttle Orbiter Vehicle in the Lengley Research Center, Variable Density, Mach 8 Wind Tunnel. The purpose of this test was to obtain orbiter entry heating distributions and to correlate phase change paint data with thermocouple data. The orbiter was tested at 0, 30, and 35 degrees angle of attack at Reynolds numbers of 1, 2, 3, 4, and 6 million per foot. Temperature data were obtained from a total of 57 thermocouples.

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ביי אינו ניין	COEFFICIENT SCHEDULE	VARYING PARAMETER	PAGE
TITLE			
Heat Transfer Coefficient Ratios on	A	Y(BF), ALPHA,	1-62
Orbiter Body		/ 67 /WI	•
Heat Transfer Coefficient Ratios on	ф	2Y/B, ALPHA,	63-140
Orbiter Wing		1M/ H5 1MH/ H1	

COEFFICIENT SCHEDULE:

A: H/HO VS. X/L

B: H/HO VS. X/C

STOCKER OF THE TOTAL STOCKER STOCKER OF THE STOCKER

# NOMENCLATURE

Symbol	Plot Symbol	Definition
ъ	-	Model skin thickness, inches
b/2	-	Percent span, wing
c	-	Specific heat of model material, BTU/lbm-°R
$c_{\mathbf{p}}$	-	Specific heat at constant pressure of airstream, BTU/lbm-°R
g	-	Gravitational constant, ft/sec2
h	Н	Heat-transfer coefficient, BTU/ft2-sec-oR
h <sub>ref</sub>	НО	Reference heat transfer coefficient, BTU/ft2-sec-°R
H	-	Enthalpy, BTU/1b
Ho	-	Stagnation enthalpy, BTU/lb
Haw	-	Adiabatic wall enthalpy, BTU/lb
H <sub>W</sub>	-	Model wall enthalpy, BTU/1b
k	-	Thermal conductivity coefficient, BTU/ft-sec-oR
M	MACH	Mach Number
ORB	ORB	Orbiter
P	-	Static pressure, psia
Po	-	Stagnation pressure, psia
Poo	-	Tunnel free stream pressure, psia
ģ	-	Heat flux, BTU/ft2-sec
dot	-	Stagnation-point heat-transfer rate calculated using Fay and Riddell's equation, BTU/ft2-sec.
R	-	Gas constant, Ft-lb/slug-°R
r	HAW/HT	Adiabatic wall temperature ratio, TAW/To
$R_{ullet}$	-	Reynolds Number
$R_{n}/1$	RN/L	Unit Reynolds number, per foot

# NOMENCLATURE (Continued)

Symbol	Plot Symbol	<u>Definition</u>
rs	-	Radius of scaled one-foot sphere, inches
T	-	Temperature, °R
To	-	Stagnation temperature, °R
T <sub>w</sub>	-	Model wall temperature, °R
t	-	Time, sec.
u	-	Velocity, ft/sec
W	-	Density of model material, Lbw/ft3
x	-	Longitudinal distance coordinate, feet
α	ALPHA	Model angle of attack, deg.
μ	-	Viscosity of air
ρ	-	Density of air

# Subscripts

Symbol	Plot Symbol	Definition
aw	-	Adiabatic wall
•		Tunnel free stream conditions
0	0	Tunnel stagnation conditions
w		Model well conditions

# Superscripts

Symbol	Plot Symbol	<u>Definition</u>
•	_	Conditions behind shock

# NOMENCLATURE (Concluded)

# ADDITIONS TO STANDARD NOMENCLATURE

Symbol_	Plot S <u>ymbol</u>	Description Description
h/h <sub>ref</sub>	н/но	ratio of local heat transfer coefficient to reference heat transfer coefficient.
X/L	X/L	longitudinal position expressed as ratio of distance from orbiter nose to actual orbiter length.
x/c	x/c	chordwise position expressed as ratio of distance from leading edge to chord length.
Y(BP)	Y(BP)	distance from orbiter centerline in outboard direction.
2Y/B	2Y/B	spanwise position expressed as ratio of distance from fuselage centerline in the outboard direction to one half the span length.
ß	BETA	angle of sideslip, degrees.
δ <sub>e</sub>	ELEVON	surface deflection angle, positive deflection trailing edge down; degrees.
δ <sub>r</sub>	RUDDER	rudder deflection angle, degrees.

#### CONFIGURATION INVESTIGATED

The orbiter was an 0.006-scale representation of the modified VL70-000089B lines. The main body lines were defined by Grumman drawing SS-H-00326-11 and the nose was defined by Drawing SS-H-00326-15. The orbiter was constructed of Grumman material "G" (Stycast) with thin skin, 15-5 PH stainless steel inserts. The inserts were located on underside centerline region, left-hand wing underside and left-hand windshield. Thermocouples were spot welded to the skin and clamped in bundles at convenient locations within the model. The model had no provisions for elevon, rudder, or body flap deflections.

The model configuration consisted of the following components:

Components	Description
B <sub>10</sub>	Fuselage per -89B lines, 2A configuration
c <sub>5</sub>	Orbiter canopy used on fuselage B <sub>10</sub>
D <sub>7</sub>	Manipulator housing per -93 lines, 2A configuration
F <sub>14</sub>	Aft body flap used on fuselage B <sub>10</sub>
M <sub>3</sub>	OMS Pods per -94A lines, 2A configuration
V <sub>5</sub>	Centerline vertical tail, double wedge airfoil with rounded leading edges
W87	Wing per -93 lines. Used on fuselage B <sub>10</sub>

Table III and Figure 1 provide a detailed description of the model components.

#### MODEL INSTRUMENTATION

The orbiter was instrumented with 57 iron-constantan thermocouples spot welded to thin skin (nominally 0.030-inch) 15-5 PH stainless steel inserts. The leads were 50 feet in length and were connected to 18 pin connectors. The exact location of each thermocouple is presented in Table IV and illustrated in Figure 2.

#### TEST FACILITY DESCRIPTION

The Langley Mach 8.0 Variable-Density Hypersonic tunnel is located in Building 1247D and is under the direction of the Aerophysics Division. This tunnel is used for fundamental aerodynamic and fluid dynamic investigation over large Reynolds Number ranges obtaining pressure and heat transfer measurements. The test medium is air and is heated by a combination of dowtherm and electrical resistance heaters. The models are sting mounted with injection from the bottom of the test section after flow has been established. The tunnel has an axially symmetric contoured nosale. The tunnel cross section is 18 inches in diameter with a core of 4 to 14 inches depending on pressure.

# Examples of operating conditions are as follows:

Stagnation pressure (psia)	15 to 2930
Stagnation temperature °(R)	1160 to 1510
Mach Number	7.5 to 8.0
Reynolds Number (1/Ft)	0.1 x 106 to 12.0 x 106
Running time (sec, for Exhaust into vacuum tank Exhaust into Atmosphere	90 600

## TEST PROCEDURES

Heat transfer data were obtained by measuring the temperature rise over a period of time from a total of 57 iron-constantan thermocouples. The model was injected into the flow stream from the bottom of the test section and held on tunnel center line for approximately 3 seconds, during which time temperature measurements were taken. Model angle of attack had to be preset manually before securing the test section for testing.

A maximum of 54 thermocouples could be recorded at any given time. Temperature measurements were collected through the Beckman Data Acquisition system. The thermocouple leads were routed through the model support system and connected to a connector panel. Leads that were exposed to flow conditions were wrapped with asbestos tape.

Prior to testing, a thermocouple heat response check, through the data system, was performed on all thermocouples. As an aid in making this check, a fiberglass mask with thermocouple locating holes in it was provided. By placing the tip of a soldering iron to the hole, an accurate and quick response could be obtained.

The model was leveled in pitch and roll by means of a leveling block which attached to the top of the orbiter.

#### DATA REDUCTION

Heat transfer data were computed using the following equations and procedures:

The thermocouple heat-transfer data were reduced by the one-dimensional thin wall equation:

$$\dot{q} = \text{Web } \frac{dT_W}{dt}, \text{ Btu/ft}^2 - \text{sec}$$

where the symbols are as defined in the Nomenclature section.

The theoretical stagnation-point heat-transfer rate was calculated using Fay and Riddell's equation:

$$\dot{q}_{ot} = 0.94 \ (\rho_w \mu_w) \ (\rho_o' \mu_o' / \rho_w \mu_w) \ (H_o - H_w) \ (du/dx)$$

where

$$\mu = \frac{0.0232 \times 10^{-6} \text{ T}^{0.5}}{1 + (220/\text{T})}$$

and

$$\frac{du}{dx} = (1/r_s) [2RT (1 - P_{\infty}/P_0^*)]^{0.5}$$

Local heat transfer coefficient for each thermocouple was computed by:

$$h_{local} = \frac{\dot{q}}{r T_0 - T_W}$$

at 
$$r = 1.0, 0.85$$

The ratio of the local heat-transfer coefficient to the reference heat-transfer coefficient was computed by:

DATA REDUCTION (Concluded)

$$\frac{h}{h_{ref}}$$

where

$$h_{ref} = \frac{\dot{q}_{ot}}{rT_o - T_w}$$

at r = 1.0

The ratio of the heat-transfer coefficient at adiabatic wall-temperature ratios of r = 0.85, 1.0 to the reference heat-transfer coefficients was computed.

TABLE I.

OH13			DATE: June 1973
	TEST CO	INDITIONS	•
			1.
MACH NUMBER		STAGNATION PRESSURE	STAGNATION TEMPERATUR (degrees Fahrenheit)
MINOT NOMBER	(per unit length)	(pounds/sq. inch)	(degrees ramement)
8.0	1 x 10 <sup>6</sup> /ft	200	775
8.0	2 x 10 <sup>6</sup> /ft	450	850
8.0	$3 \times 10^{6}/ft$	700	925
8.0	$4 \times 10^{6}/\text{ft}$	950	900
8.0	$6 \times 10^{6}/ft$	1470	970
			·
		!	
	,		·
	None		
BALANCE UTILIZED:	None		COEFFICIENT
	CAPACITY:	ACCURACY:	TOLERANCE:
WE	•		
NF	**************************************		
SF	•		
AF			
PM			
RM			
YM			
COMMENTS: Model	instrumented with	57 iron-constanta	n thermocouples.

TABLE II.

,			-				TE	ST	RUN	NUM	BER	<b>S</b>								5 76	न	<u>}</u>	
SEN																				^	HAWAHT	MACH	
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13

DATA SETS REFER TO LOWER FUSELAGE SURFACE.
DATA SETS REFER TO ORBITER CANOPY.
DATA SETS REFER TO LOWER WING SURFACE.

## : TABLE III. - MODEL DIMENSIONAL DATA

MODEL COMPONENT: Body - (B10)		
	<i>\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\\</i>	
GENERAL DESCRIPTION: Fuselege. 24 Config	guration lightweig	ht orbiter.
	l scale = 0.00593	
DRAWING NUMBER V170-000089B	93. 94A	
DIMENSION:	FULL SCALE	MODEL SCALE
Length in	1328.3	7.87682
Max Width in (@ X <sub>0</sub> = 1528.3)	265.0	1.57145
Max Depth in (@ $X_0 = 1480.52$ )	2148.0	1.47064
Fineness Ratio	5.012	5.012
Area FT <sup>2</sup>		/
Max Cross-Sectional	456.4	0.01605
Planform .		
Wetted		
Base		

MODEL COMPONENT:	Orbiter Canopy	(C <sub>5</sub> )		
GENERAL DESCRIPTION:	Orbiter canopy	for lightweight	Model Scrle	= 0.00593
		· · · · · · · · · · · · · · · · · · ·		
DRAWING NUMBER	VL 70 0	00092	_	
DIMENSION:	•	FULL SC	ALE MOD	DEL SCALE
Sta. find bulkhead in.		391.0	2.	31863
Ste. TE - in.	•	560.0		32080
Cenopy body intersection	n, in.	391.0		31863
•	•			
•		را در		
		-	•	

MODEL COMPONENT: Manipulator House	sing (D <sub>7</sub> )	
GENERAL DESCRIPTION: 2A configuration of the series of the 2A configuration of the series of the ser		3
DRAWING NUMBER V170-000	093, SS-A-00092	
DIMENSION:	FULL SCALE	MODEL SCALE
Length in	881.0	5.22433
Max Width - in.	51.0	0.30243
Mox Depth - in.	23.0	0.13639
Fineness Ratio	•	-
Areo		
Max Cross-Sectional		-
Planform		_
Wetted	•	<u>:</u>
Base	•	•
•		

# Location at:

% fuselege BP = 0.0
WP = 500.0 in FS
X<sub>0</sub> 426.0 to X<sub>0</sub> 1307.0 in. FS

MODEL COMPONENT:	Body Flap (F <sub>4</sub> )	
GENERAL DESCRIPTION:	Aft body flap used on lightweight	orbiter
configuration, model		
DRAWING NUMBER	VL-70-000094A, SS-A-0092	
DIMENSION:	FULL SCALE	MODEL SCALE
Length, in.	84.70	0.50227
Max Width, in.	265.00	1.57145
Max Depth	_	-
Fineness Ratio		
Areo, FT <sup>2</sup>		
Max Cross-Section		
Planform	142.64	0.00502
Wetted	_	•
Base	38.65	0.00136

MODEL COMPONENT:	OMS Pod (M3)		
GENERAL DESCRIPTION:	2A Lightweight orb	iter configuration	per Rockwell
•			
lines VL70 000094A.			
	<del></del>		
Model Scrle = 0.00593			
DRAWING NUMBER	VL70 -000094	A. SS-A-00092	
DIMENSION:	•	FULL SCALE	MODEL SCALE
Length .		346.0	2.05178
Max Width		108.0	0.64044
Max Depth	·	113.8	0.67483
Fineness Ratio			
Area			
Max Cross-Sec	tional	-	
Planform	•		
Wetted		-	••
Bose		_	

% of OMS pod
Zo = 463.9 in FS: WP400 + 63.9 = 463.9 in FS
Yo = 80.0 in FS
Length: Xo 1214.0 to Xo 1560.0 = 346.0 in FS

TABLE III Cont	inued.	
MODEL COMPONENT: WING (Way) Lightweight Orbiter	الله الله الله الله الله الله الله الله	
GENERAL DESCRIPTION: Orbiter configuration per Ro	ockwell lines VL7	0-00093.
Model Scale = 0.00593.		
Note: Dihedral angle is defined at the	lower surface of	the wing st
the 75.33 percent element line pro	ojected into a pl	ene perpendicul
TEST NO.	DWG. NO. VL70	0-000093
DIMENSIONS:	FULL-SCALE	MODEL SCALE
TOTAL DATA  Area (Theo.) Ft <sup>2</sup> Planform  Spar. (Theo In.	2690.0 936.682 2.265	0.09459 5.55452 2.265
Aspect Ratio Rate of Taper Taper Ratio Dihedral Angle, degrees Incidence Angle, degrees Aerodynamic Twist, degrees	1.177 0.200 3.500 3.000 +3.000	1.177 0.200 3.500 3.000 +3.000
Sweep Back Angles, degrees Leading Edge Trailing Edge 0.25 Element Line Chords: -in.	45.000 -10.24 35.209	45.000 -10.24 35.209
Root (Theo) B.P.O.O. Tip, (Theo) B.P. MAC Fus. Sta. of .25 MAC W.P. of .25 MAC B.L. of .25 MAC	689.24 137.85 474.81 1136.89 299.20 182.13	4.08919 0.81745 2.81562 6.74176 1.77426 1.08003
EXPOSED DATA  Area (Theo) Ft <sup>2</sup> Span, (Theo) In. BP108  Aspect Ratio  Taper Ratio	1752.29 720.68 2.058 0.2451	0.06162 4.27363 2.059 0.2451
Chords Root BP108 Tip 1.00 b  MAC Fus. Sta. of .25 MAC W.P. of .25 MAC B.L. of .25 MAC	562.40 137.85 393.03 1185.31 300.20 143.76	3.33503 0.81745 2.33067 7.02889 1.78019 0.85250
Airfoil Section (Rockwell Mod NASA)  XXXX-64 $T/C\Theta$ Root b = 0.485 $T/C\Theta$ Tip b = 1.00	0.10	0.10
Data for (1) of (2) Sides  Leading Edge Cuff Planform Area Ft <sup>2</sup> Leading Edge Intersects Fus M. L. @ Sta  Leading Edge Intersects Wing @ Sta	120.33 560.0 1035.0	0.00423 3.32080 6.13755

## TABLE III. - Concluded.

MODEL COMPONENT: VERTICAL (V5)			
GENERAL DESCRIPTION: Centerling rounded lending edge.	e vertical ta:	il double wedge Firioi	1 WICE
Model Scrle = 0.00593			<del></del>
DRAWING NUMBER:	VL 70-000095.	SS-A-00092	
DIMENSIONS:		FULL-SCALE	MODEL SCALE
Area (Theo) Ft <sup>2</sup> Planform Span (Theo) In Aspect Ratio Rate of Taper Taper Ratio Sweep Back Angles, degrees Leading Edge Trailing Edge 0.25 Element Line Chords: Root (Theo) WP Tip (Theo) WP MAC Fus. Sta. of .25 MAC W. P. of .25 MAC Airfoil Section Leading Wedge Angle Trailing Wedge Angle Leading Edge Radius Void Area Blanketed Area	Deg	413.25	0.01453  1.87222 1.675 .507 .404  45.000 26.249 41.130  1.59220 0.62323 1.18487 8.67856 3.76863 0.00 14.92 0.01186 0.00045

Orbiter Thermocouple Location TABLE IV.

T/C Skin * Location   Pemarks   T/C   Skin * Location   Remarks					7						
T/C	Thick!	y - b/2	*/I - */c	Remarks		T/C No.	Thick	y - b/	2×/1 - ×/c	Remark	5
1	.033	Y=.047	.1536	Winds	hield	31	.0315	Y=.415	1.00	Puselage	
2	.031	Y=.047	.1612	1		32	.0315	b/2=.4	0 .225	Wing	
3	.034	Y=.047	.1.594			33	.233		.250		
4	.0375	Y=.213	.1588			34	.0312		.300		
5	.0375	Y=.196	.1657			35	.033		.400		
6	.0335	Y=.178	.1724			36	.0335		.500		
7	.033	Y=0	.0875	Fusel	age	37	.032		.600		
8	.032		.170	J		38	.0315		.730		
9	.031		.125			39	.0315		.800	,	
10	.0305		.150			40	.0310	•	.900		
11	.030		.175			41	.034	b/2=.6	0 .175		
12	.031		.200			42	.032		.200		
13	.0295		.250			43	.031		.300		
14	.0295		.300			44	.033		.400		
15	.0295		.400			45	•032		.500		
16	.0302		.500			46	.032				
17	.0312		.600			47	.0325		.700		
18	.0315		.700			48	.031		.800		
19	.031		.800			49	.0315		.875		
20	.0295		.900			50	.035	b/2=.	.250		
21	.030		1.00			51	.033		.300		_
55	.0305	1	1.025			52	.033		.400		
23	.0289	Y=.415	.350			53	.0315		.500		
24	.0285		-375			54	.032		.600		
25	.0315		.400			55	.032		.700		
26	.0325		.500			56	.0335		.800		
27	.0320	7	.600			57	.033		.850		
28	.0315		.700								
23	.032		.800								
30	.0315		.900		1						

<sup>\*</sup> model scale, in.

x/1=(t/c's i-31) x/c=(t/c's 32 57)

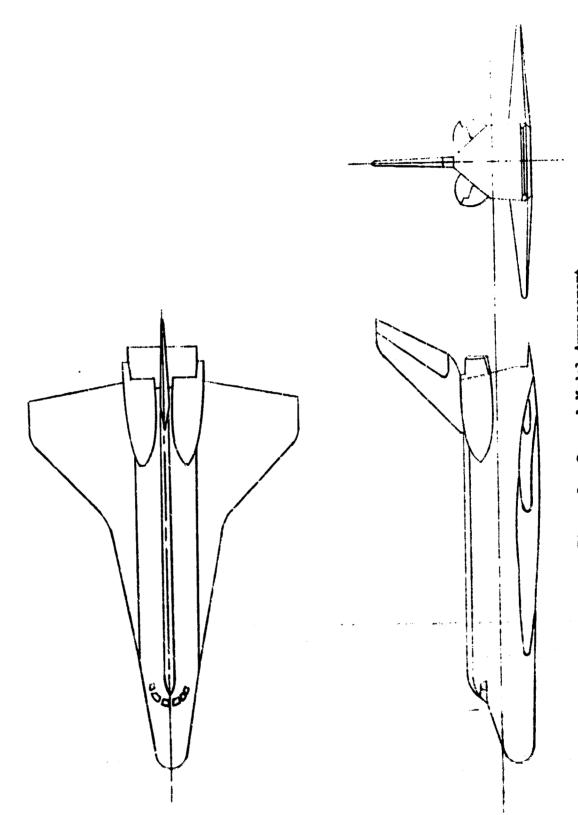
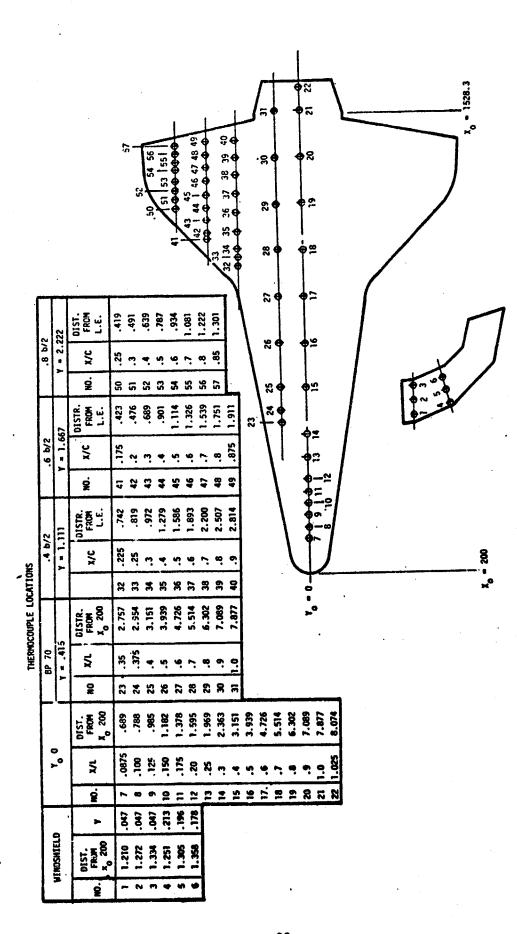


Figure 1. Ceneral Model Arrangement.



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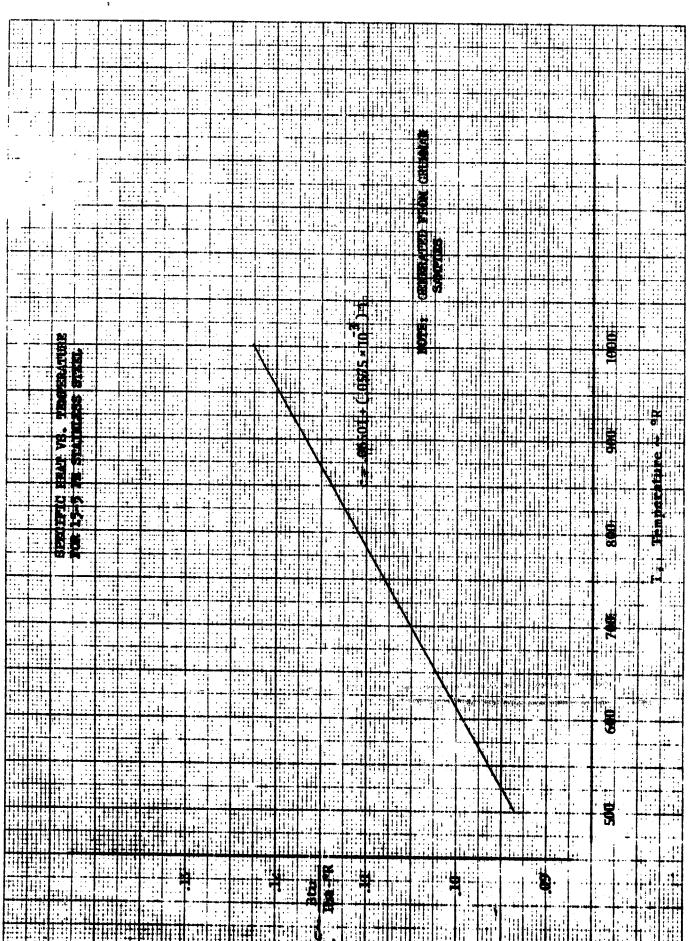


Figure 3. Specific Heat Curve.

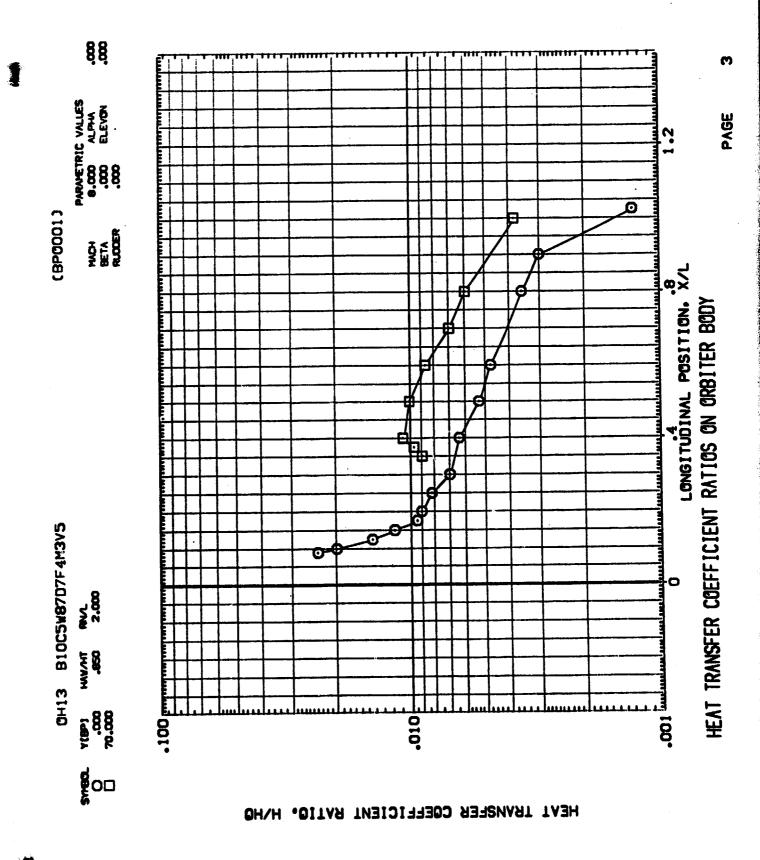
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HEAT TRANSFER COEFFICIENT RATIO. H/H

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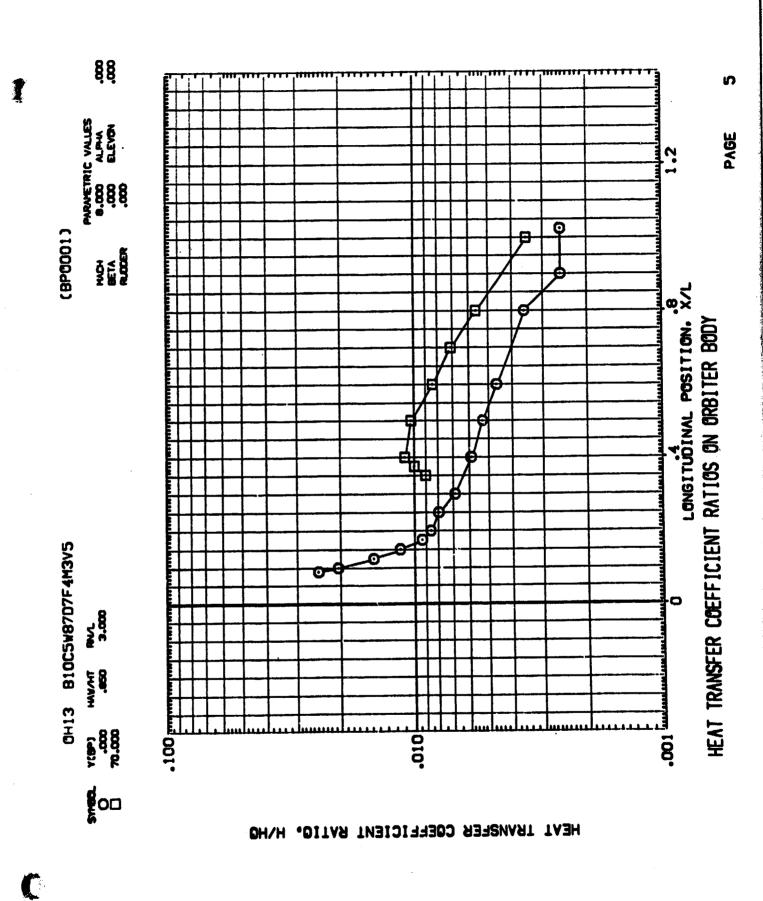
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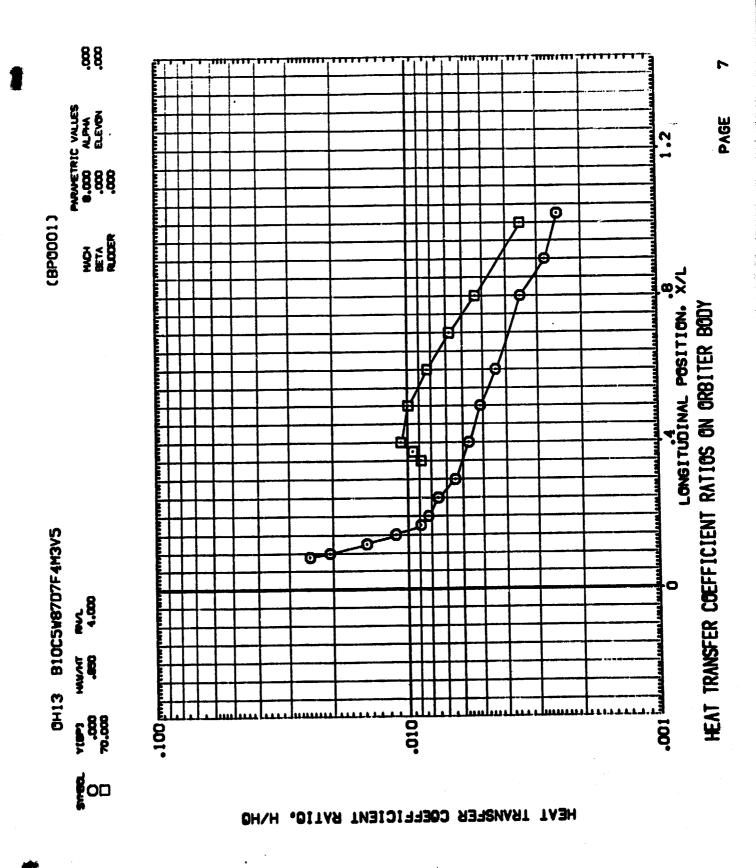
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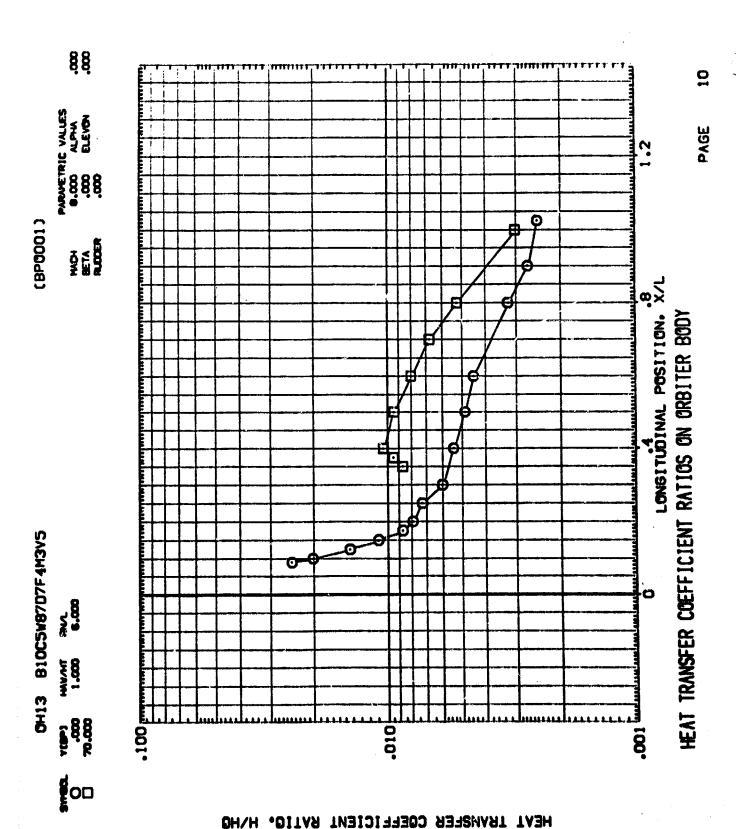




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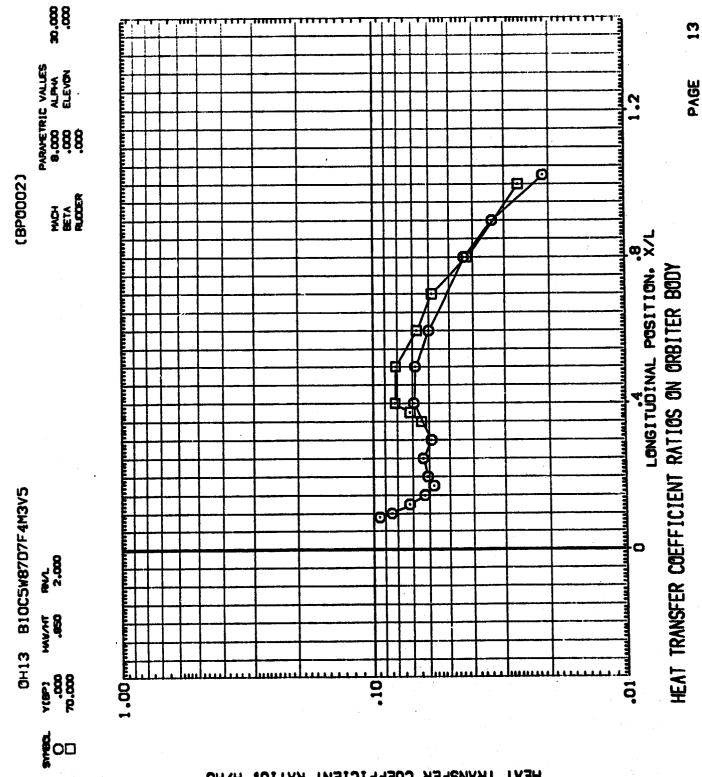


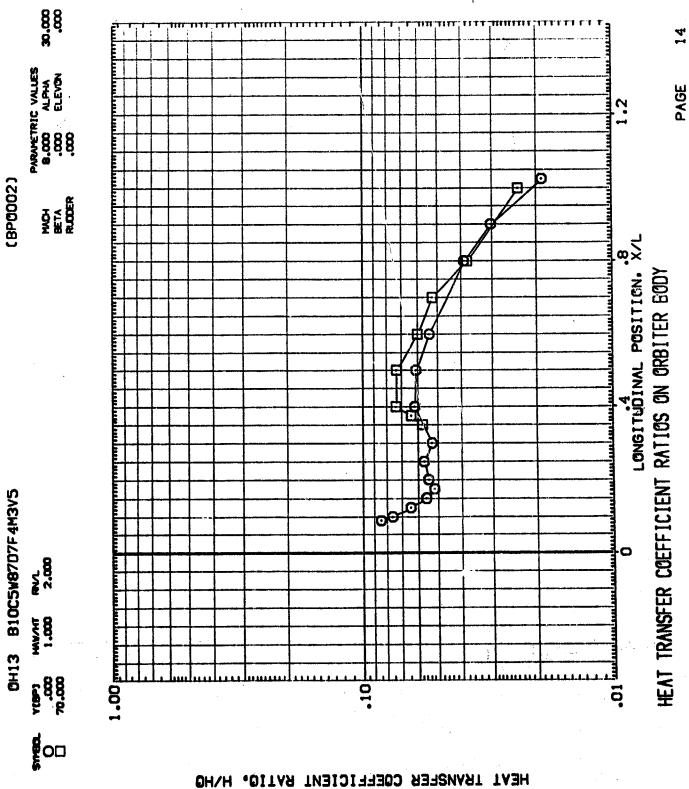




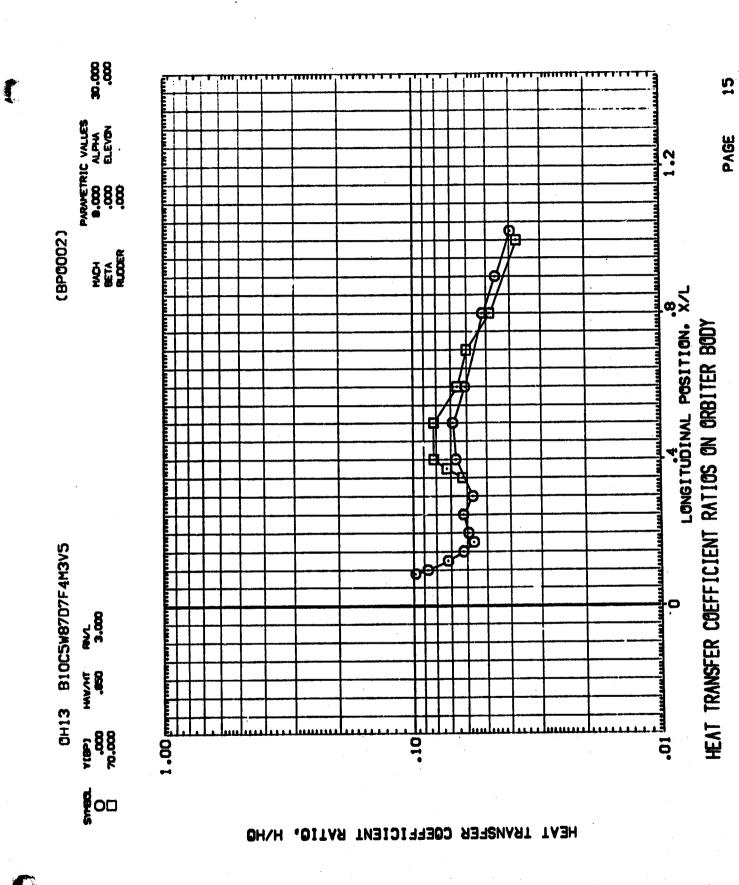
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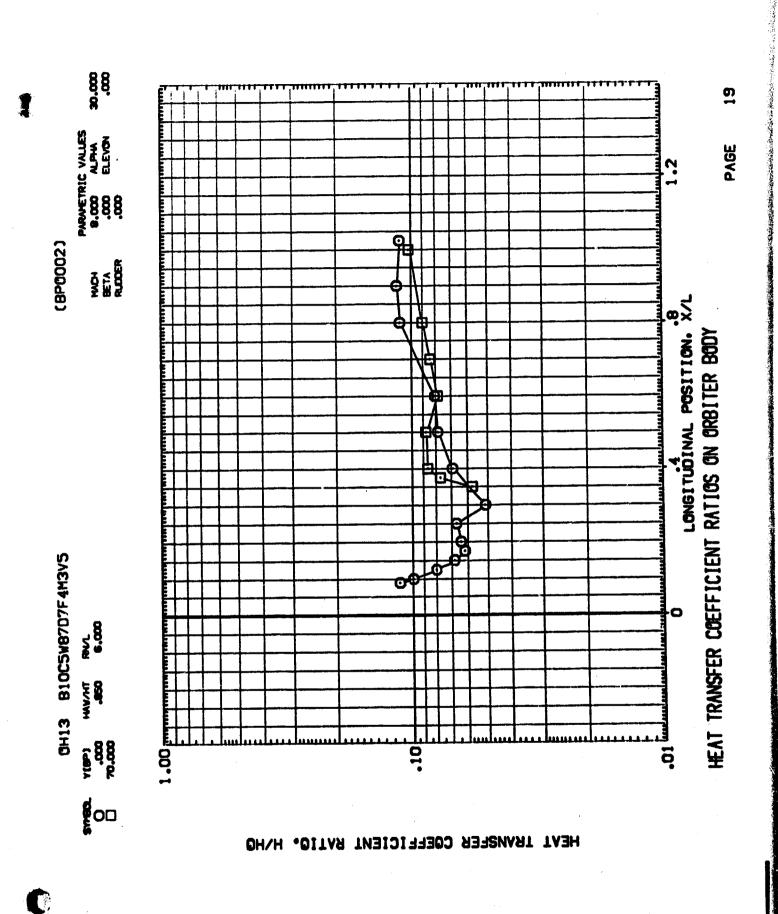
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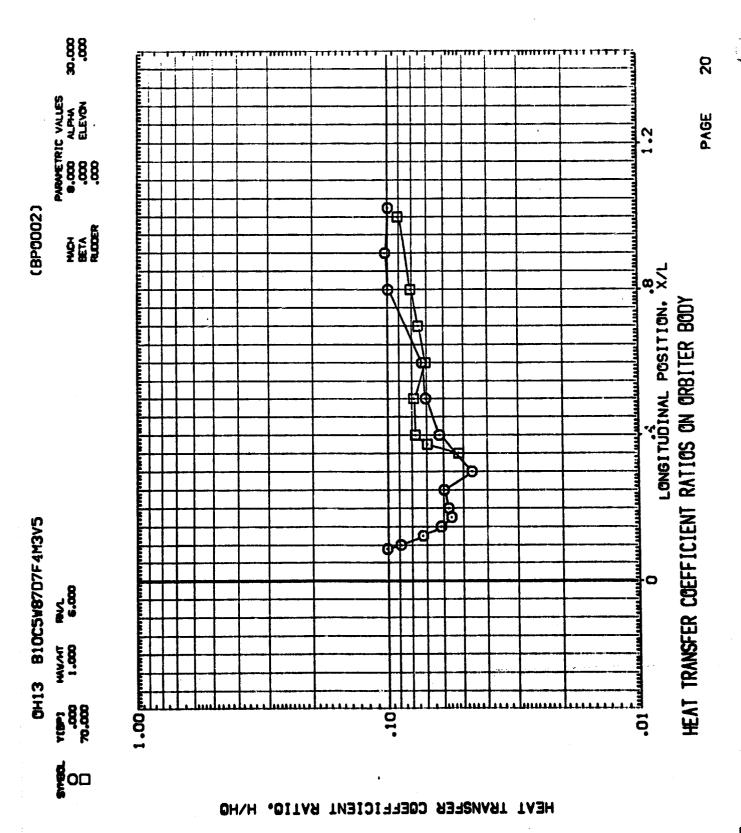
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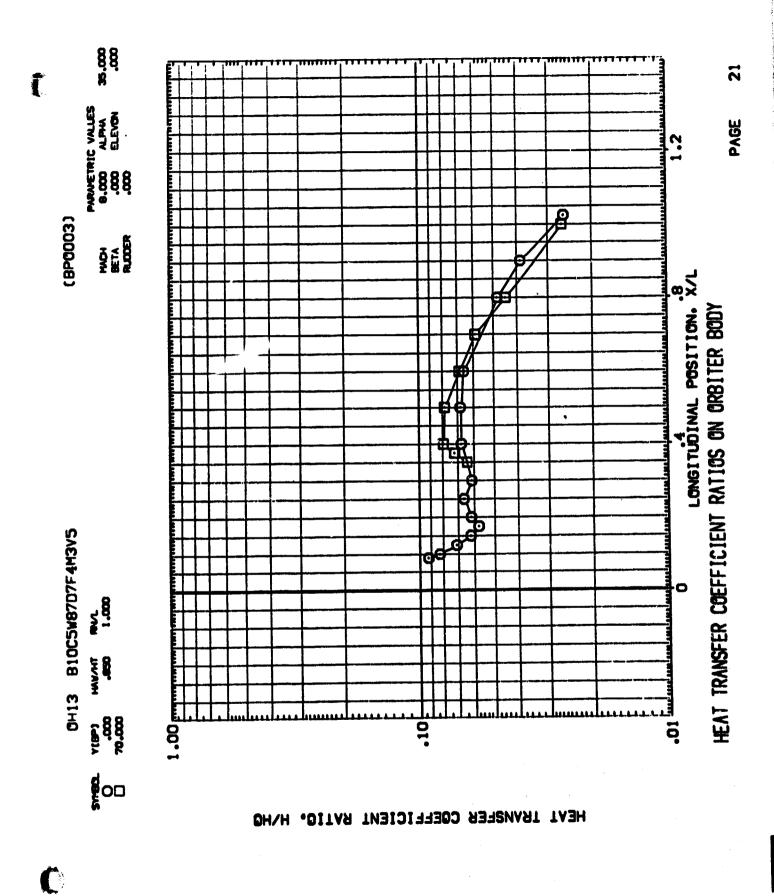
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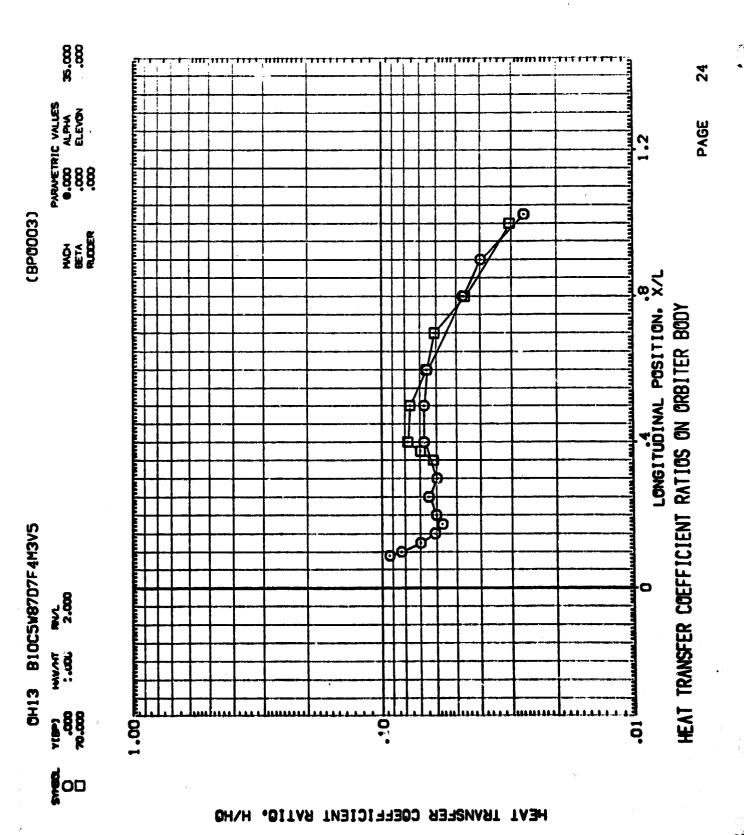






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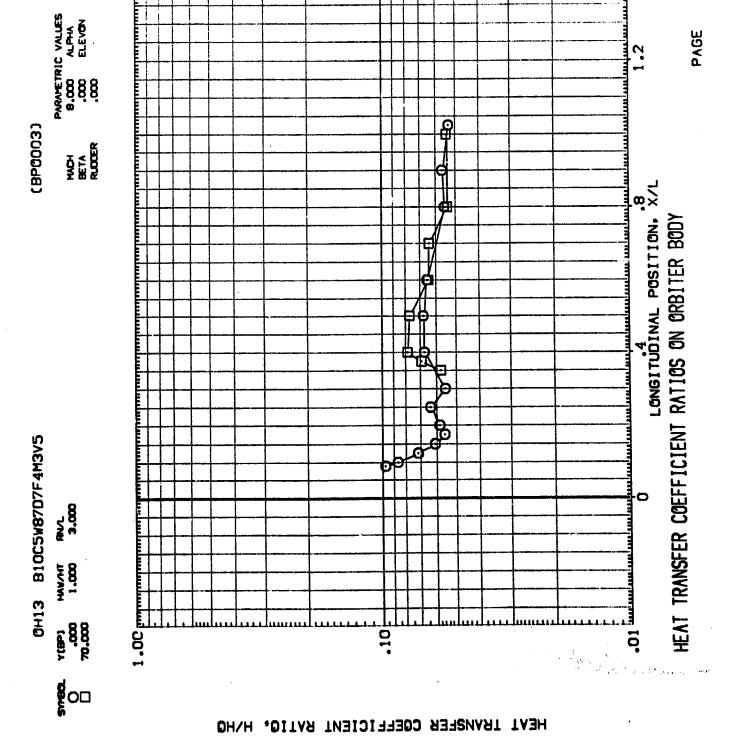
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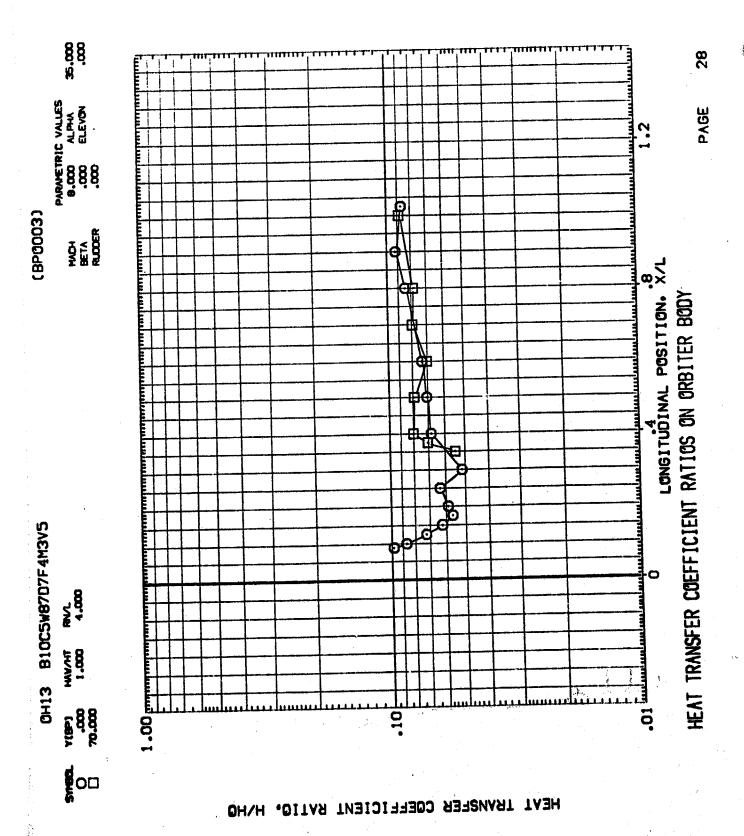


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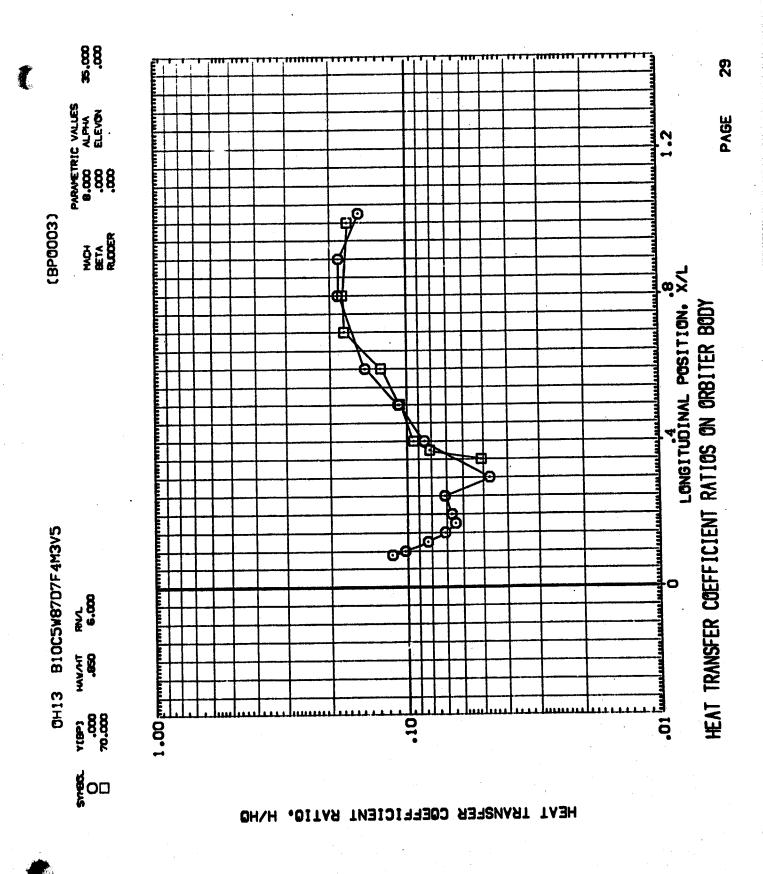
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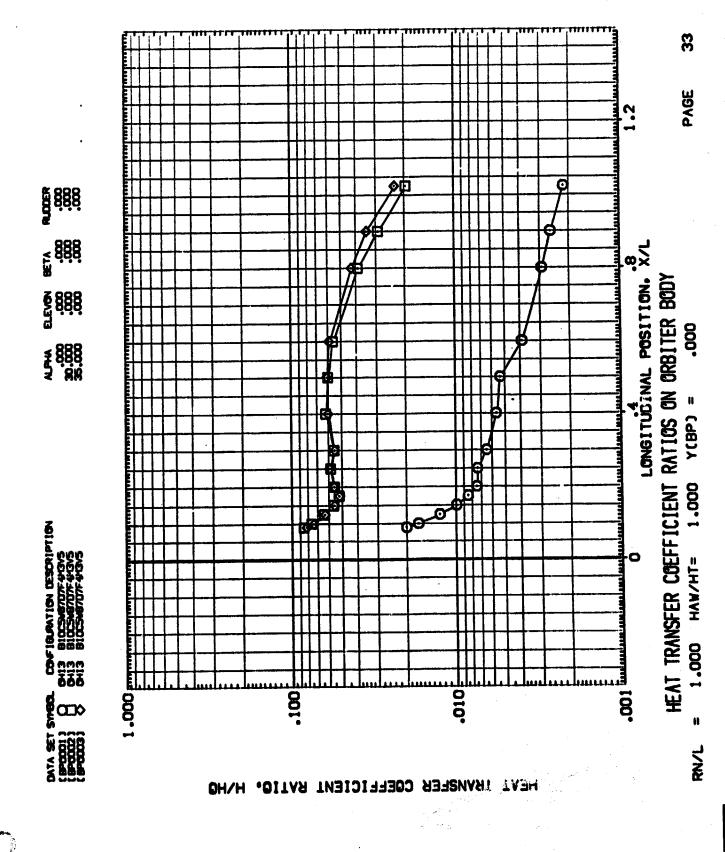
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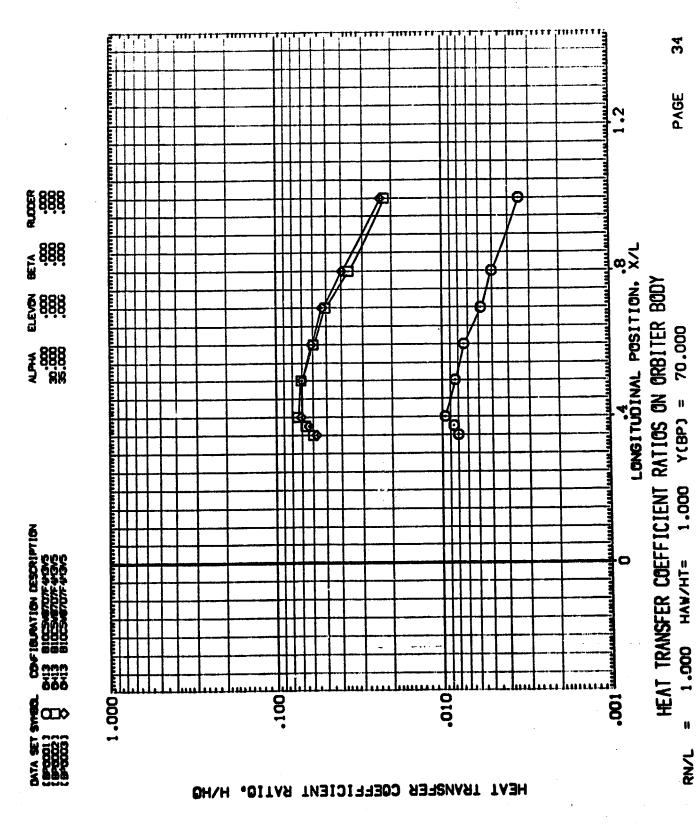
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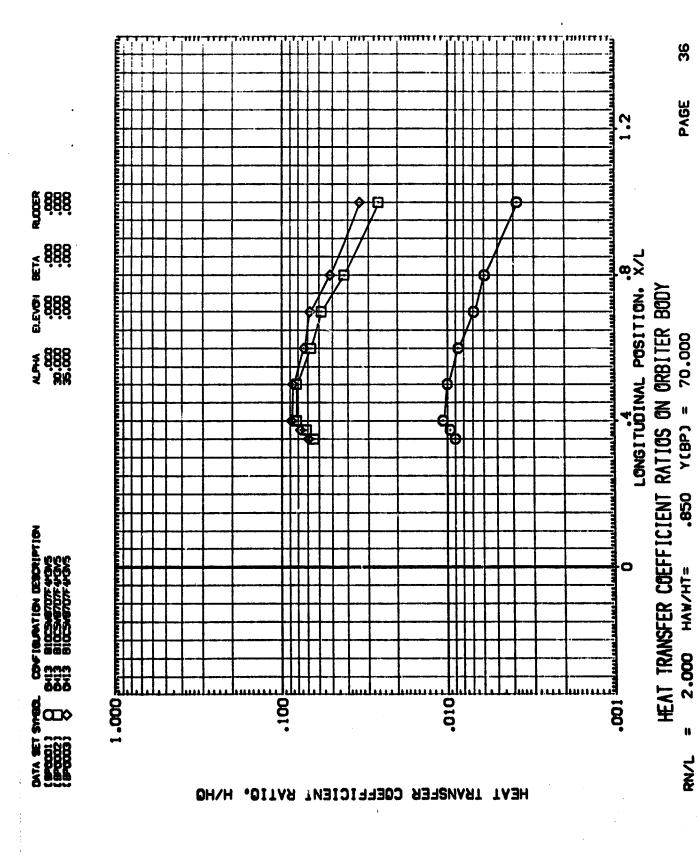




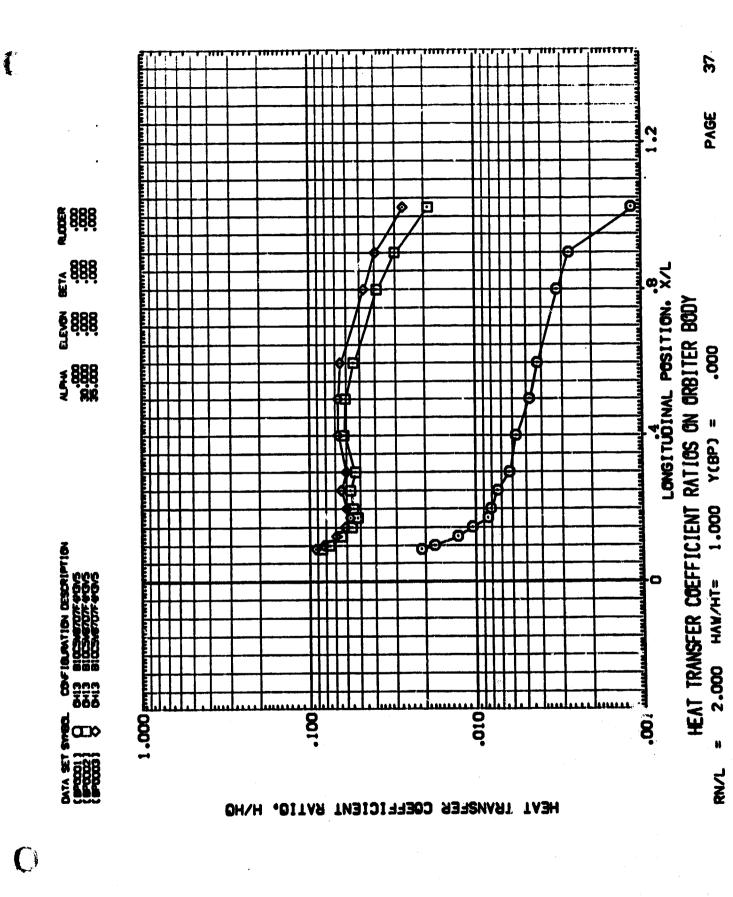


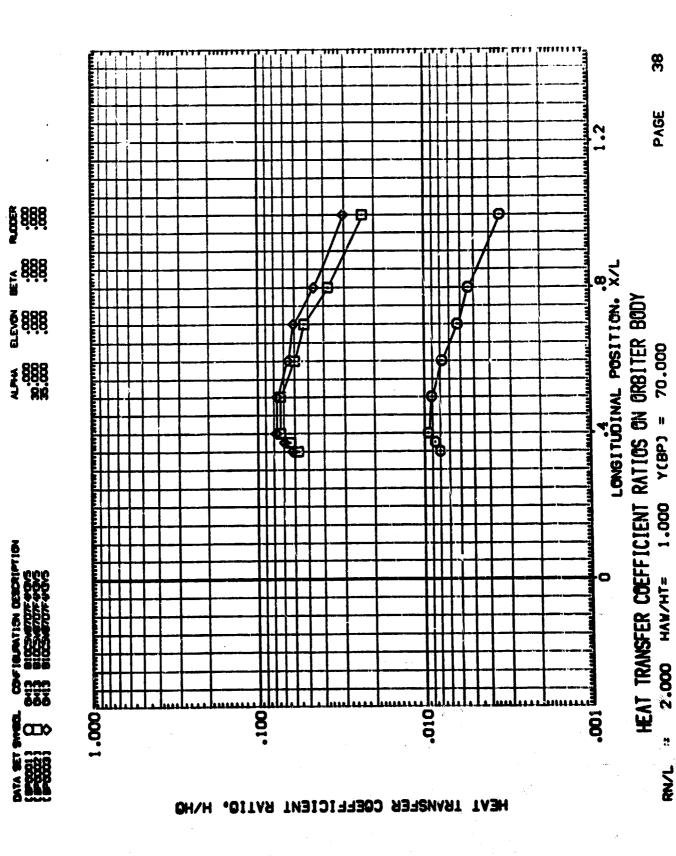
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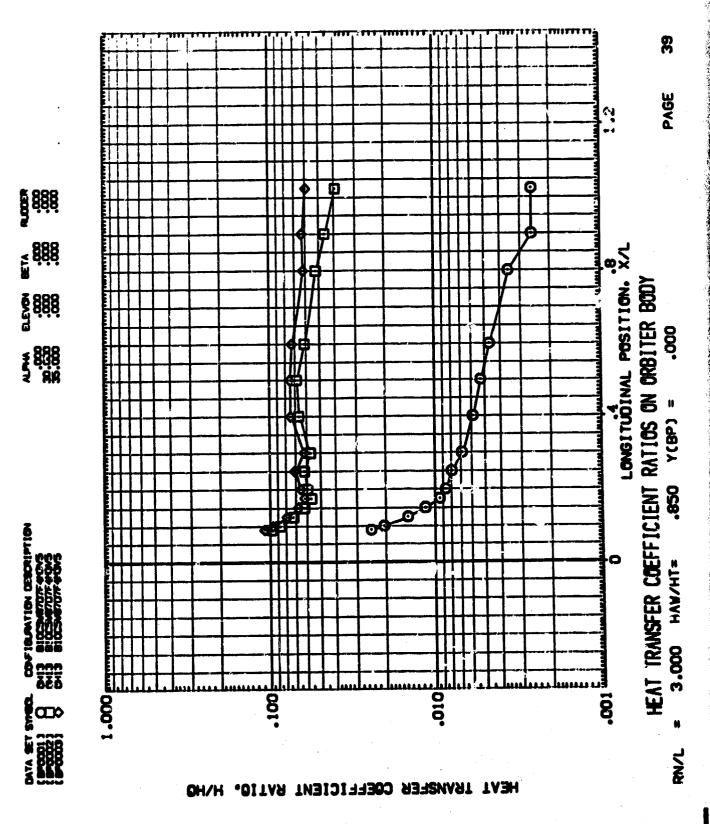


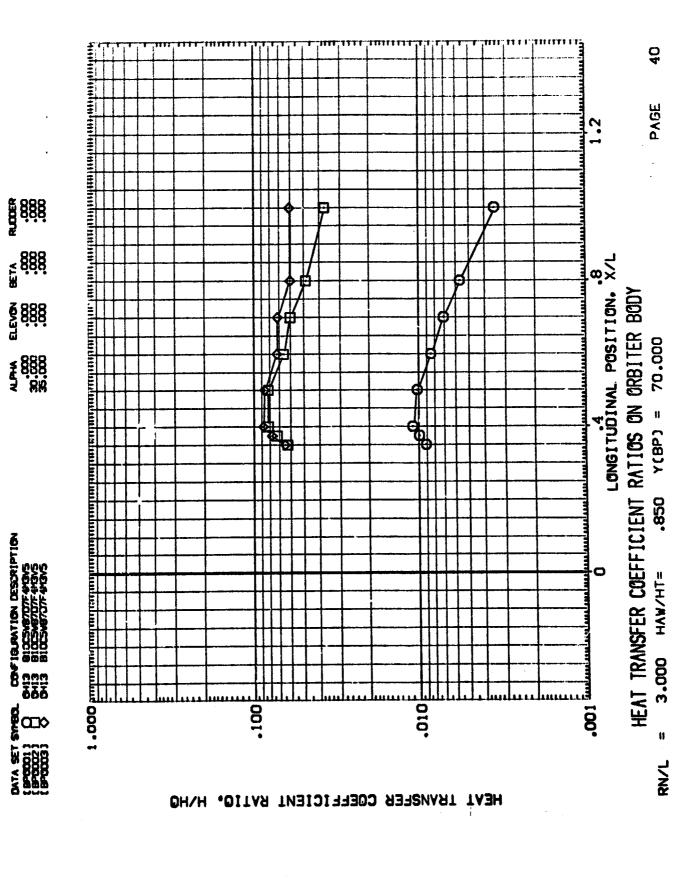


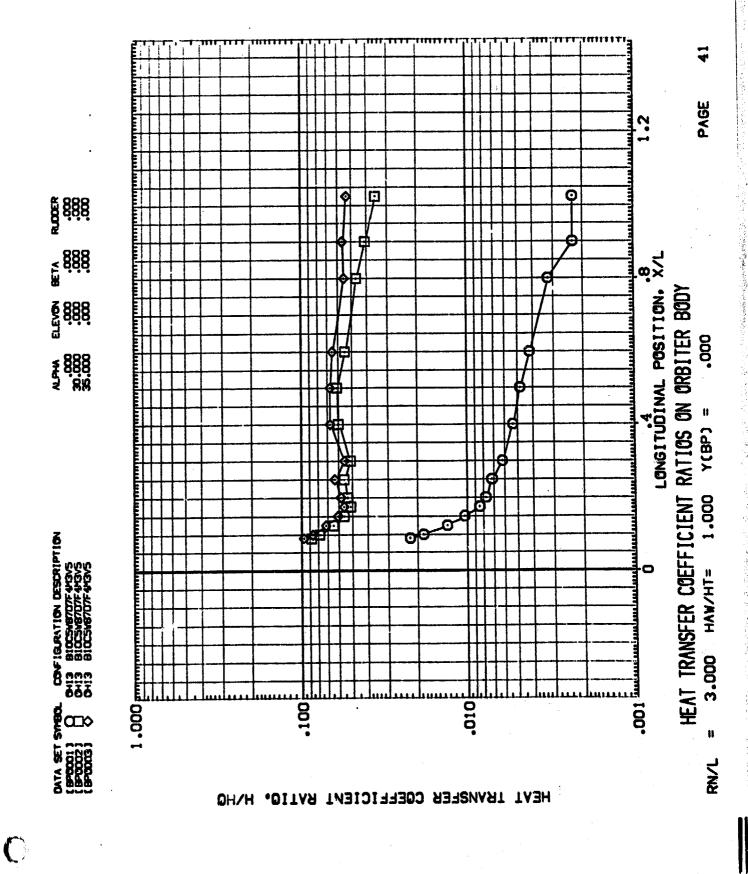




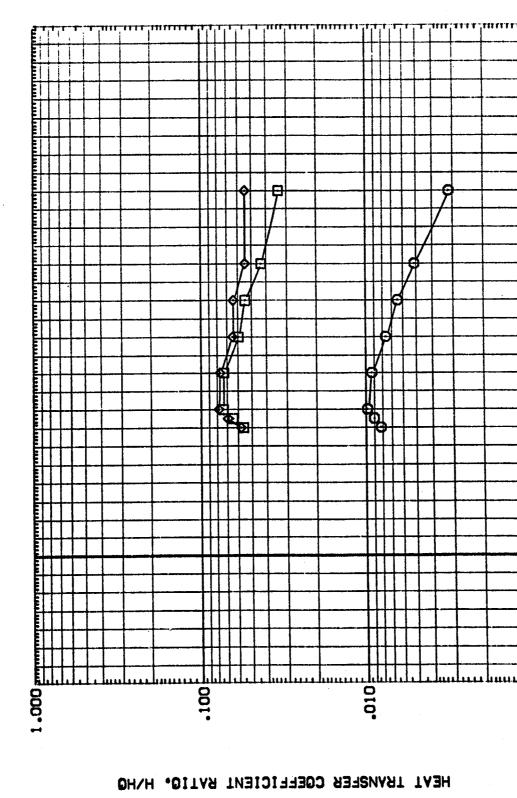








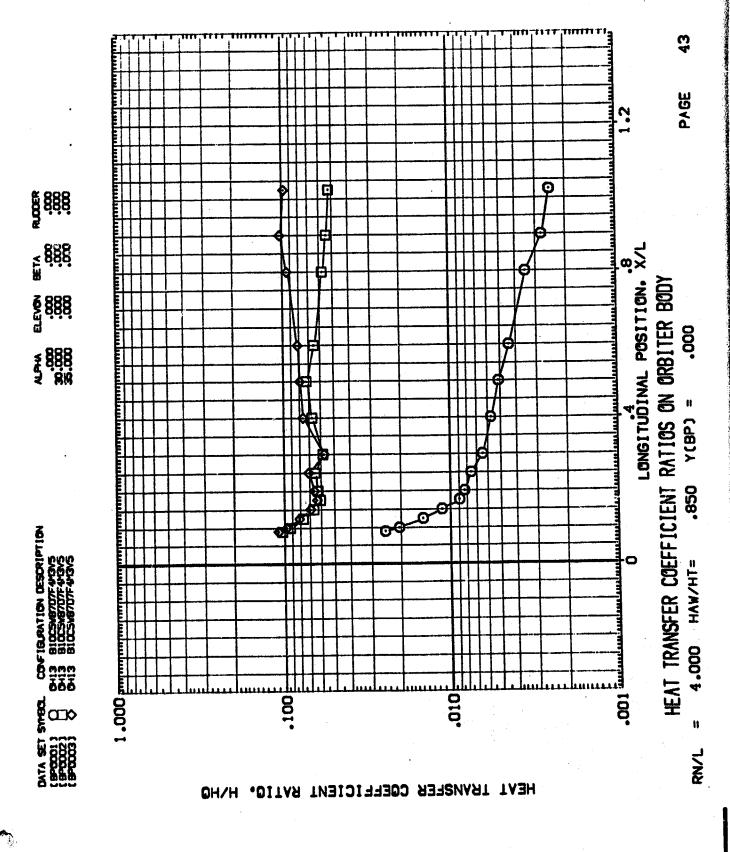
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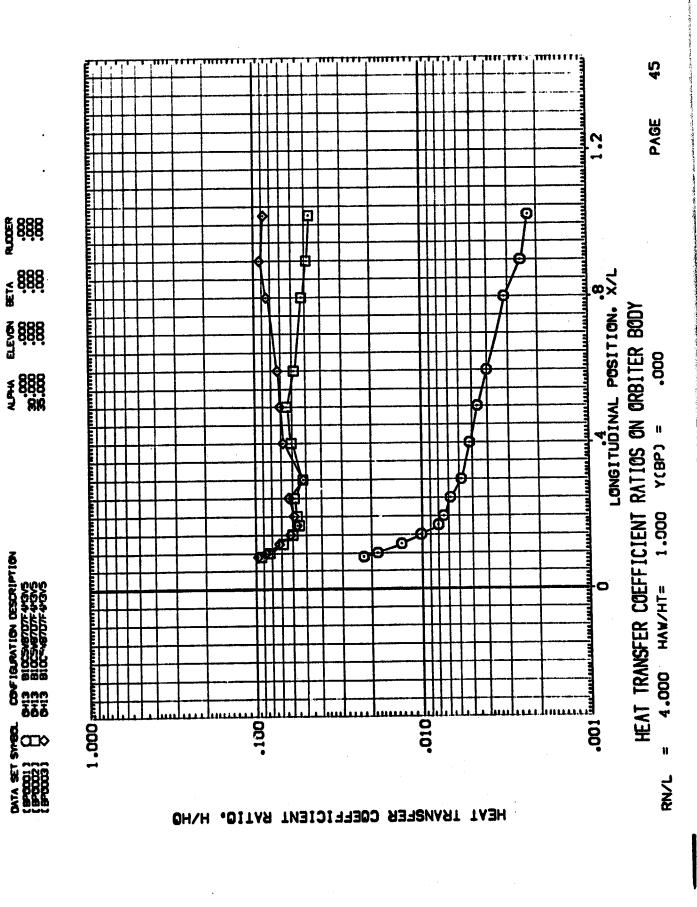
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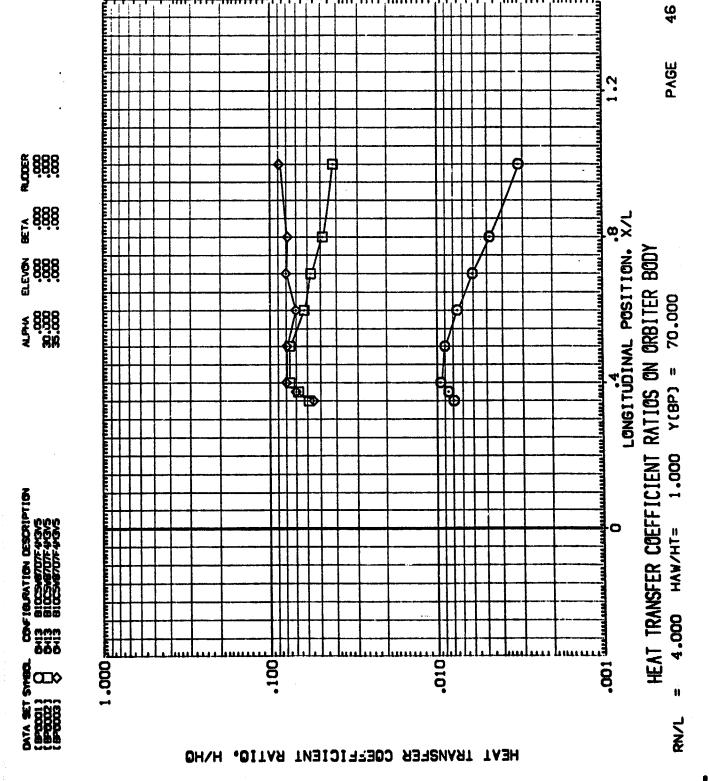
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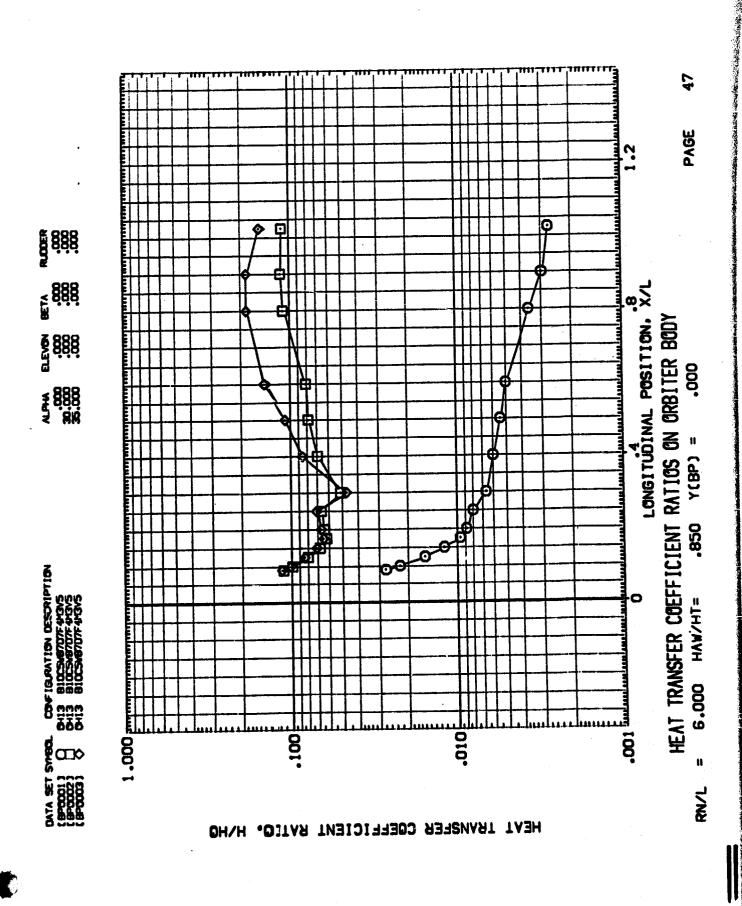
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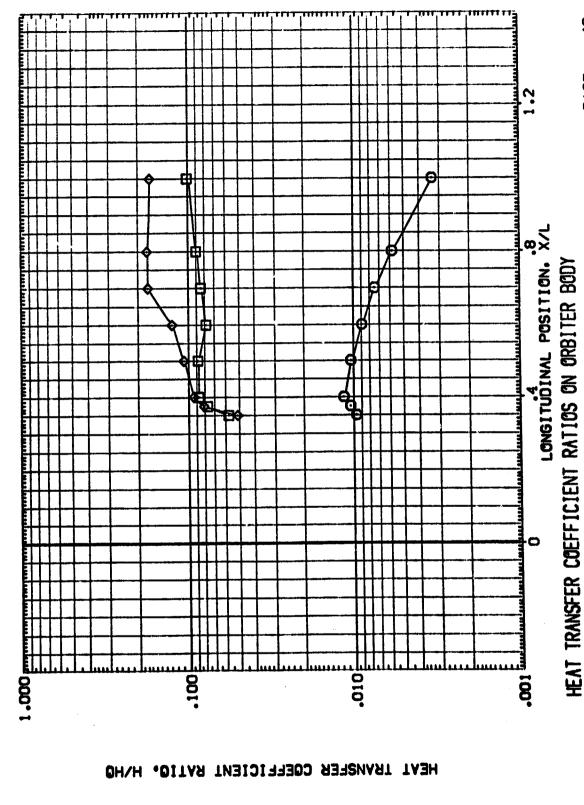


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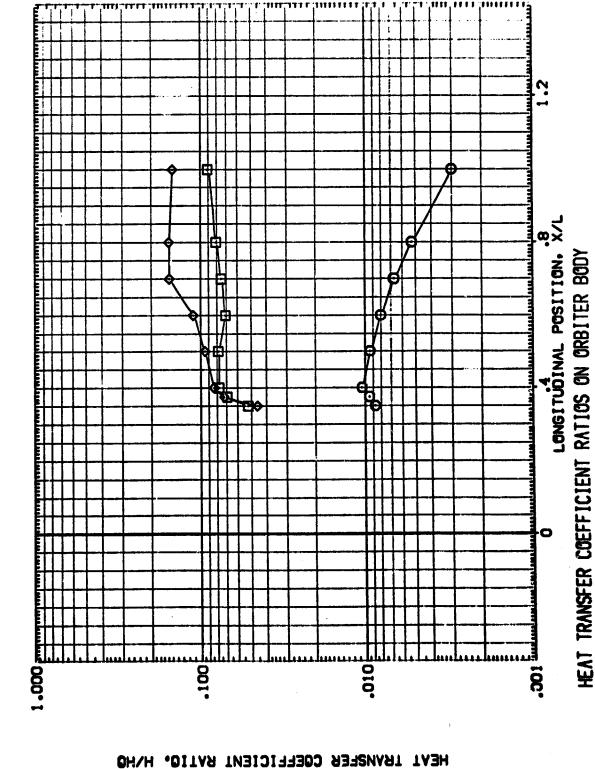
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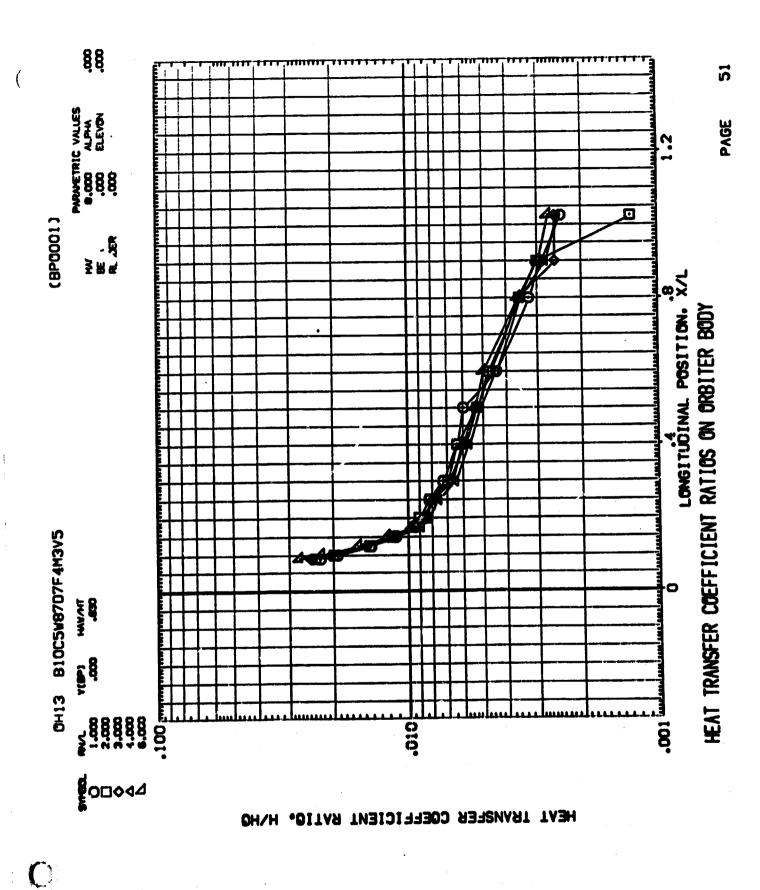
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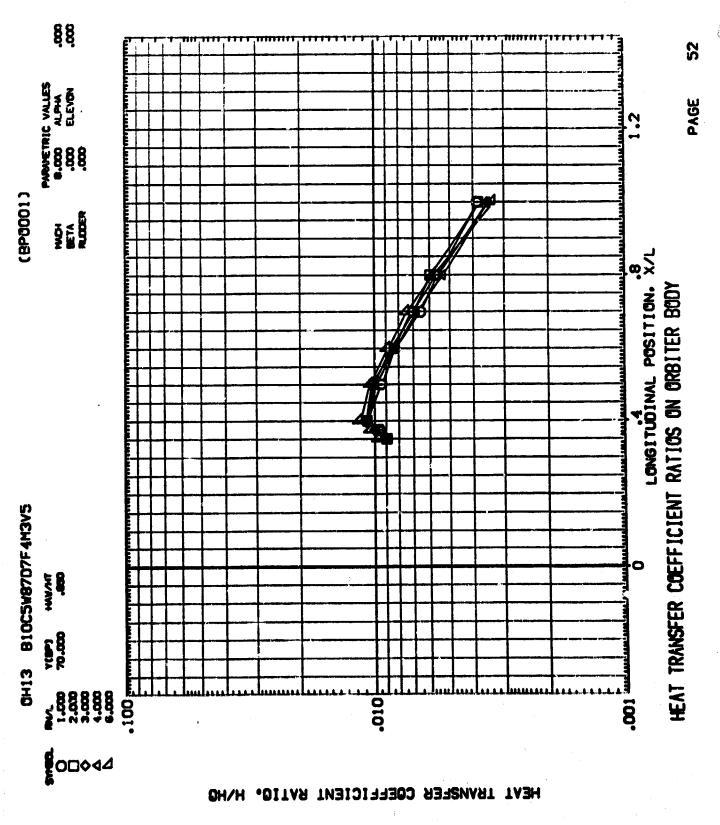
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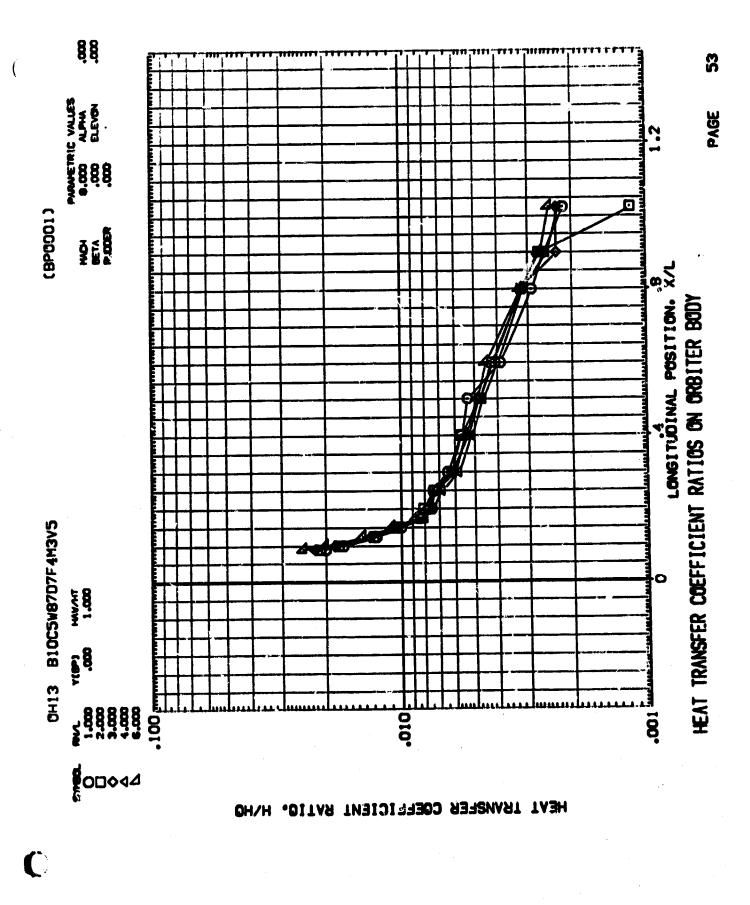
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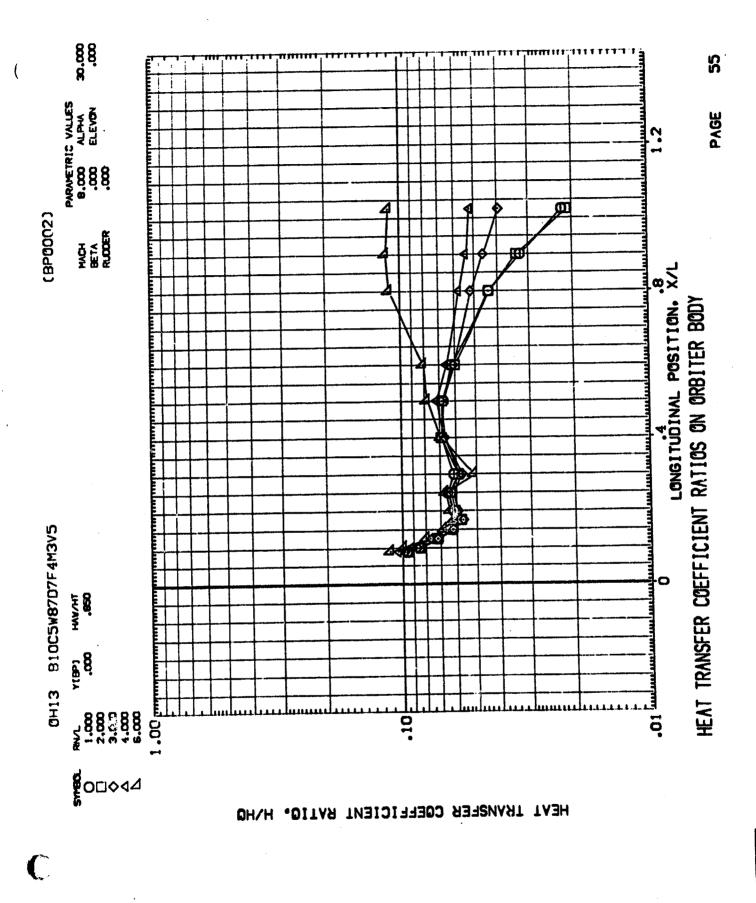




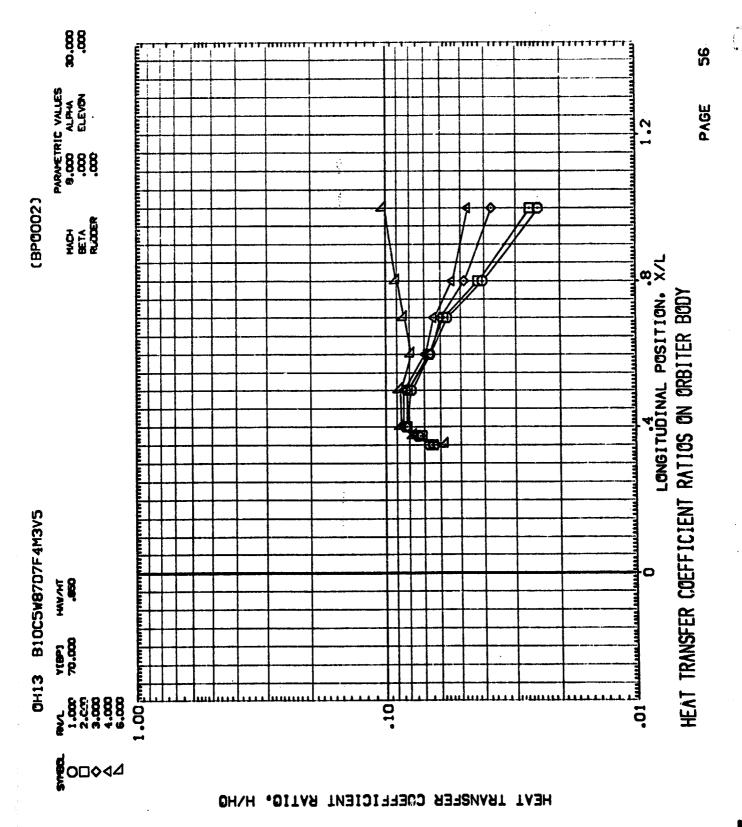
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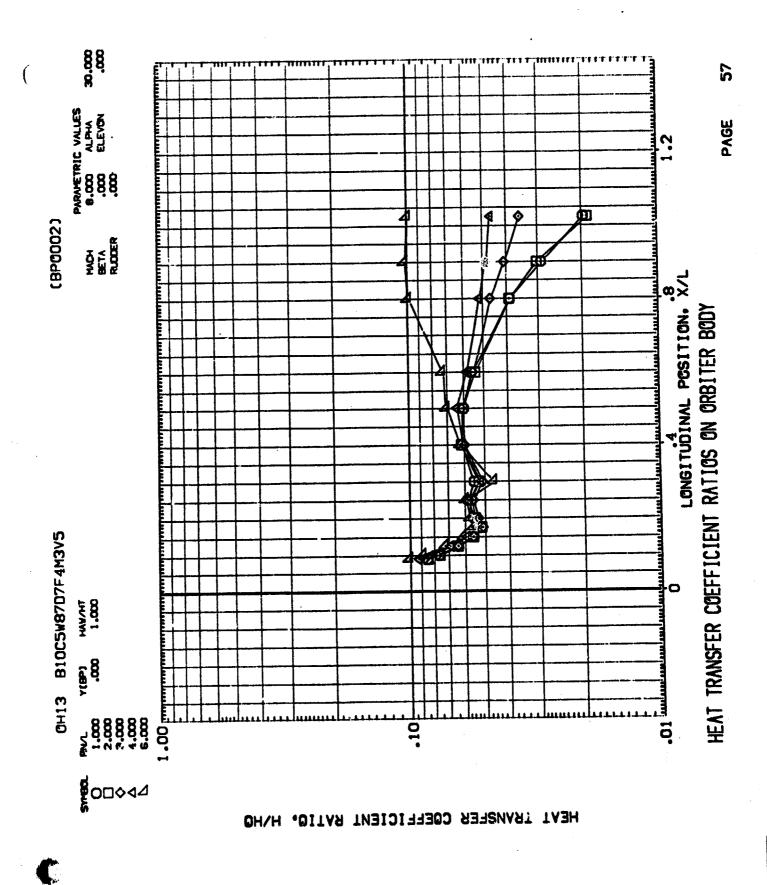
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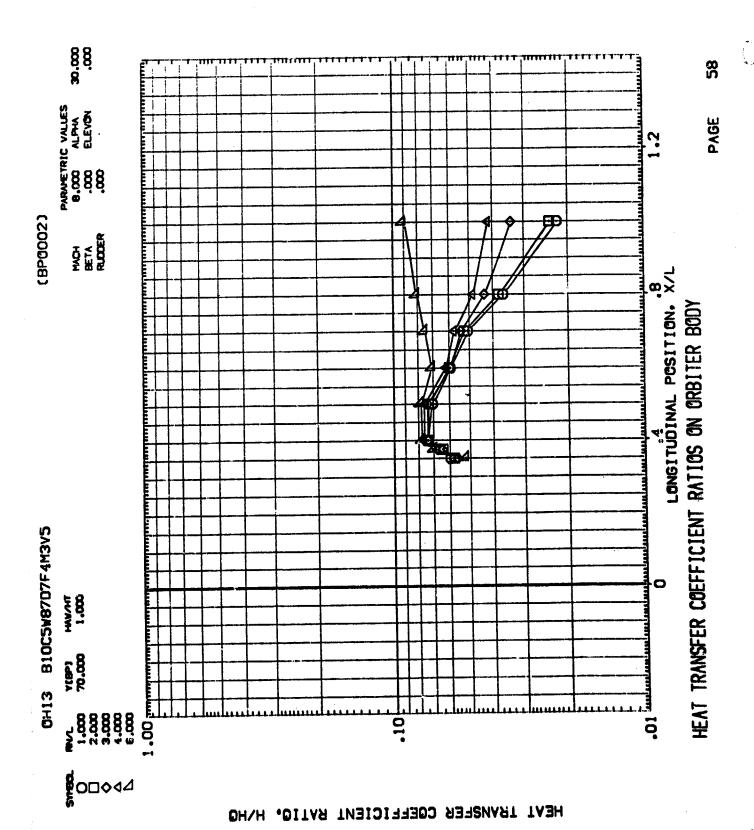
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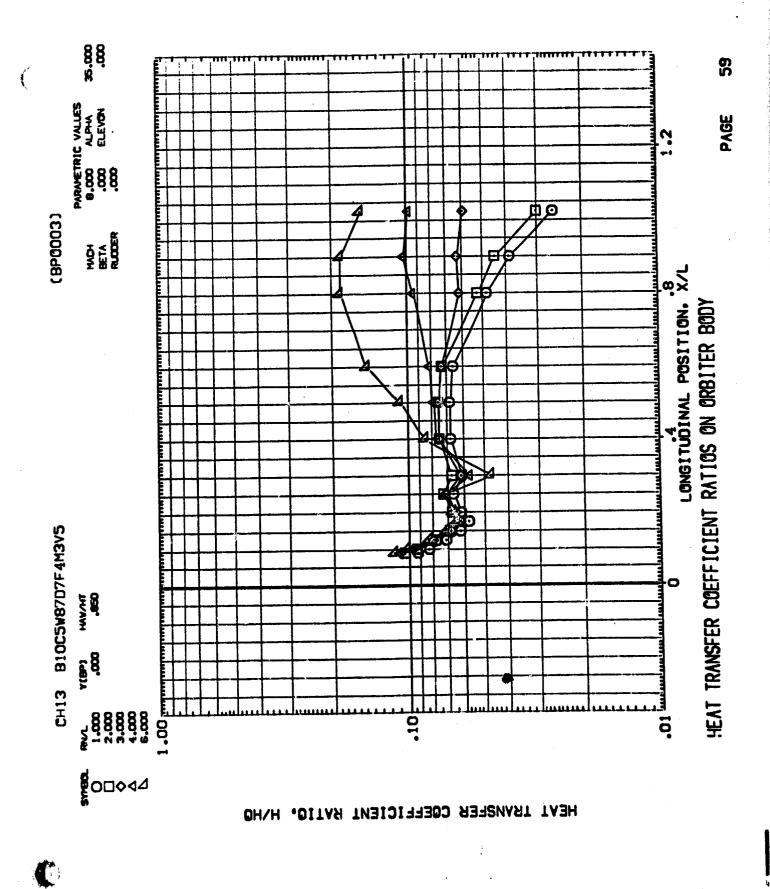


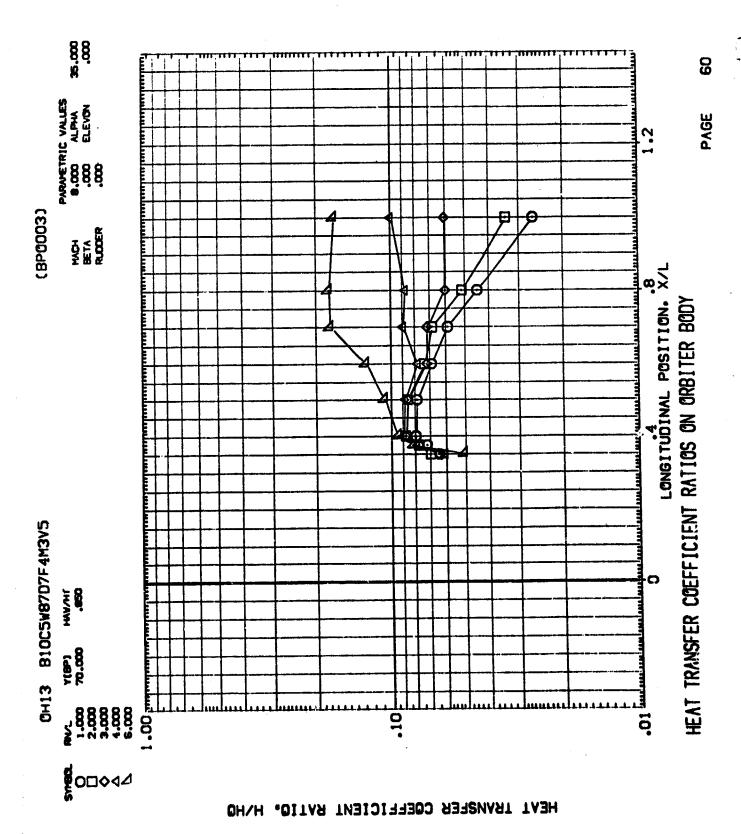


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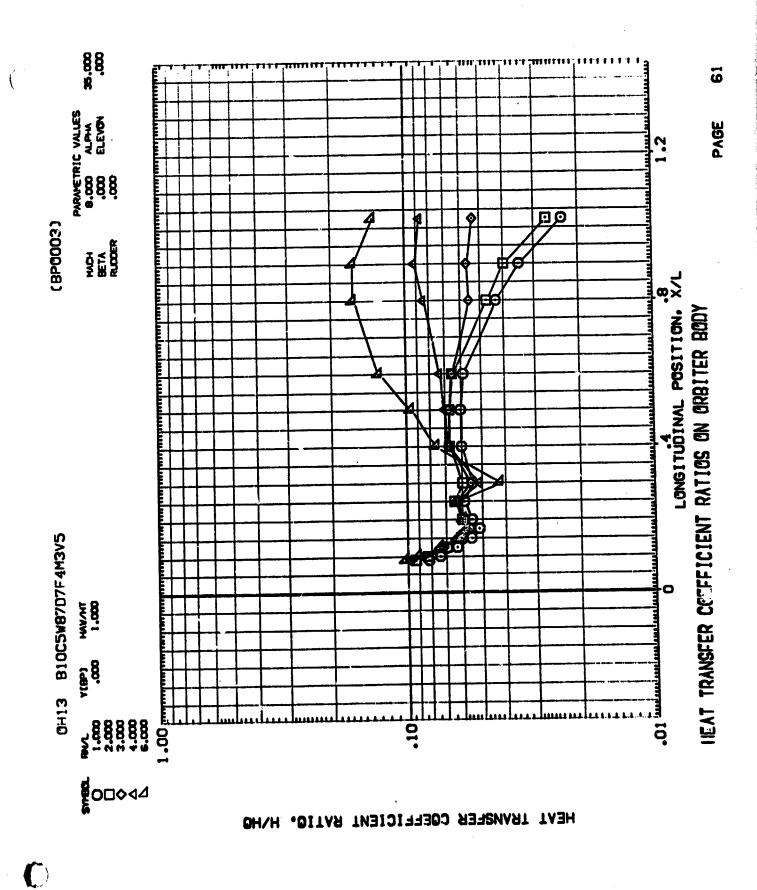


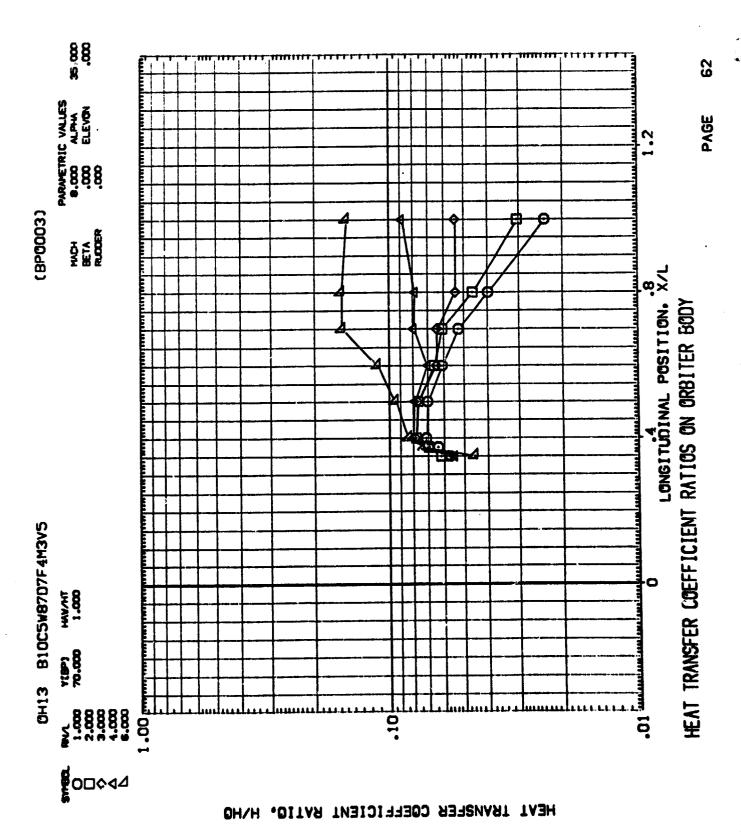




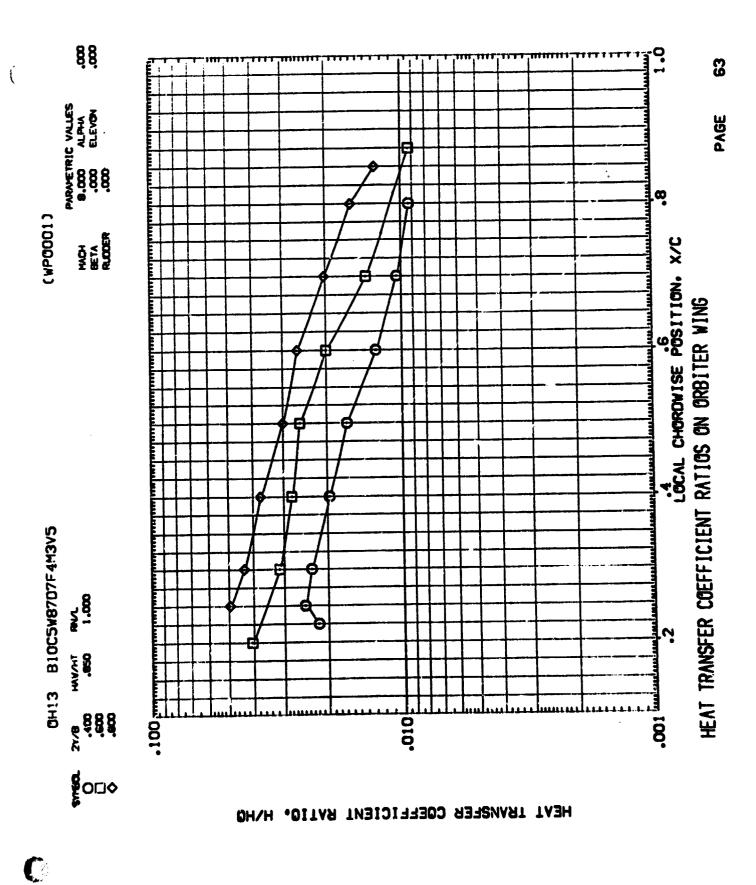


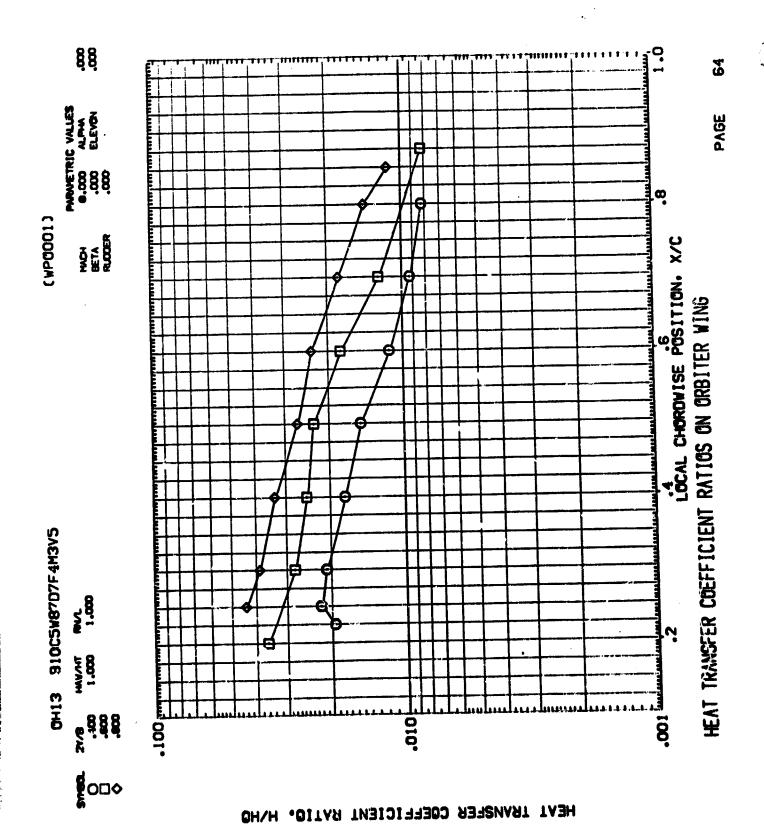




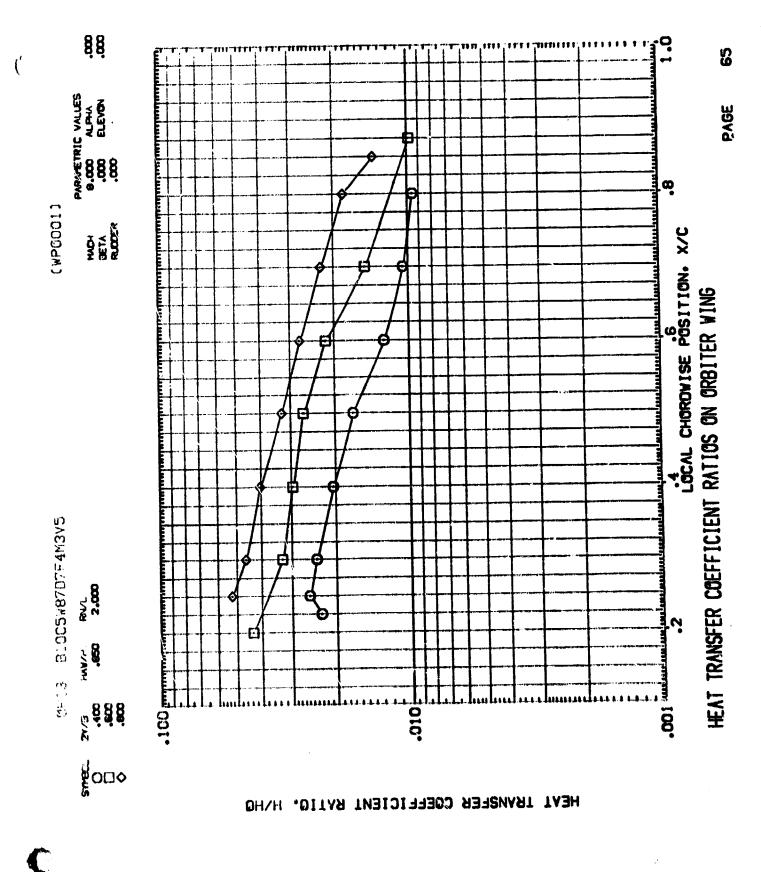




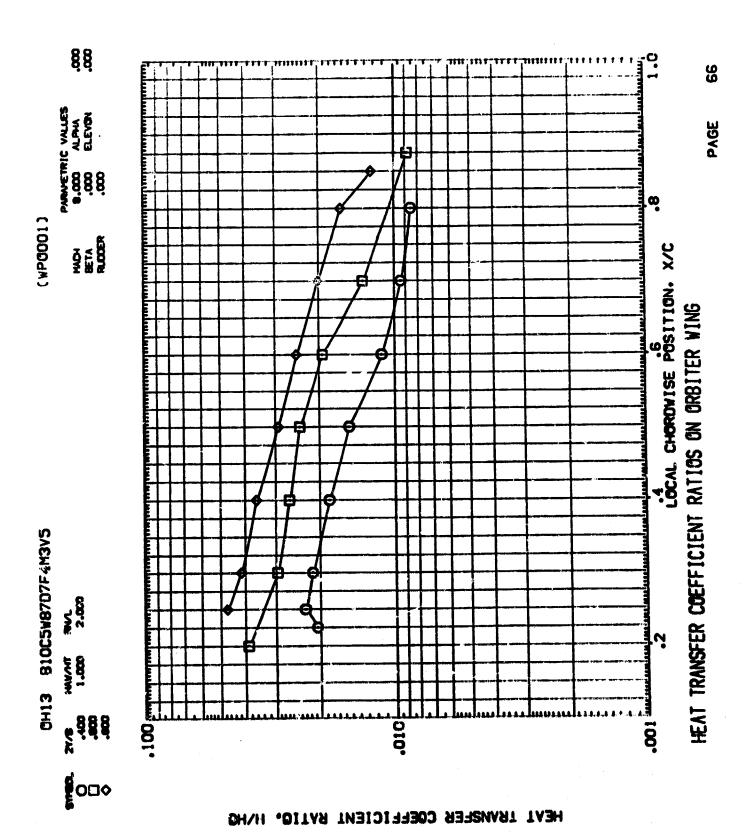




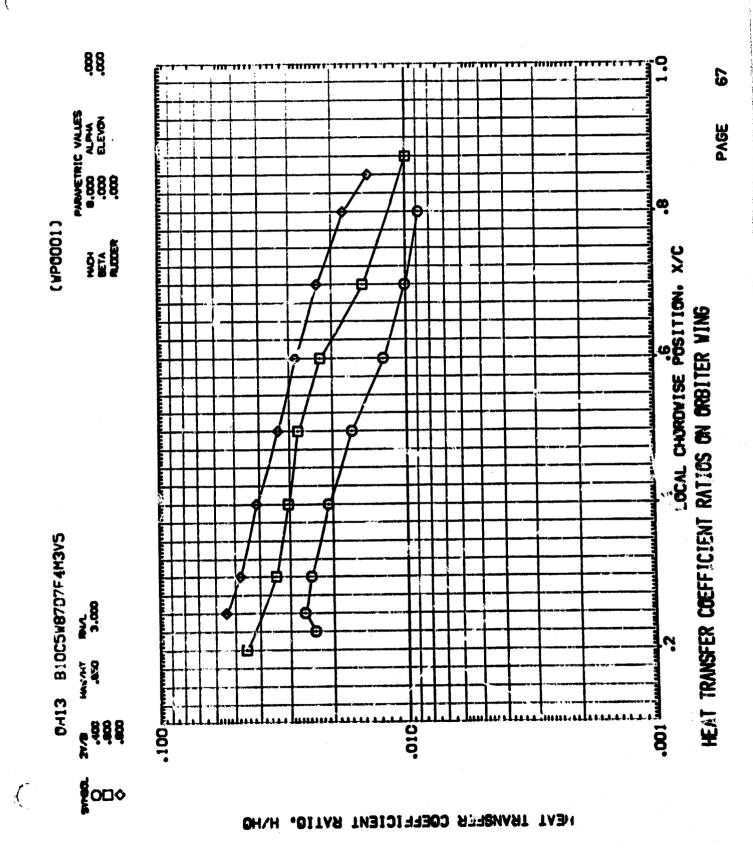


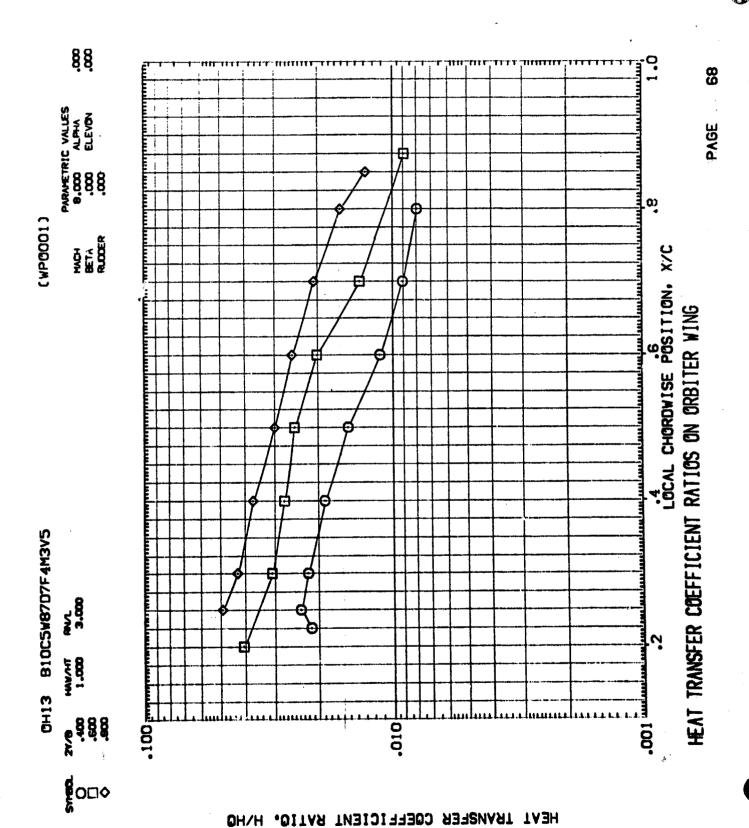


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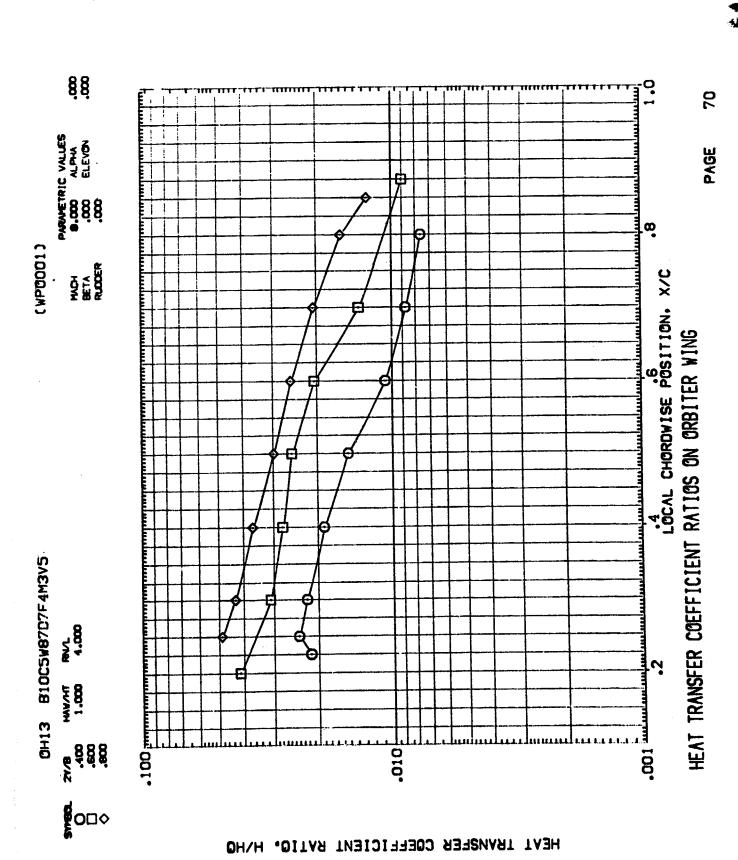








HEAT TRANSFER COEFFICIENT RATIO.





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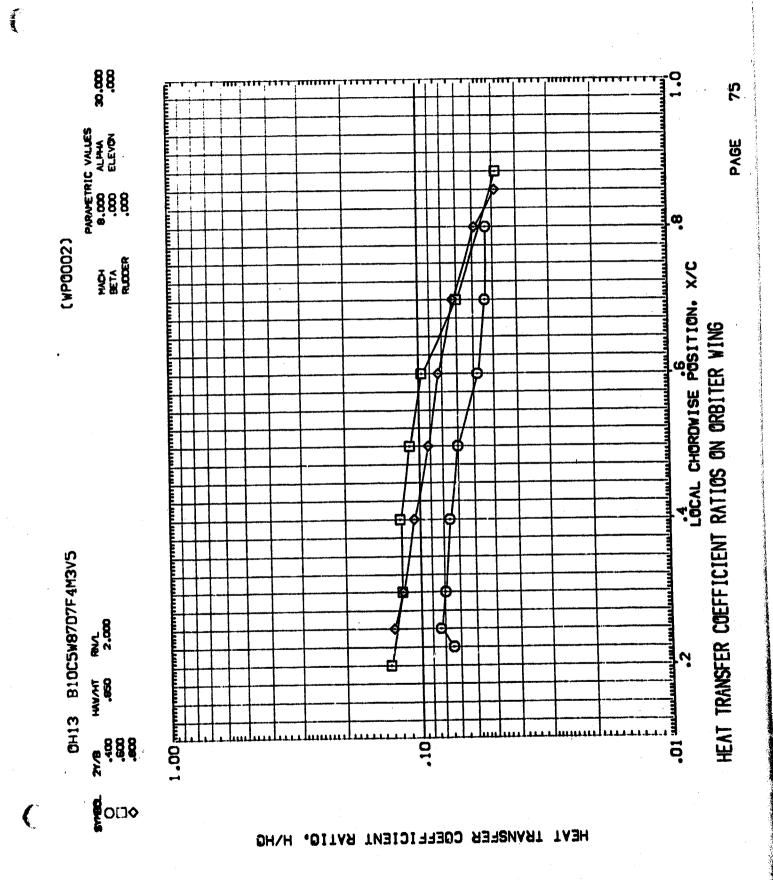


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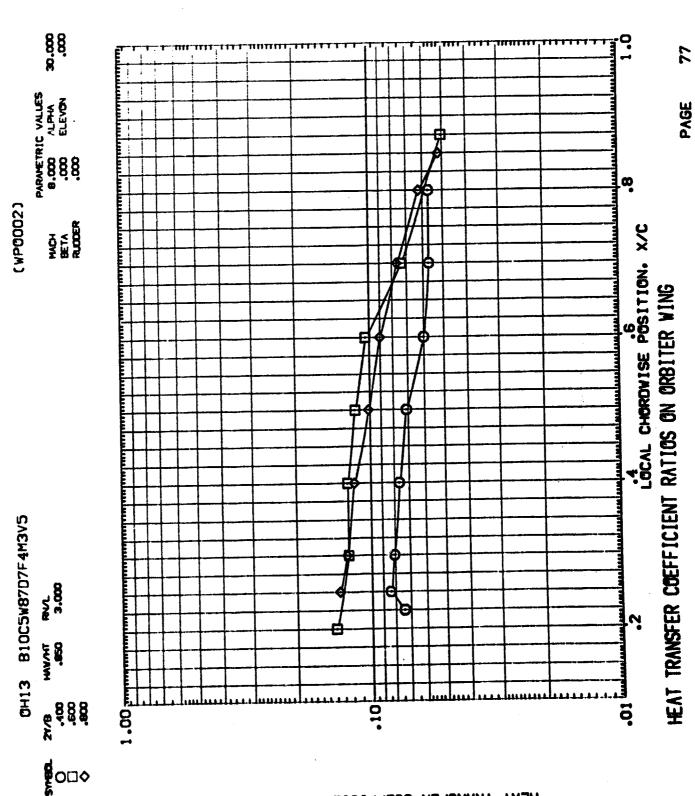
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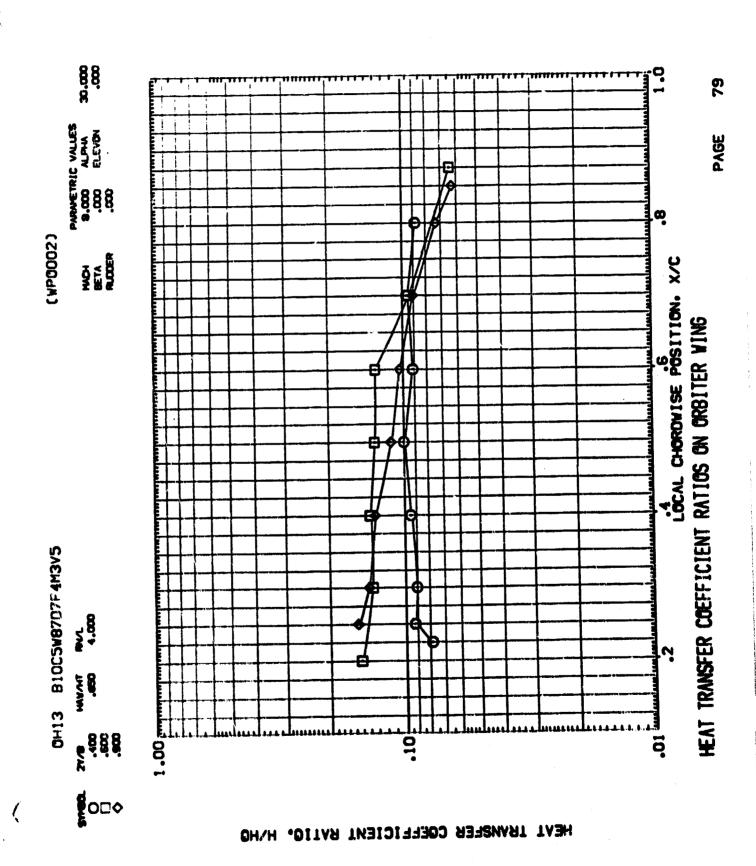
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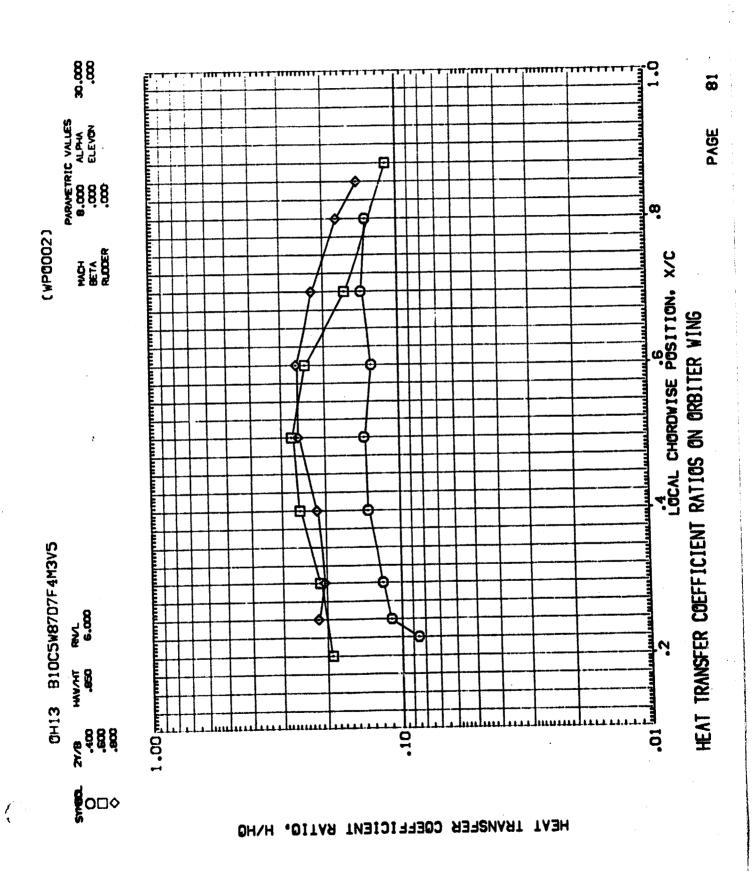
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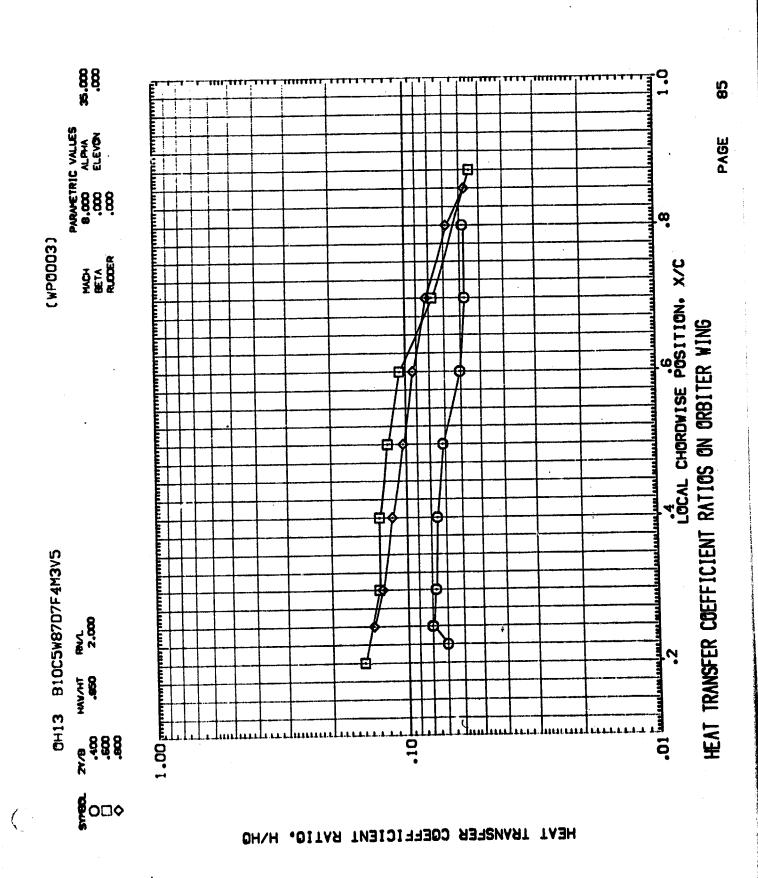
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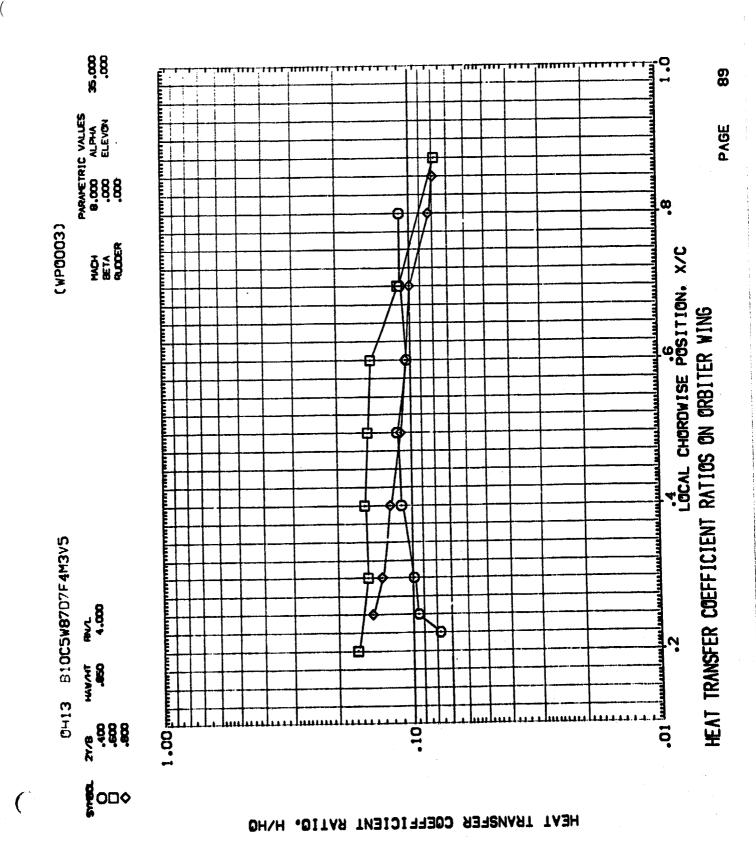


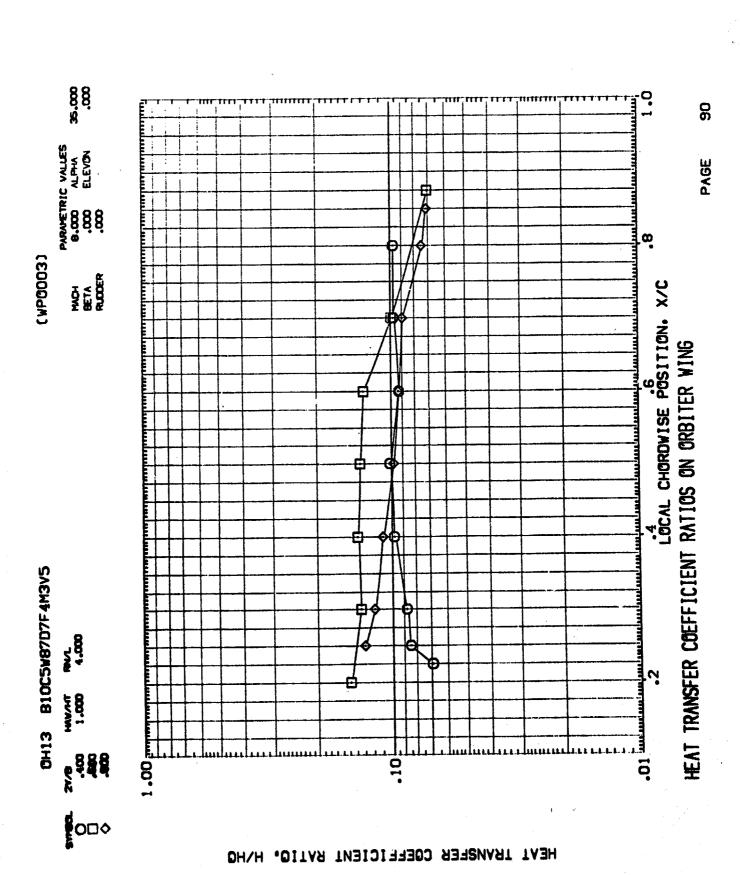
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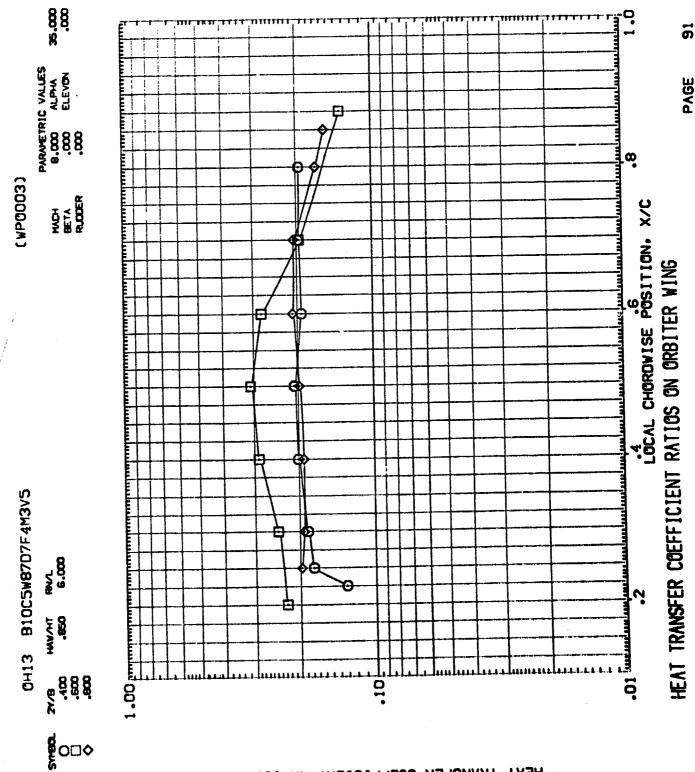
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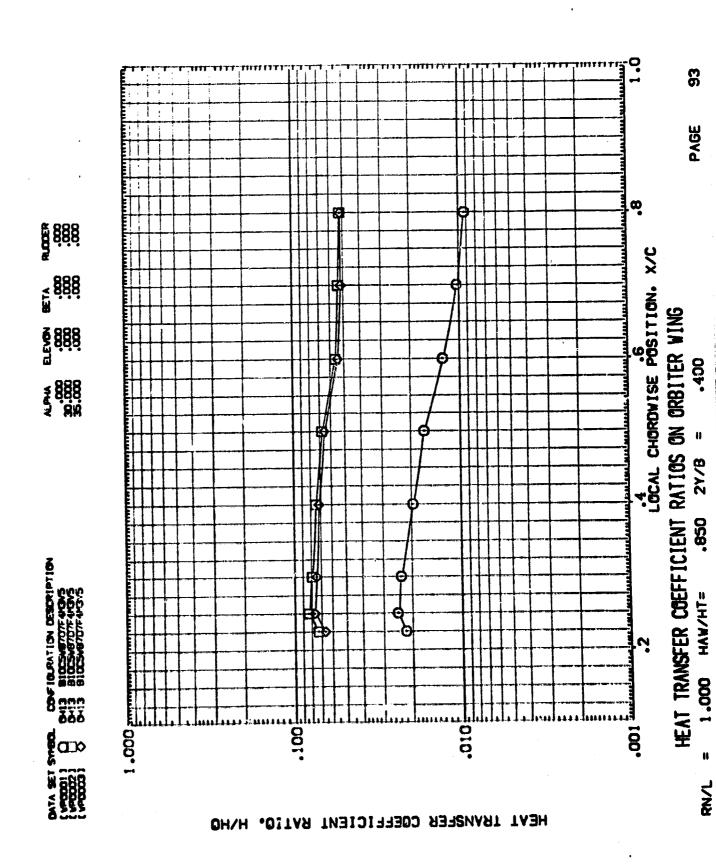


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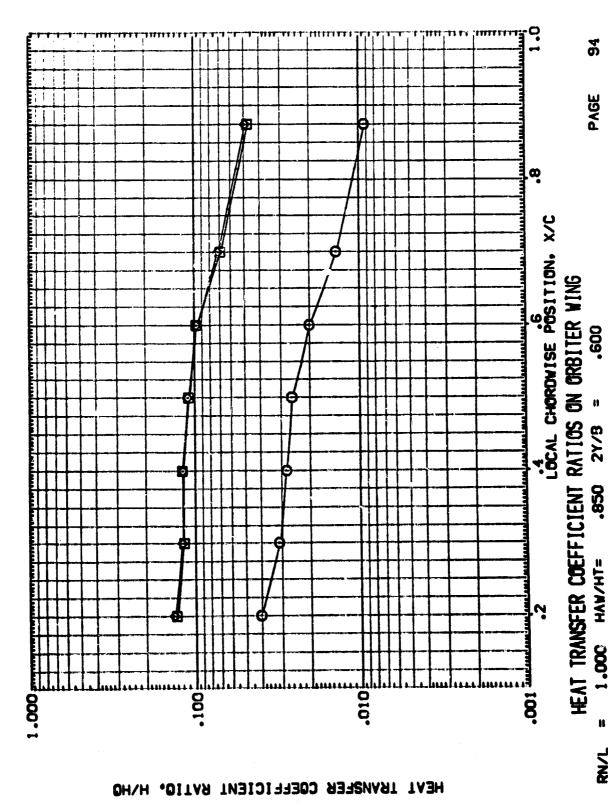
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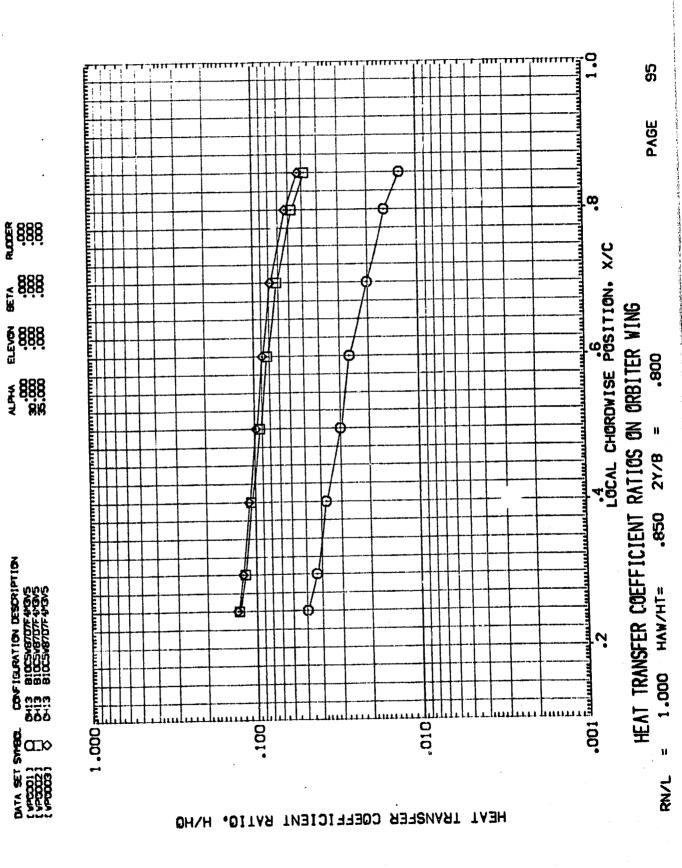


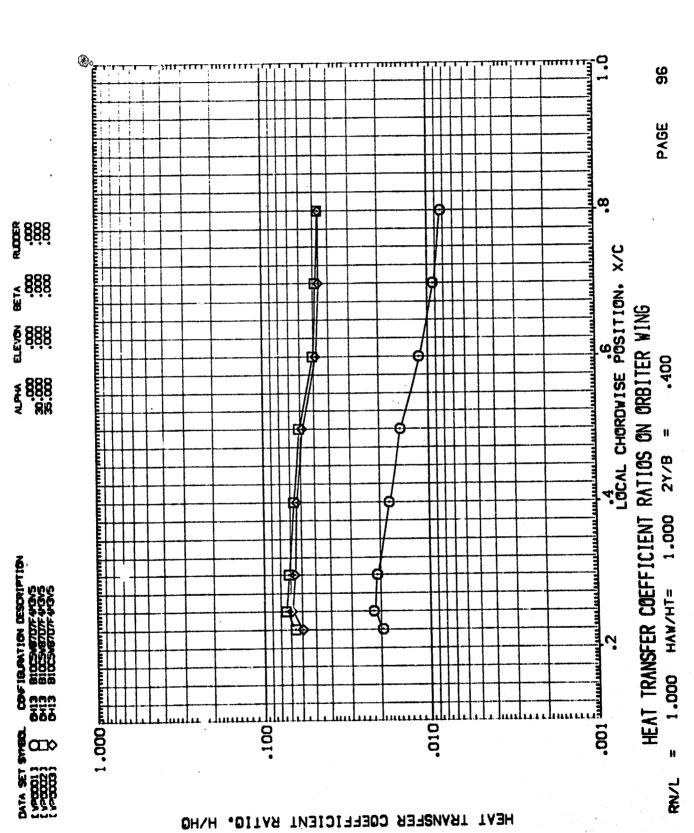




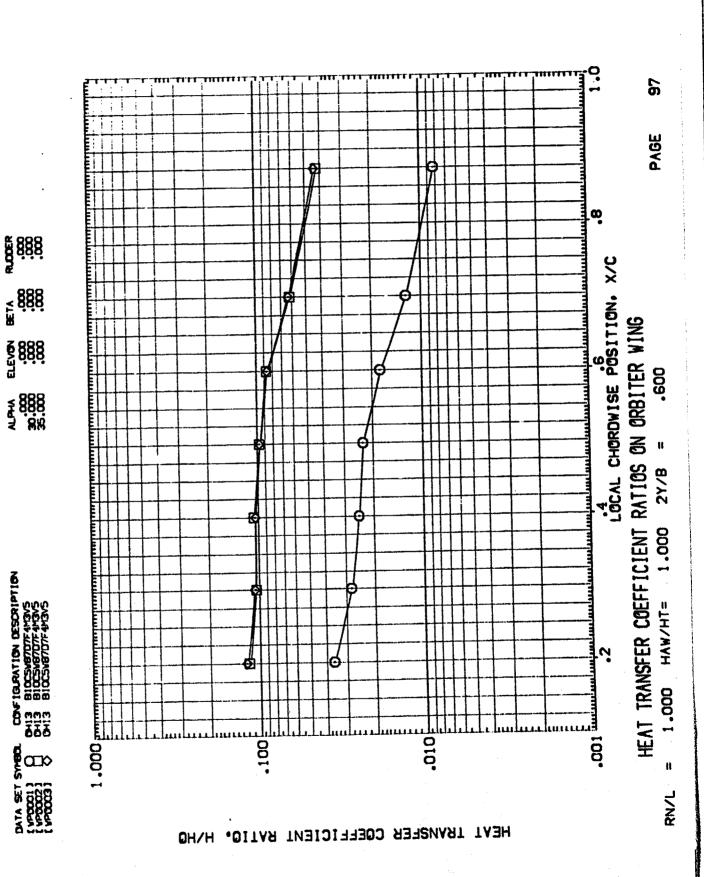






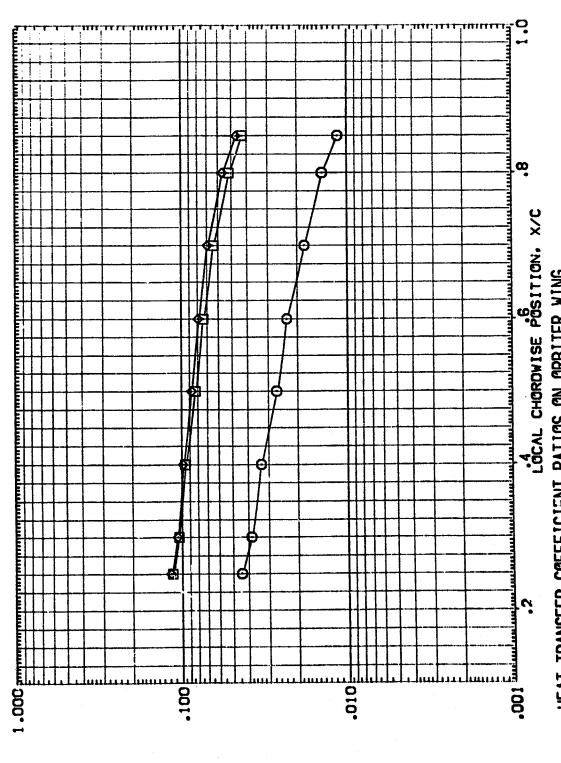






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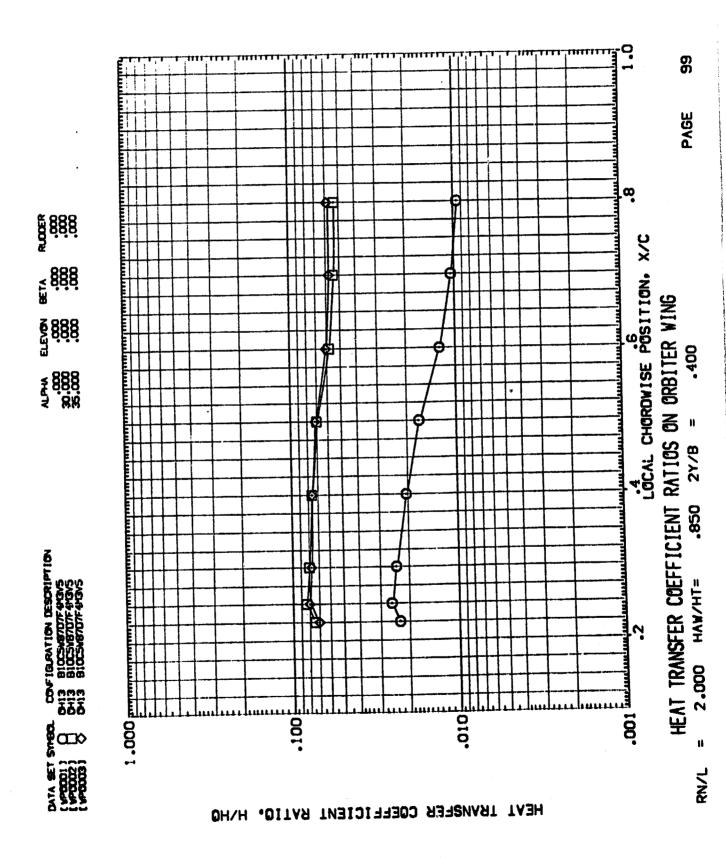




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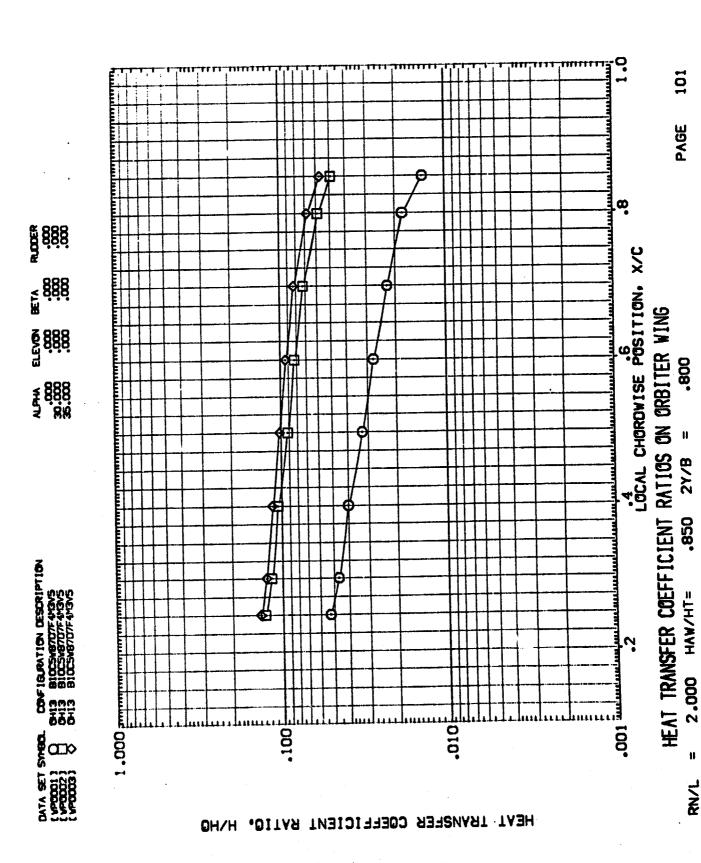
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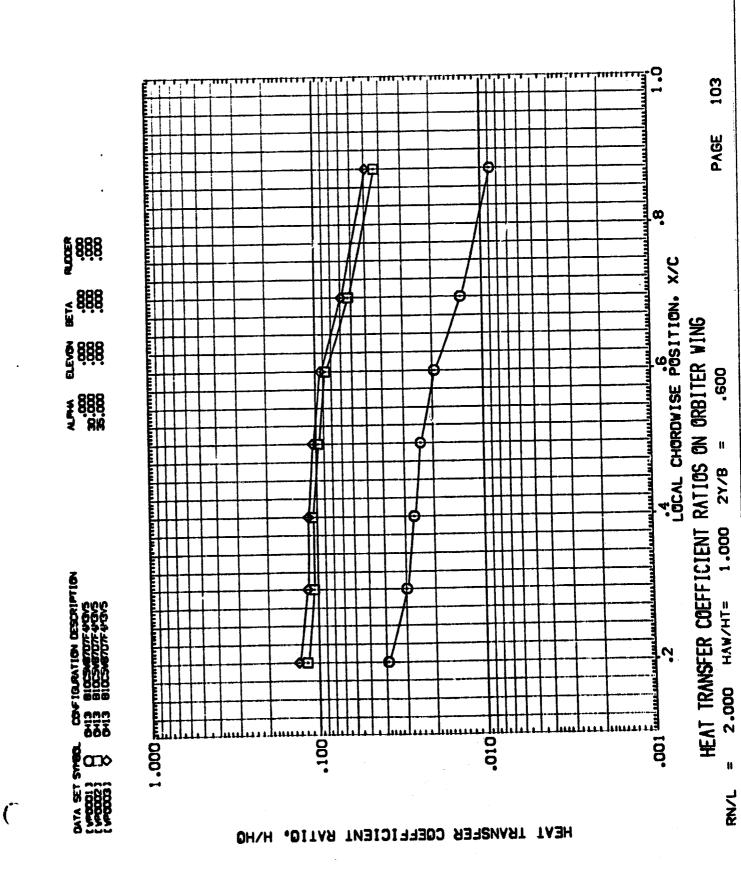
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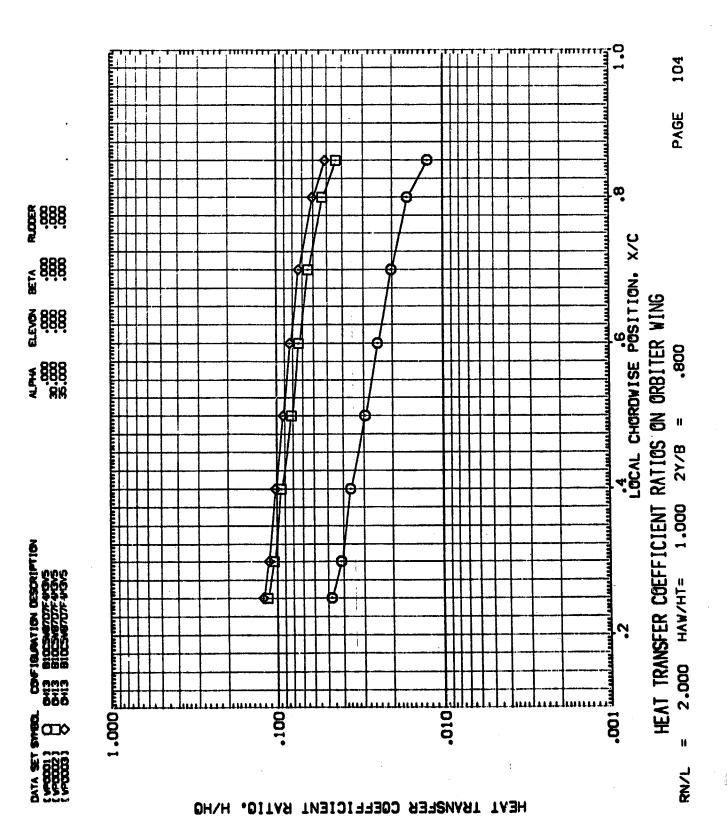
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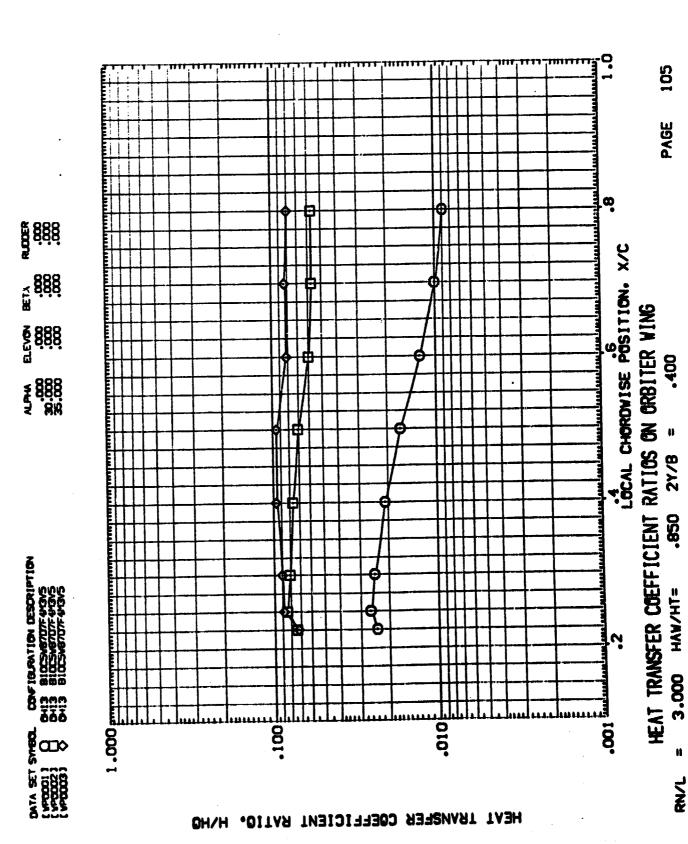


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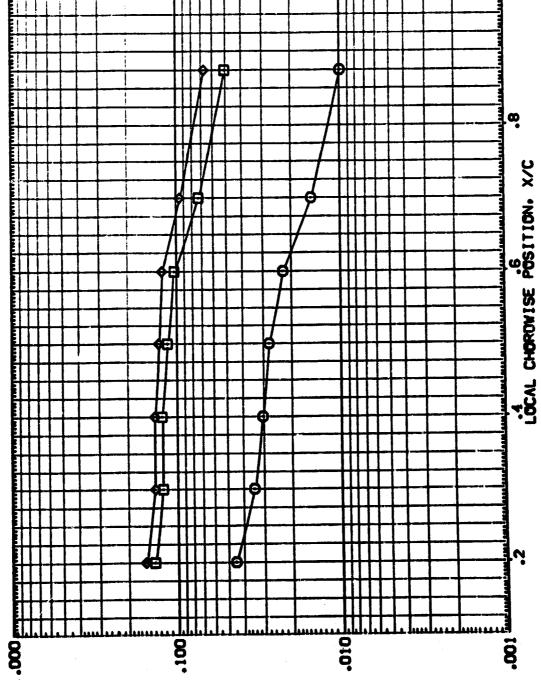
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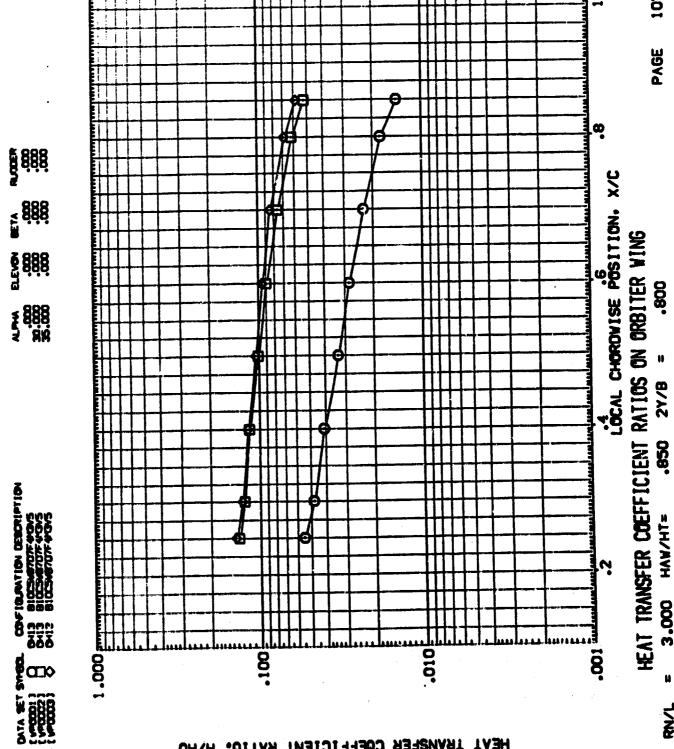
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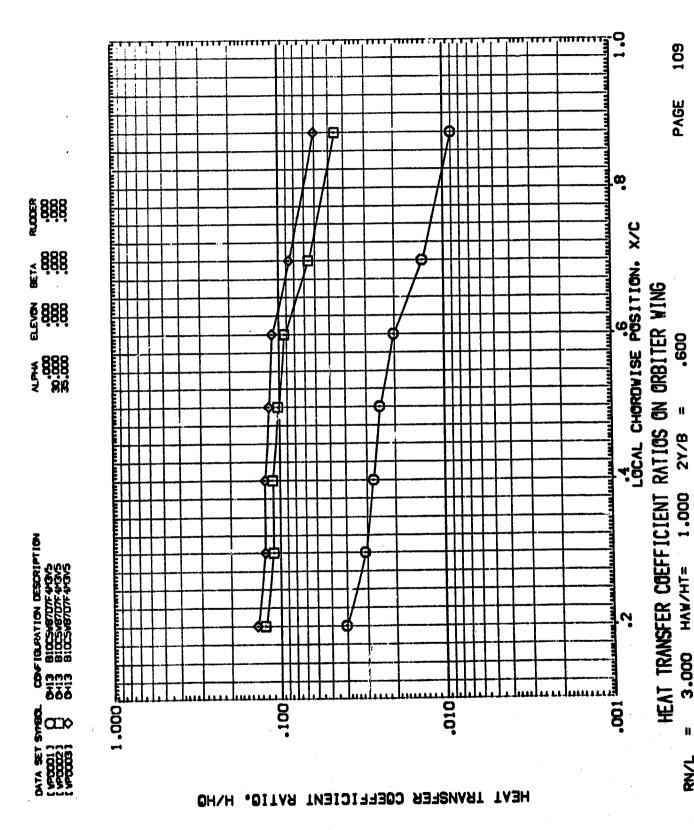
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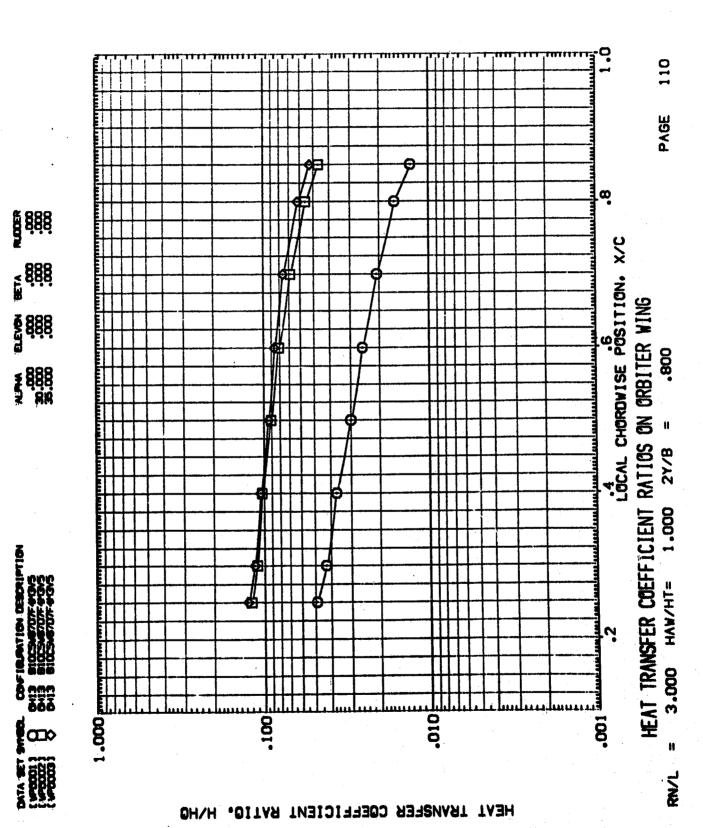
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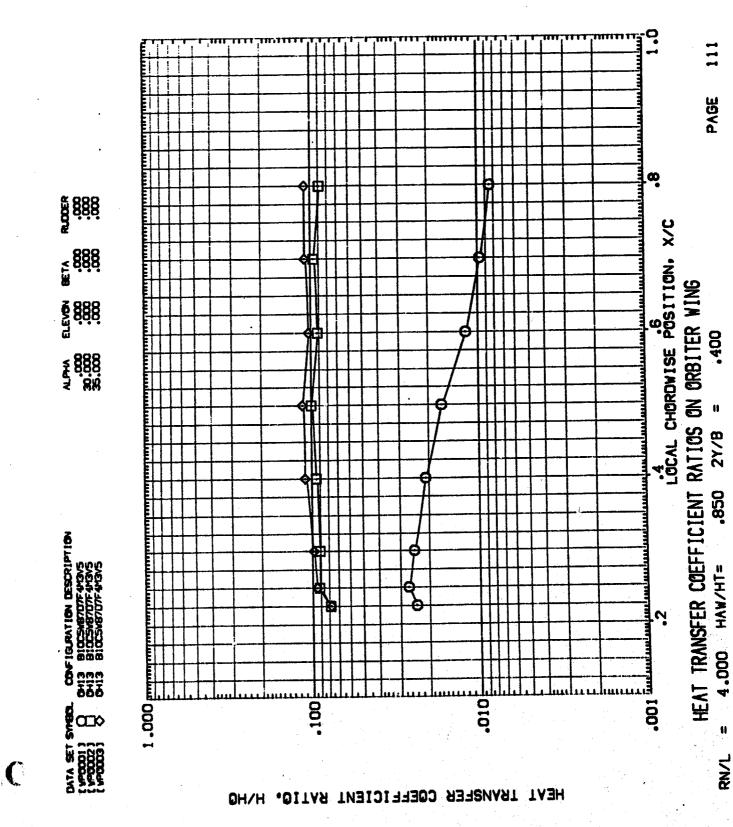
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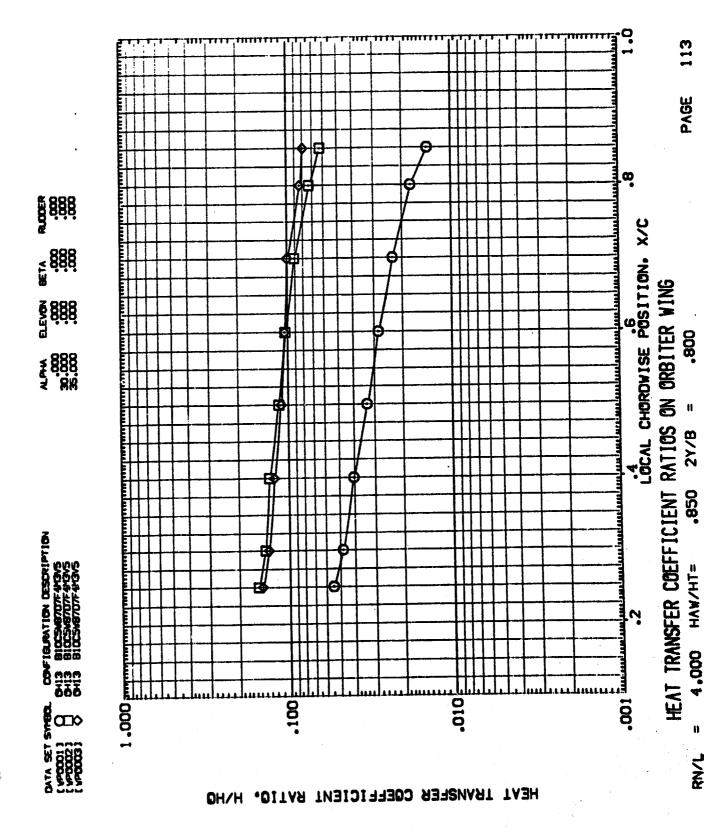
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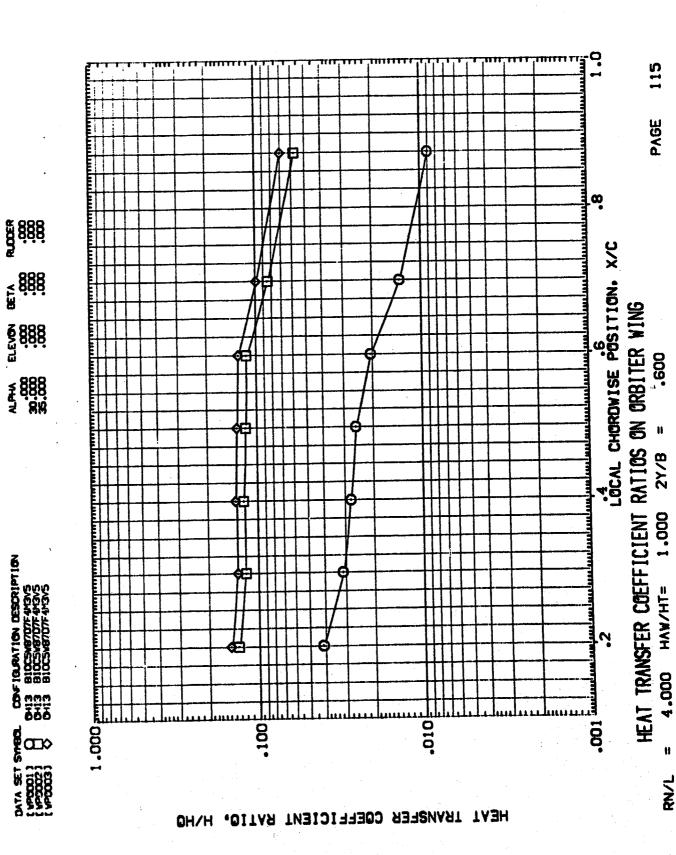
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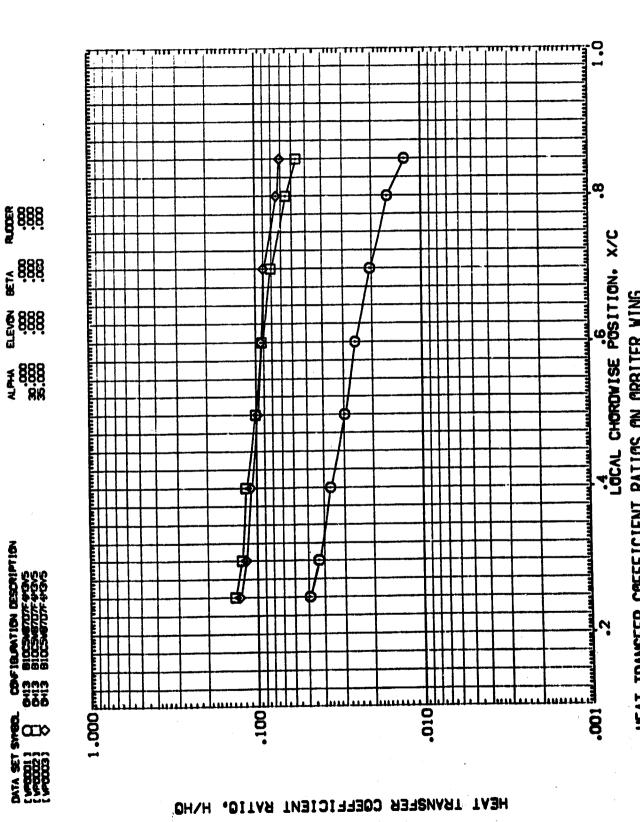
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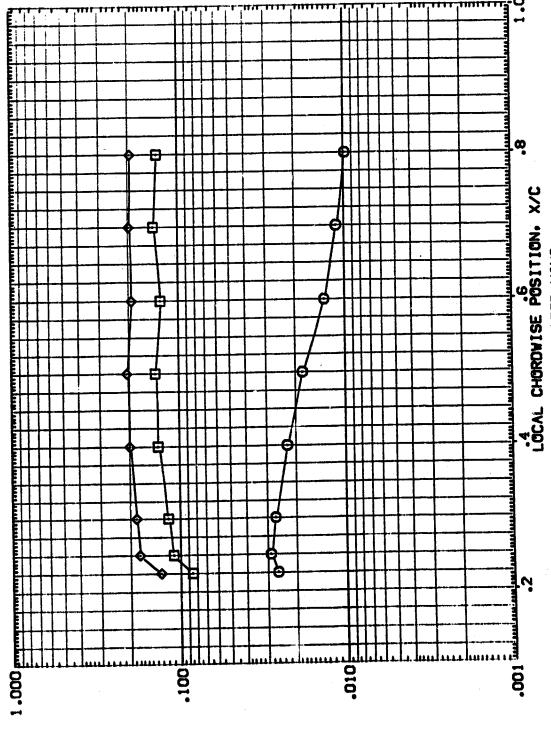
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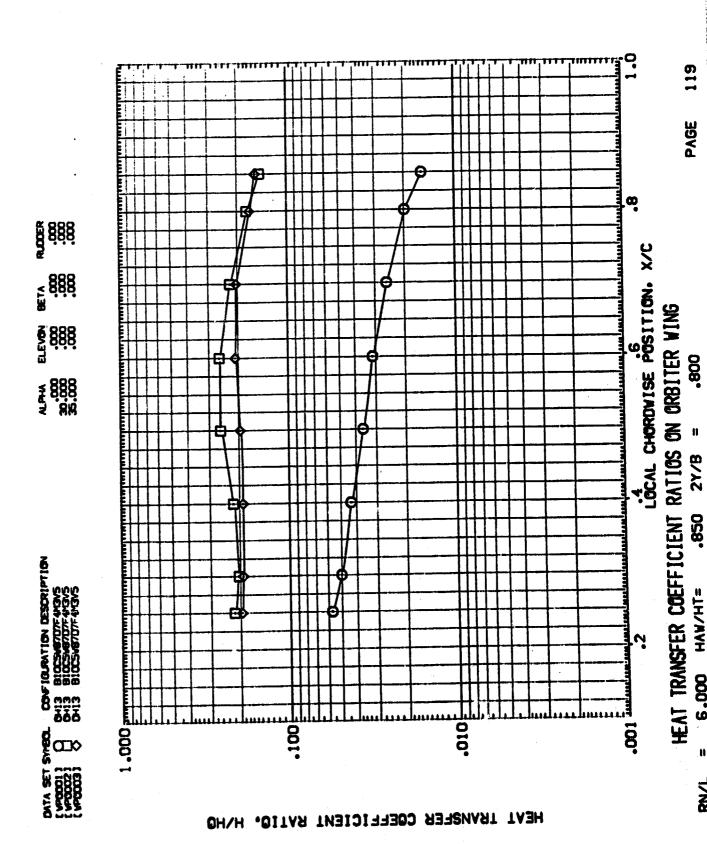
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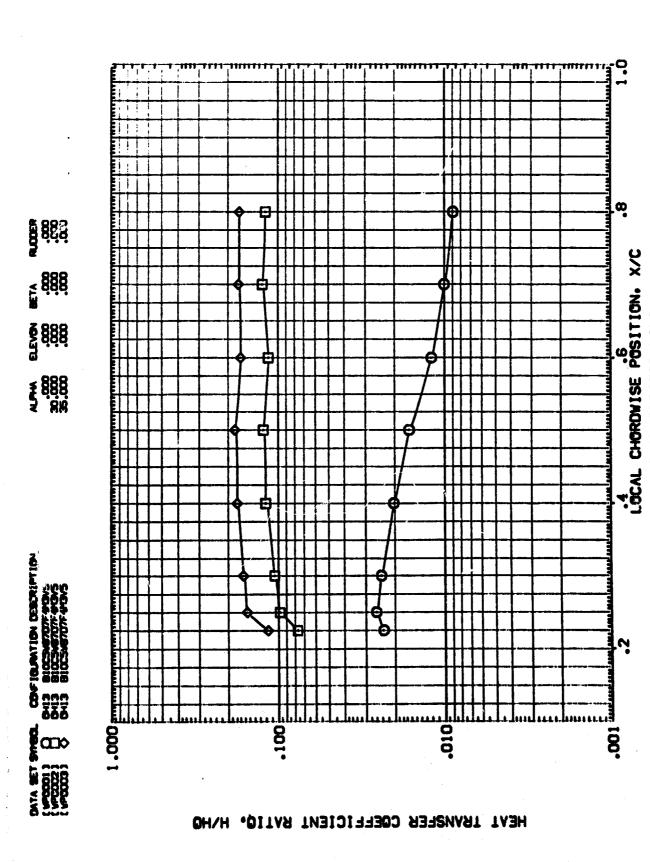
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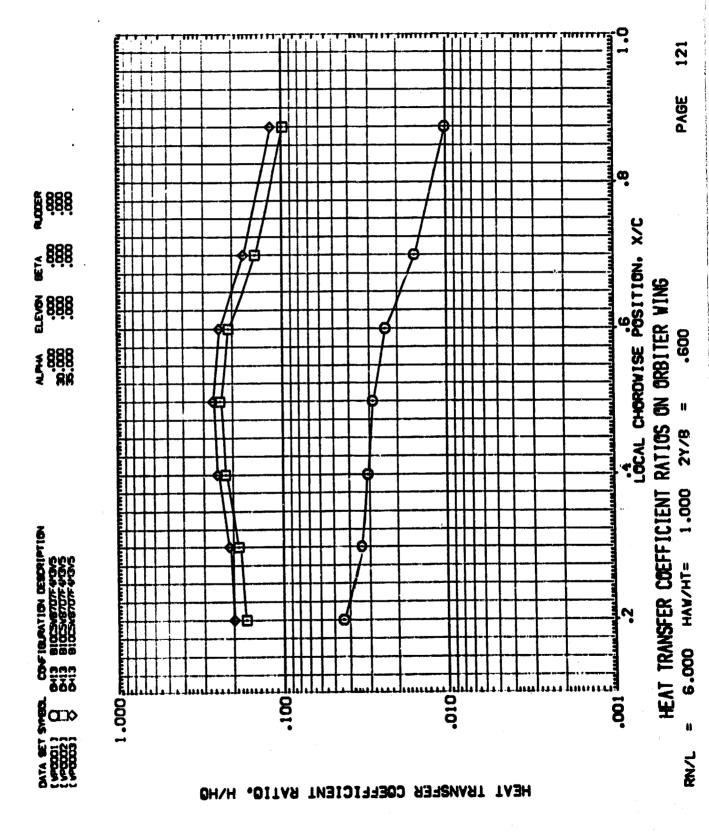
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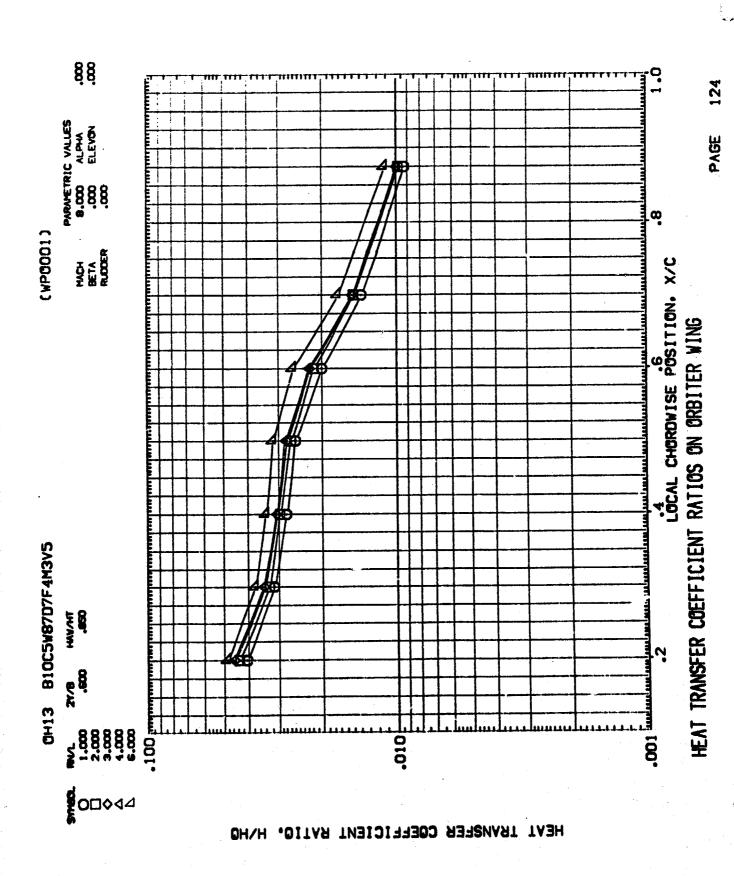
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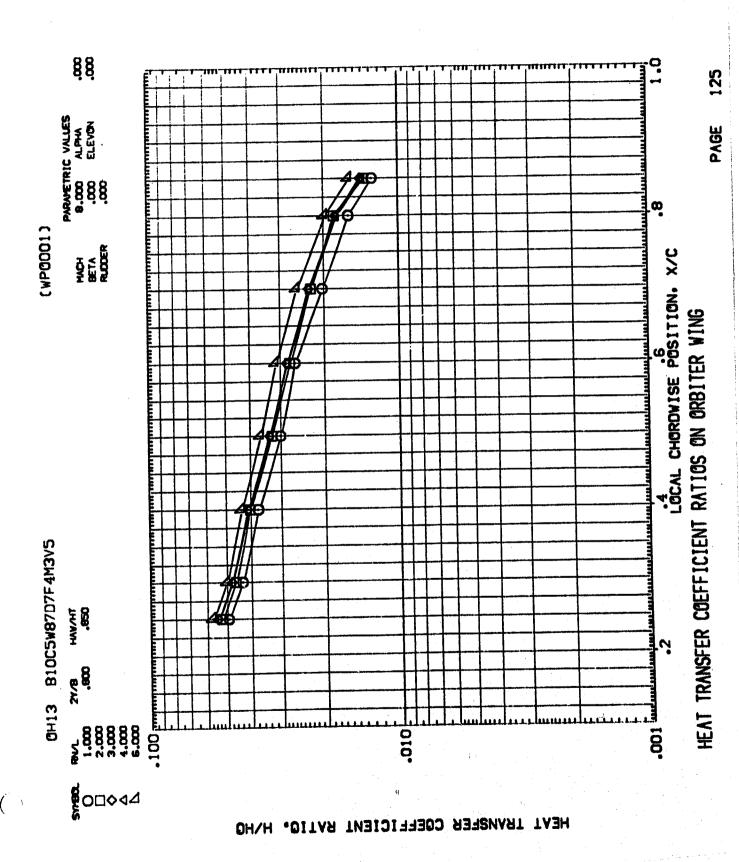
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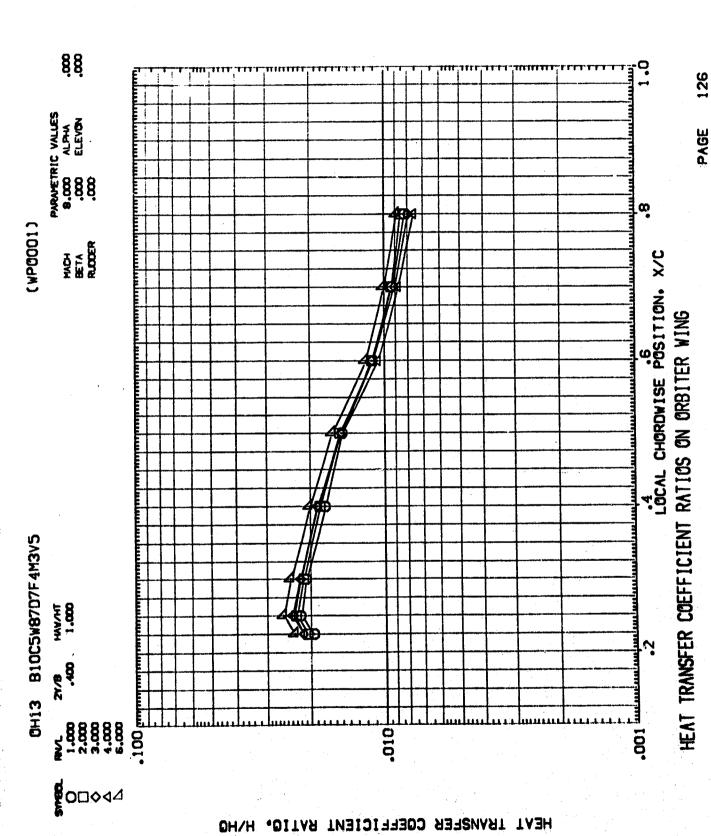
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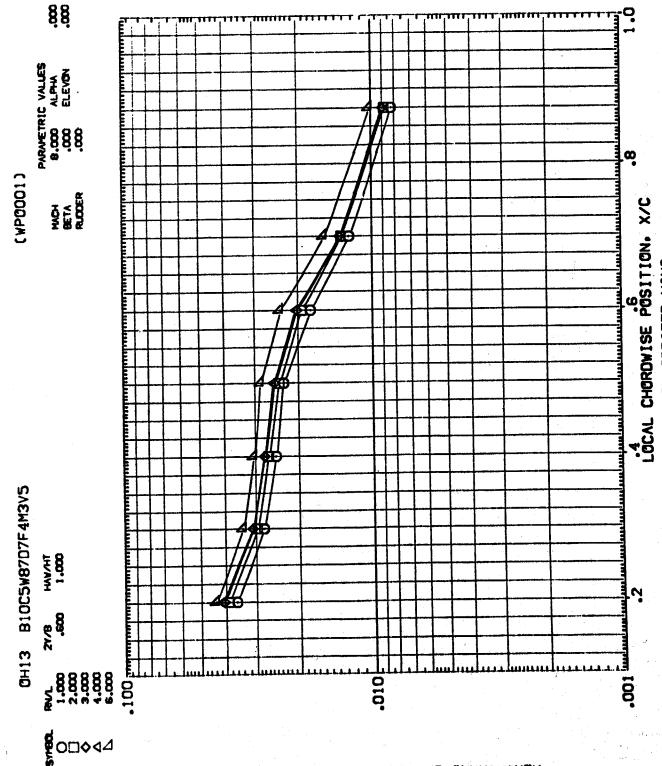
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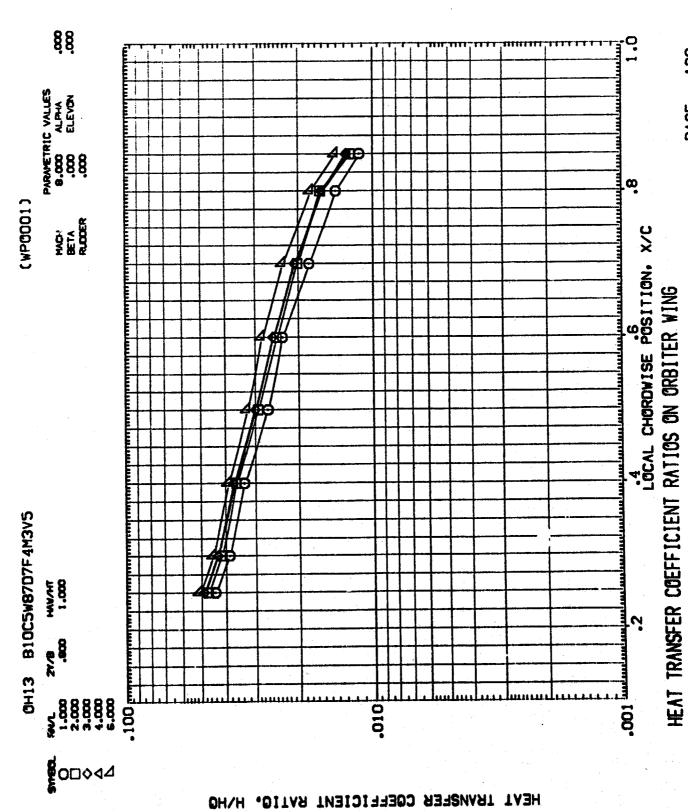




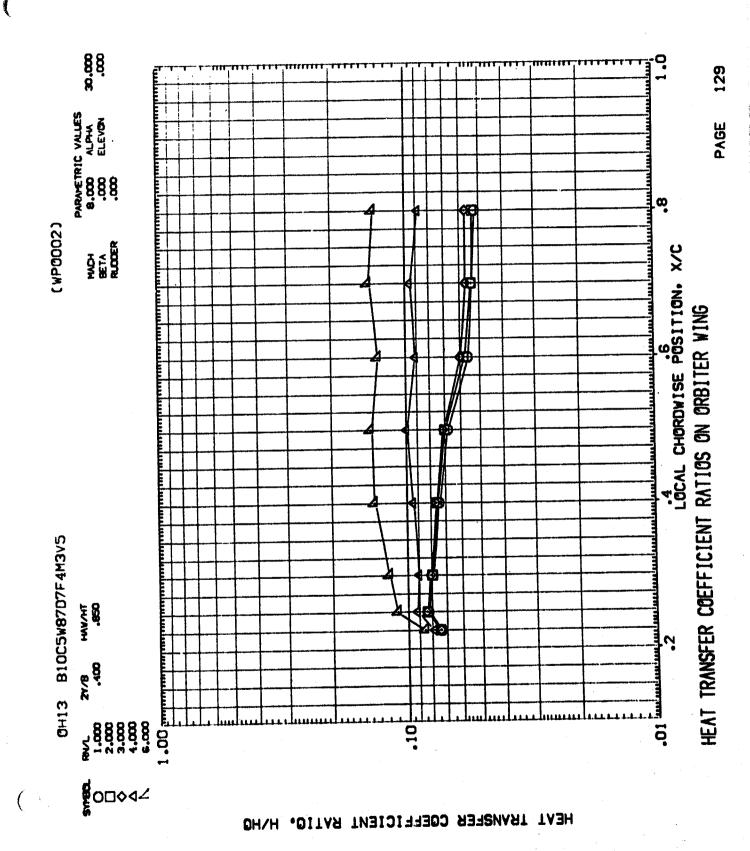
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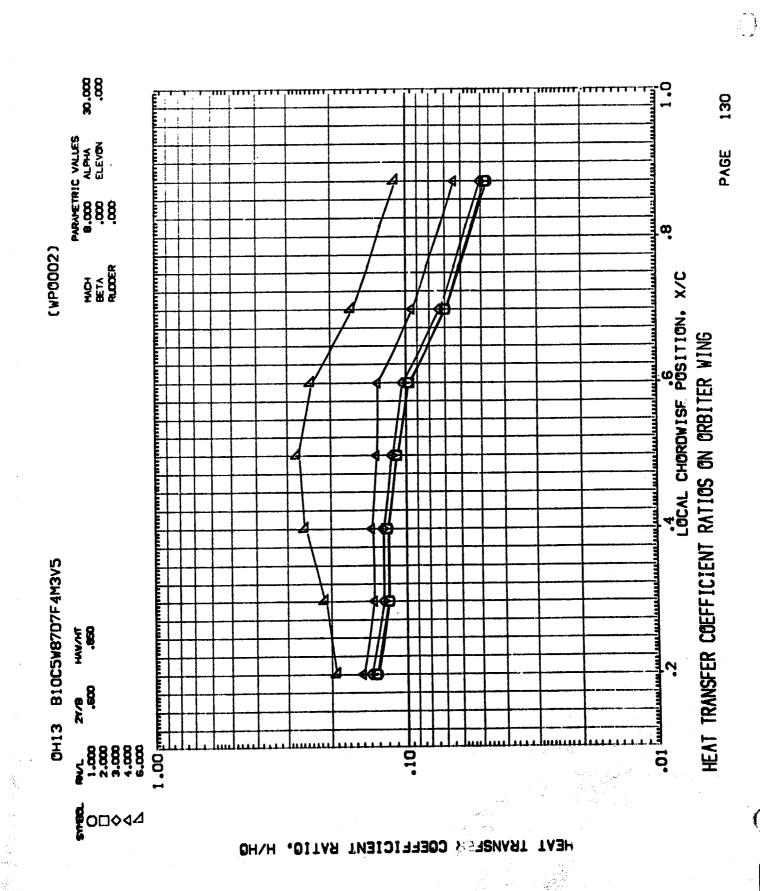
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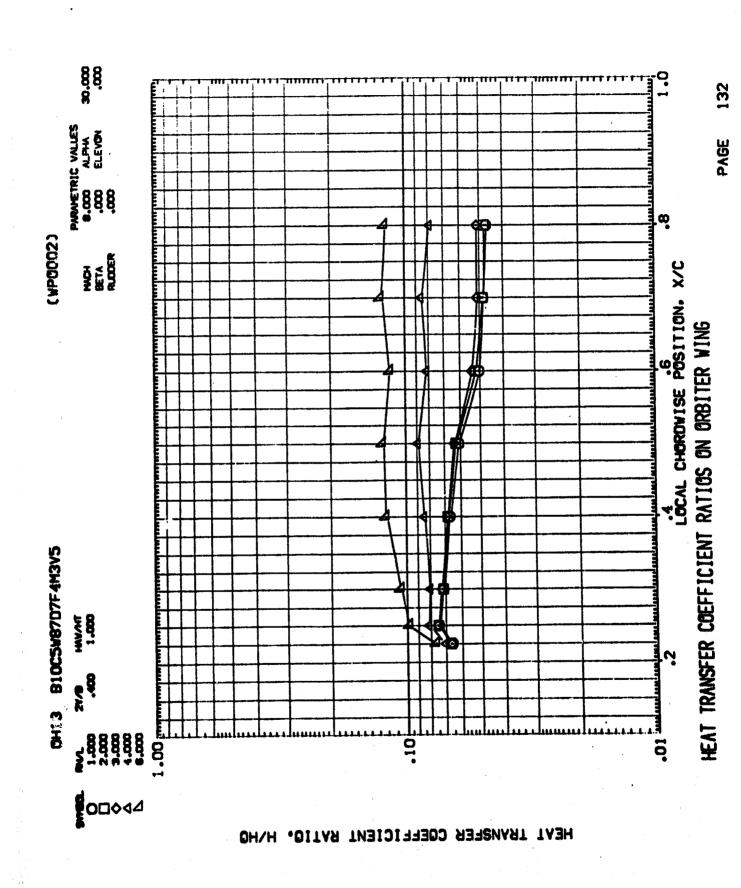
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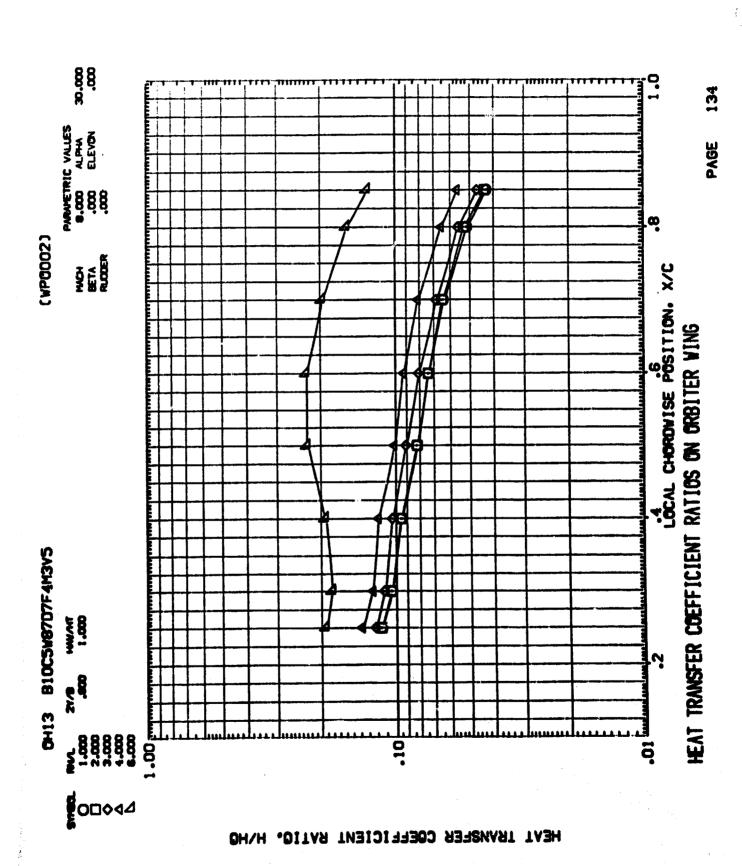




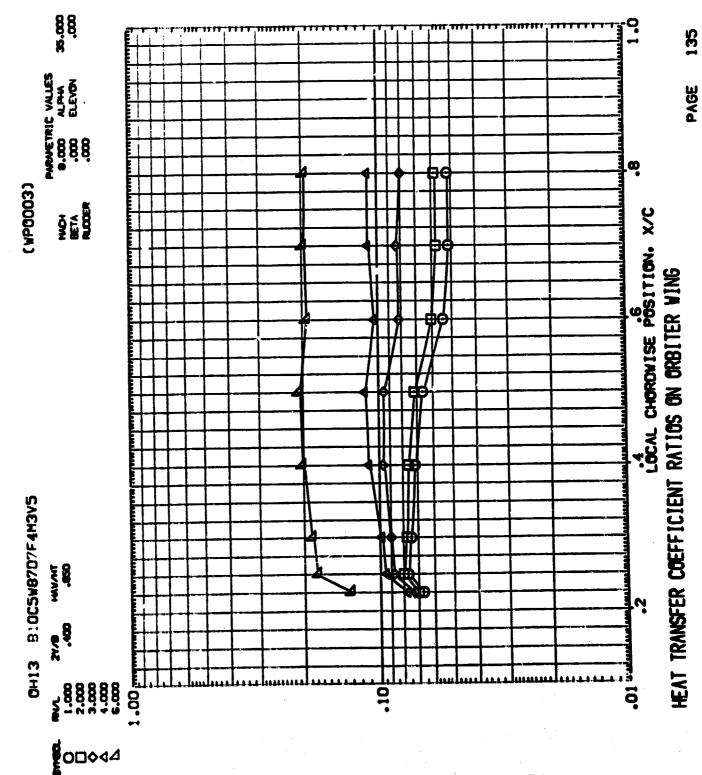




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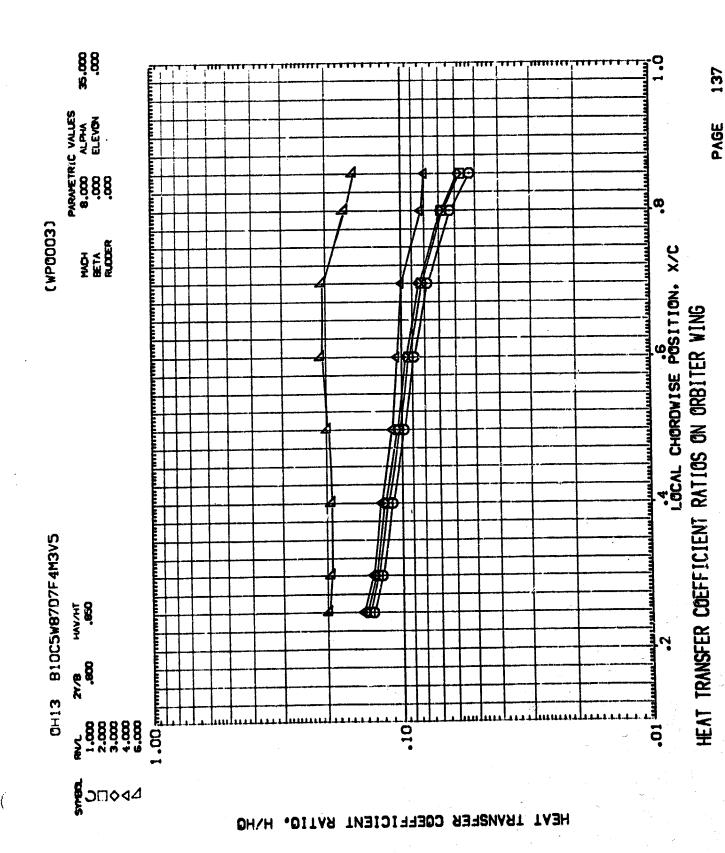


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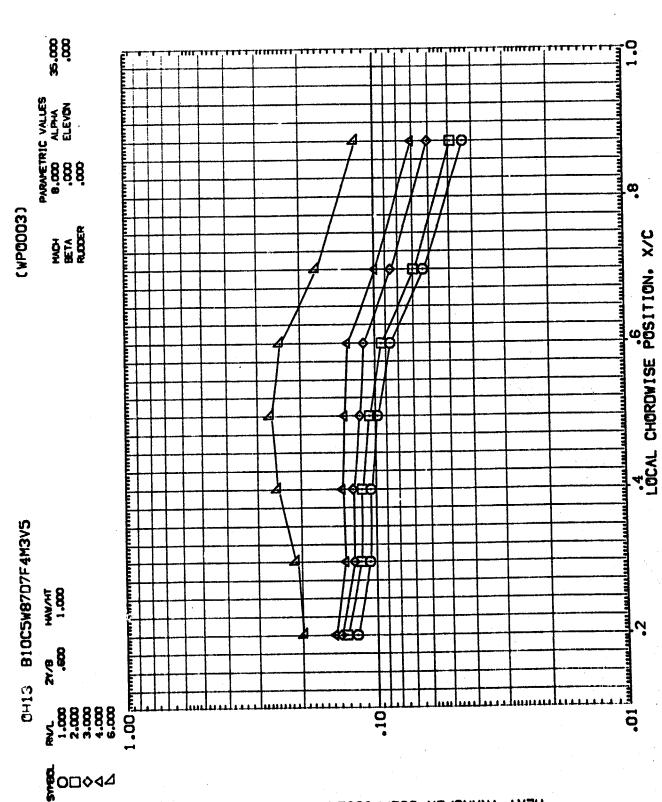
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HEAT TRANSFER COEFFICIENT RATIO. HAHO

## APPENDIX TABULATED SOURCE DATA

Tabulations of plotted data are available on request from Data Management Services.

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TABILATED HEAT TRANSFER DATA FOR CHES LARC VOHT-644
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                           DEPENDENT VARIABLE HAND
        ONS BLOCO-BROTE HOVE
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                  MA. (4) = 4,000 HMAMT(2) = 1,000
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DATE 23 JUL 74
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TABILATED HEAT TRANSFER DATA FOR ONES LARC VOHT-644

DATE 23 AL 74

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MA. ( 5) = 6.000 HMANT( 2) = 1.000

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SECTION (1)80DY

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EFFECENT VARIABLE HAYO

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                                   MACH = BETA = RUDDER =
                                                               = 169.790 TO
TABLATED HEAT TRANSFER DATA FOR OUTS LARC NO-4-644
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    DATE IS AL 74
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DATE 23 AL 74

## ORS BLOCS-BTOTF-BOVS

CEPECEAT WAINELE HAD MA. (1) = 1.000 HMANT(2) = 1.000 SECTION ( 1)BODY

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۵ = 40.225 MCH = 7.865 PO DEPOCES WRITELE HAD MA. (2) = 2.000 HMMT(2) = 1.000

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       MCH # 7.946 FO # 851.221 TO
TABLEATED HEAT TRANSFER DATA FOR CHIS LARC VOIN-644
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                     DEPENDENT VARIABLE HAND
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DATE 23 JUL 74

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= 1383,661 = 651.961 TO

DEPENDENT VARIABLE HAND ₩WH B RAV. 13) = 3.000 HAMMIT (2) = 1.000

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.0336 .0466 .0410

MACH = 7,980

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TABLATED HEAT TRANSFER DATA FOR ONLS LARC VOICE A
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                                                       CHIS BIOCOMBTDTF 44 GV5
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PAGE 11

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DATE 23 JUL 74

8.040 FO

# HO#

 $RAV_{c}$  (5) = 6.000 HALMER (2) = 1.000

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SECTION ( 1)BODY

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(BPOD02)

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              MACH :: BETA :: RUDDER ::
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TABLATED HEAT TRANSFER DATA FOR OHIS LARC YORT-642

ONIS SIDCOMBIDIFANGVS

DATE 23 JUL 74

PAGE 13

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TABILATED HEAT TRANSFER DATA FOR CHES LARC VOHT-644
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(BPCDCB)

PAGE 14

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DEPENDENT VARIABLE HAND
                  OHIS BIDCS-BTDTF-4HBV5
                                             HAMATT (2) = 1.000
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                                                RAV. (1) = 1.000
                                                                            SECTION (1)BODY
DATE 23 JUL 74
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= 407.136 TO MACH = 7.885 FO CEPENDENT VARIABLE HAHO 680 R.M. (2) = 2.000 HMANT(1) =

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= 1303.428 HD

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= 1303.428 HO = 407.136 TO MACH = 7.885 PO DEPENDENT VARIABLE HAPO RM. (2) = 2.000 HM.MT(2) = 1.000

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.0000 70.0000 SECTION (1) BODY

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    TABULATED HEAT TRANSFER DATA FOR CHIS LARC VICHT-644
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                MCH = 7,955 PO
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DATE 23 JUL 74

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DEPENDENT VARIABLE HAYO

7(1) = .650 MOH = 7.979 FO = 622.612 FO = 1360,239 HO.

.087

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   . 080
                                                                                                                                = 1363,365 HO
         ₽
         = 1360,299
(Broots)
                                                                                                                                = 1396.702 TO
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         = 852.612
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          WMCH = 7.979 FO
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                   DEPENDE VARIABLE HAYD
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PAGE 17

TABULATED HEAT TRANSFER DATA FOR CHIS LARC YOHT-644

DATE 23 JAL 74

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OHS BIOCSUSTOFFAMOVS

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MA. ( 5) = 6.000 HALMIT (2) = 1.000

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SECTION ( 1)BODY

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DEPENDENT VARIABLE HAND

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BETA ==
RUDDER ==
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     TABILLATED HEAT TRANSFER DATA FOR OHIS LARC VOHT-644
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(BPCDDA)

ONS BICCLED THONS WITH 500 DEG PAINT HALANT ( 2) = 1.000

DEPENDENT WRIMBLE HAND

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DATE 23 AL 74

MA. (1) = 3.000

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PAGE 21
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 TABILATED HEAT TRANSFER DATA FOR OHIS LARC VOHT-644
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PAGE 22



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TABILATED HEAT TRANSFER DATA FOR CHIS LARC VIHIT-644

DATE 23 JUL 74

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(MPCD01)

				OH3 BIOCOMBTDTF4HGV5	87DTF-4M5V5						(MPCD01)		•	. !
3V (4) =	4.000		HALMT( 2) =	1,000	# HOM	7,980	8	= 855	855.101	ဥ	= 1374.570	£	H	.087
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TABILATED HEAT TRANSFER DATA FOR CHI3 LARC VOHT-644

CHIS BIOCSUBTOTF 44GV5

(1000A)

DEPENDENT VARIABLE HAHO RAVI. (5) = 6.000 HALMIT (2) = 1.000.6000 .6000 .4000 SECTION ( 1) WING

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DATE 23 JUL 74

DATE 23 JUL 74

PARAMETRIC DATA

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SAED = 2660,0000 SQ.FT. WARP = LAED = 474,6000 IN. WARP = BRED = 936,7000 IN. ZMRP = SCALE	.000. .000. .000. .000.			MACH :: BETA :: RUDGER ::	8,000 ALPHA = ,000 BLEVON = ,000	30.000
RAV. (1) = 1.000 HAWHT(1) =	.650	# HOH	7.778 60	= 7,778 PO = 189,790 TO	= 1306,705 HO	ti
SECTION ( 1) WING	NECENDEN	DEPENDENT VARIABLE HAND	ρ			

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SECTION ( 1) WING

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TABILATED HEAT TRANSFER DATA FOR CHIS LARC VOHT-644
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  DATE 23 JUL 74
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CHIS BIDCS/BTDTF44GV5
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DEPENDENT VARIABLE HAYO HALMIT (1) = RVL (3) = 3.000 SECTION (1) WING

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27.AB

.0733 .0569 .0568

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= 1383.661 ρ = 651.961 8 HMH = 7.946 1.000 HMM/HT ( 2) = 3.000 RAVL (3) =

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DEPENDENT VARIABLE HAYD SECTION ( 1)MING

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PAGE

CHES BLOCOLOTTE 446V5

DEPENDENT VARIABLE HAHO RM, ( 5) = 6.000 HMJMT( 2) = 1.000

SECTION ( 1)MING

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(MPODDR)

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           ( 50 JUL 74 )
PAGE 31
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                                     ALPHA ...
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RUDGER =
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                                                                   = 163,902
TABILATED HEAT TRANSFER DATA FOR CHIS LARC VOHT-644
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DATE 23 JUL 74

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DEPENDENT VARIABLE HAYD 9 9 **.** SECTION ( 1)MING 27.9

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= 1395.163 HO = 698.624 TO 8 MCH = 7,955 DEPENDENT VARIABLE 4/10 650 HEATT( 1) = RVL (3) = 5.000 SECTION ( 1)MINE

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TABILATED HEAT TRANSFER DATA FOR OHIS LARC VOHT-644
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    DATE 23 JUL 74
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PAGE 33

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DATE 23 AL 74



CATE 23 JUL 74

ONS BIDCH-BIDTH-WOVS

AN. (5) = 6.000 HALME(2) = 1.000
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DATE 23 AL 74

(20 14, 74 )

(MODD4)

## CHES BLOCOLOTOFF MOVS WITH 500 DES PAINT

PARAMETRIC DATA	8.000 ALFHA = 35.000 .000 BLEVON = .000	= 1369.792 HO = .072
	MACH = BETA = RUDGER =	= 670,937 TO
		= 7.950 Po
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9 = 1369,792 ρ = 670.937 MOH = 7.950 PO HALANT ( 2) = 1,000 S AV. (1) = 3.000

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TABILATED HEAT TRANSFER DATA FOR CHIS LARC VOHT-644
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BREF = 936,7000 IN.
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DATE 23 JAL 74

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ONS BIDGS-BTDTF446VS

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SECTION ( 1) CANOPY

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DEPENDENT VARIABLE HAYD

SECTION (1)CANDPY

108 . 108 PAGE 39 = 1421.927 HO = 1421.927 HD (00001) = 1411.324 TO = 1411.324 TO TABILATED HEAT TRANSFER DATA FOR CH13 LARC VINIT-644 HWANT( 2) = 1.000 MACH = 8.040 PO CH 040.8 = HOAM DEPENDENT VARIABLE HAND DEPENDENT VARIABLE HAYD OHS BIOCOLOTOTEMBYS .850 HALMHT ( 1) = Rev. (5) = 6.000 RAV. (5) = 6.000 SECTION ( 1) CANOPY SECTION (1)CANOPY 0000 .1575 .2312 .2674 9000 .1572 .1972 .223 DATE 23 JUL 74 1,000 2,000 3,000 3,000 3,000 3,000

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DATE 23 JUL 74

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SCALE =

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DATE 23 JAL 74

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MACH = 8,040 PO .630 HWANT( 1) = Rev. ( 5) = 6,000

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PAGE 43



.108 . 108 PAGE 45 = 1363,365 HO = 1363.365 HO (000003) = 1396.702 TO = 1396.702 TO TABILLATED HEAT TRANSFER DATA FOR 0413 LARC VOHT-644 8.03e PO MACH = 8.038 PO DEPENDENT VARIABLE HAND CEPENDENT VARIABLE HAYD ONS BIDCS-B7D7F4MSV5 ₽ HALMIT (2) = 1.000 .850 HWATT( 1) = RAVL (5) = 6.000 Rev. (5) = 6.000 SECTION (1)CANDPY SECTION (1)CANDPY 0000 .017 .021 .029 9000 62.10. 68.10. 62.20. DATE 23 JUL 74 1,000 2,000 3,000 1.000 2.000 3.000

DATE 23 AL 74

(CHOODDA) (20 JUL 74 ) PARAMETRIC DATA CHES BEICCHAFTEREVS WITH 500 DEG PAINT

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DEPENDENT VARIABLE HAYO SECTION (1) CANDPY

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DEPENDENT VARIABLE HAYD

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SECTION ( 1) CANDY

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         ( 20 JUL 74 )
PAGE 47
                             ALPHA ...
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                   PARMETRIC DATA
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       TABILATED HEAT TRANSFER DATA FOR CHUS LARC VOHT-644
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BREF ...
SCALE ...
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DATE 23 JUL 74

CHIS BIOCS-BROTF-44/GV5

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                                                                                = 696.624
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                                                                                 HMH = 7.955
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TABILATED HEAT TRANSFER DATA FOR CHIS LARC VOHT-644
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  DATE 23 JUL 74
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(C)

DATE 23 JAL 74

(BPO204) (20 JUL 74 )

PARAMETRIC DATA

## REFERENCE DATA

WACH = 8.000 ALPHA = BETA = .000 BLEVON = RUDGER = .000	670,937 TO = 1389,792 HO	
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2.5716 3.1418 3.6158 3.6629 3.1364 2.6642 2.8252 2.7732 3.1754 3.3043 3.2908 4.408 3.327 2.323 2.743 2.743 2.9163 3.0735 2.5922 2.7562 

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            ( 12 THE 02 ) ( 1000M)
                                      ALPHA ...
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                                                                       = 1230.889 HO
                        PARAMETRIC DATA
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                                       HACH =
BETA =
RUDDER =
                                                                       = 182.026 TO
TABILATED HEAT TRANSFER DATA FOR CHIL3 LARC VOHT-644
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                                                                        MCH = 7.772 FO
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PAGE 57

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DATE 23 JAL 74

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= 1404.168 (MOGDI)

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= 646,050

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7.945

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7.945

DEPENDENT WARIABLE (DOT

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= 1374.570

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= 855,101

MACH # 7,980. PO

7,980

DEPENDENT WATABLE ADOIT

108

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= 1421.927

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= 1411.324

8

0**7**078 =

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8,040

DEPENDENT WATABLE ADOIT

CATE 23 ML 74

ONS BIDCSMBTDTF4M3V5

RAV. (5) = 6.000 MACH (1) = 8.040

DEPENDENT VARIABLE (CO)T SECTION ( 1) WING

.4009. 60009. 6000 21.0

.6047 1.2336 1.6057 .7092 1.4091 1.1245 2X 270 360 360 379

.6055

(MPOQD1)

DATE 23 JUL 74

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LREF = 474,8000 IN.

BGEF = 936,7000 IN.

SCALE = .0059 SCALE
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= 1340.003 HO ρ = 401.225 8 T.883 7.863 HOM (1) = 2.6248 2.6248 2.0950 1.8894 1.6375 1.3274 1.1060 2.4472 2.2075 1.5904 **2.6199** 2.6697 2.9119 2.000 1.6772 1.6961 1.6152 1.7043 1.5680 1.2936 1.2421 1.2421 RA. (2) = 

DEPENDENT WRIMBLE ODDI

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SECTION ( 1)MING

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TABILATED HEAT TRANSFER DATA FOR CHIS LARC NOHT-644
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   DATE 23 JUL 74
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PAGE

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7.649 12.645 13.0214 6.0362 14.561 13.7933 9.3077 15.5934 15.7574 6.6702 14.1960 15.6056

DATE 23 AL 74

OUS BLOCS-BTOTF-44GVS

DEPENDENT VARIABLE ADOIT

RAAL ( 5) = 6.000 MACH ( 1) = 6.040

SECTION ( 1) WING

27.7

9.2372 10.5914 13.5769 0.5315 11.2270 9.5699 

7.4834

(MODE)

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## ONLS BLOCS-BROTF 4NGV5

( 20 JUL 74 )

RANETRIC DATA

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             ALPHA =
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                                                = 1246.346 HD
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                                                                                                                                                                                      = 407.136 TO
            HACH = BETA = RUDDER =
                                                 = 183.902 TO
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                                                 MCH = 7.774 PO
                                                               CEPENDENT VARIABLE GOOT
             .000.
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                                                  7.774
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2.225
2.2197
1.9609
1.9960
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REFERENCE DATA
              S4EF = 2490,0000 50,FT.

LAEF = 474,6000 IN.

S4EF = 936,7000 IN.

SCALE = .0059 SCALE
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2.3586
2.2067
1.9558
1.4517
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                                                   RAC (1) = 1.000
                                                                                                            1.3628
1.5727
1.4611
1.5538
1.1127
1.1127
1.1023
                                                                 SECTION ( 1) WING
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DEPENDENT VARIABLE COOT

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SECTION ( 1)MING

4.0075

4.4081 4.0746 3.7314 3.3471 3.0802 2.7753 2.2857 1.9286

4.1327 4.1053 5.8191 5.4149 2.5709

2.310 2.567 2.567 2.367 2.025 2.025 1.9317

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DATE 23 JUL 74

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CHIS BIDCSMETERFAMBV5

Fav. ( 5) = 6.000 MACH ( 1) = 8.038

SECTION ( 1) WING

DEPENDENT VARIABLE ODDIT

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xxC .700 10.3761 10.1651 10.4996 .800 10.3009 8.9076 .850 8.3941

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PAGE 67
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TABLLATED HEAT TRANSFER DATA FOR CHI3 LARC VOHT-644
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                                                                                                                                                                                                                                                                                                     7.945
                                                                                                    7.72
                                                                                                                                                                                                    MCH (1) =
                                                                                                                                                                                                                                                                                                     = (') HOM
                                                                                                    MCH (1) =
                                                         REPERBACE DATA
                                                        = 2690,0000 $0.FT.
= 474,8000 IN.
= 936,7000 IN.
E = .0059 SCALE
                                                                                                    RAC (1) = 1.000
                                                                                                                                                                                                     RMA. (2) = 2.000
                                                                                                                                                                                                                                                                                                                                                                                                        RN. (4) = 4.000
                                                                                                                                                                                                                                                                                                      RVA. (3) = 3.000
                                                                                                                                                                                                                                                                                                                                                                                                                         SECTION (1) CANDPY
                                                                                                                       SECTION (1) CANDPY
                                                                                                                                                                                                                       SECTION (1) CANOPY
                                                                                                                                                                                                                                                                                                                        SECTION (1) CANDY
                                                                                                                                                                                                                                                                                                                                                                  6.1006
6.6041
7.7296
                                                                                                                                                                                                                                         ago:
                                                                                                                                                                                                                                                                                                                                                                                                                                           9000
                                                                                                                                         9000
                                                                                                                                                                  2.0242
2.8597
3.5063
                                                                                                                                                                                                                                                                                                                                          0000
                                                                                                                                                                                                                                                                   3.9153
5.4697
                                                                                                                                                                                                                                                                                      6.4702
    CATE 23 JUL 74
                                                                                                                                                                                                                                                                                                                                                           1,000
2,000
3,000
                                                                                                                                                          1,000
2,000
3,000
                                                                                                                                                                                                                                                          17.000
2.000
3.000
                                                                             BREF = SCALE =
                                                            68
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. 190:

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<u>8</u> 8

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Po.

.087



7.2247 7.4703 8.7299

2.000 2.000 3.000 7.47 KD

. 901 PAGE 68 = 1421.927 HO (CPOQD1) WACH = 8,040 FO = 1411,324 TO TABULATED HEAT TRANSFER DATA FOR CH13 LARC VOHT-644 DEFENDENT VARIABLE COOT CHIS BIDCOMBTDTF4MBV5 RAV. (5) = 6.000 MACH (1) = 8.040SECTION (1)CANOPY 0000 CATE 23 JUL 74

7.4P ND 8.2190 1.000 8.2190 2.000 10.3282 3.000 11.8698

.

TABILATED HEAT TRANSFER DATA FOR CHI'S LARC VOHT-644 DATE 23 JUL 74

1

(CHOODE) (20 JUL 74 )

CHIS BLOCS-BTDTF-44/GV5

REFERENCE DATA

.0000. .0000. .0000. .0000 # # # # \$2.00 \$2.00 \$2.00 \$3.00 F = 2690,0000 SQ.FT. F = 474,8000 IN. F = 936,7000 IN. LE = .0059 SCALE b

= 169,790 TO 8 HACH = 7.778 7.778 SCALE =

DEPENDENT WARIABLE COOT MACH (1) == RAC (1) = 1.000 SECTION (1)CANDY

980

1,000 2,000 3,000

.1556 .2151 .2916

8 7.883 ¥ 7.883 HACH (1) = RAV. (2) = 2.000

DEPENDENT WATABLE ODDIT

8

9

= 1340,003

δ

= 401.225

SECTION (1) CANOPY

8

.3190 .4653 .6141 1,000 2,000 3,000

8 HACH = 7,946 7,946 MCH (1) = RVL (3) = 3,000

DEPENDENT WARIABLE GOOT

.075

₽

= 1383,661

ρ

= 651,961

SECTION (1)CANDPY

1,000 1,000 2,000 3,000

85. 87. 87. 87.

= 857.278 TO 8 7,980 ¥ 7.980 HQH (1) == RM. (4) = 4.000

DEPENDENT VARIABLE GOOT

980

9

= 1373,090

SECTION (1) CANOPY

0000

**₹** 

.7061 1.0625 1.3185 1.000 2.000 3.000

PAGE

PARAMETRIC DATA

MACH :: BETA :: RUDDER ::

ALPHA = ELEVON =

ġ

오

= 1306.705

1

PAGE 70 = 1489.263 HO (20005) FAV. (5) = 6.000 MACH (1) = 8.040 MACH = 8.040 FO = 1422.212 TO TABULATED HEAT TRANSFER DATA FOR CH13 LARC WHT-644 DEPENDENT WOTABLE ODDI CHES BECCS-BTDTF-4MBVS SECTION (1)CANOPY DATE 23 JUL 74

. 110

0000

1.2467 1.8507 2.2611 1.000 2.000 3.000

1

(CHOORDS) (SD 10L 74 )

ALPHA = PARAMETRIC DATA **6** 000.00 000.00 MACH BETA .000. .000. .000. .000. REPERBACE DATA

ġ 35.000 = 1246.346 HD = 183.902 TO RUDDER = 8 MACH = 7.774 7.774 MACH (1) SREF = 2690,0000 SA.FT. LREF = 474,8000 IN. BREF = 936,7000 IN. SCALE = .0059 SCALE RNL (1) = 1.000

DEPENDENT VARIABLE about SECTION (1) CANOPY

0000

158 168 100 100 100 3.000 ₹ P 8.

₽ = 1303,428 = 407.136 TO 8 7.865 # HQ# 7.865 MACH (1) = RML (2) = 2.000

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DEPENDENT VARIABLE ODOT SECTION (1) CANOPY

0000

.3878 .4611 .6523 14 NO 1.000 2.000 3.000

ρ = 698.624 8 HACH = 7,955 7,955 HACH (1) = RM. (3) = 3.000

.078

₽

= 1395.183

DEPENDENT VARIABLE (DOT SECTION (1)CANDPY

9

**₹** 

운 = 1360,299 = 852.612 TO 8 MACH = 7.979 WACH (1) = 7.979 RV. (4) = 4.000

. 60.

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DEPENDENT VARIABLE COOF SECTION (1)CANDPY

0000

₹ 8

1.0780 1.0780 1.2823 3.000 1.000

. 9 2 PAGE = 1363,365 HO (CPOQE)3) = 1396.702 TO TABILATED HEAT TRANSFER DATA FOR CHIS LARC VOHT-644 8.038 PO DEPENDENT VARIABLE COOT CH13 B10C9AB7D7F4AGV5 ₩GH HACH (1) = 8.038 RAVL (5) = 6.000 SECTION (1)CANDPY DATE 23 JUL 74

9000

1.1666 1.3575 1.6091 1.000 2.000 3.000

TABLLATED HEAT TRANSFER DATA FCR CH13 LARC YOHT-644

PAGE 73

ONES BLOCSMBTDTF4HOVS WITH 500 DEG PAINT

PARAMETRIC DATA

ALPHA = 9.00 000:

MACH = BETA = RUDGER =

.0000 .0000 .0000 .0000

985 = 2690.0000 SA.FT.
LGS = 474.8000 IN.
BRS = 506.7000 IN.
SCALE = .0059 SCALE

REPERBACE DATA

= 1389.792 HO

= 670.937 TO

PACH = 7.950 PO

7.950

RMA. (1) = 3.000 MACH (1) =

SECTION ( 1) CANDPY

0000

84.86. 84.86. 84.86.

1.000 5.000 5.000 5.000

CEPENDENT VARIABLE ADOIT

200.

35.000

(CROSD4) (20 JUL 74 )

CATE 23 JUL 74